

Approximating strain equations for infinitesimal theory:

①  $\epsilon(x, e_i) = \sqrt{e_i \cdot C e_i} - 1 = U e_i - 1$   
no summation

②  $S_{ij} \delta_{ij} = \frac{e_i \cdot C e_j}{\sqrt{e_i \cdot C e_i} \sqrt{e_j \cdot C e_j}}$   
// change of angle  
 $\frac{\pi}{2} - \theta_{ij}$

$$\nabla y = \nabla u + I$$

$$F = H + I$$

$$\epsilon = \text{Max}(H_{ij}(x))$$

$$\forall x \in D \quad \{i, j \in \{1, 2, 3\}\}$$

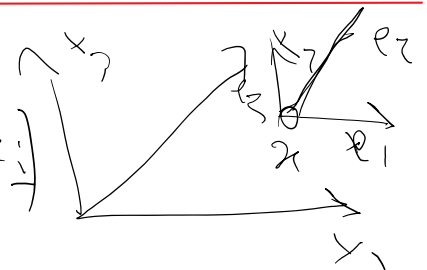


$\epsilon$  small: infinitesimal theory / approximation

Def. 82

First, we find the approximate form for relative change of length:

$$\epsilon(x, e_i) = \sqrt{C_{ii}} - 1 \quad (C_{ii} = e_i \cdot C e_i)$$



$$C = I + \underbrace{H + H^T}_{2E} + \underbrace{HH^T}_{O(\epsilon^2)} = I + 2G \rightarrow \text{Green strain}$$

$$= I + 2E + O(\epsilon^2)$$

$$E = \frac{1}{2}(H + H^T)$$

$$C_{ii} = (I + 2E + O(\epsilon^2))_{ii} = \delta_{ii} + 2E_{ii} + O(\epsilon^2)$$

$$= 1 + 2E_{ii} + O(\epsilon^2)$$

$$E = O(\epsilon)$$

$$\sqrt{1+x} = (1+x)^{\frac{1}{2}}$$

$$|x| < 1 \quad = 1 + \frac{1}{2}x + \frac{1}{2}(\frac{1}{2}-1)x^2$$

$$= \underline{\underline{1}} + 2E_{ii} + O(\epsilon^2) \quad \left[ \begin{array}{l} |x| < 1 = 1 + \frac{1}{2}x \\ + \frac{1}{2}(\frac{1}{2}x)^2 \end{array} \right]$$

$$\epsilon(x, e_i) = \sqrt{1 + \underbrace{(2E_{ii} + O(\epsilon^2))}_{O(\epsilon^2)}} = 1 =$$

very small  $\ll 1$

$$= \cancel{1} + \frac{1}{2}(2E_{ii} + O(\epsilon^2)) + \left(\frac{1}{2}\right)\left(-\frac{1}{2}\right)\underbrace{(2E_{ii} + O(\epsilon^2))^2}_{O(\epsilon^2)} + \dots$$

$$= E_{ii} + O(\epsilon^2) + O(\epsilon^2) = \dots$$

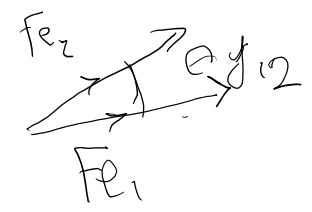
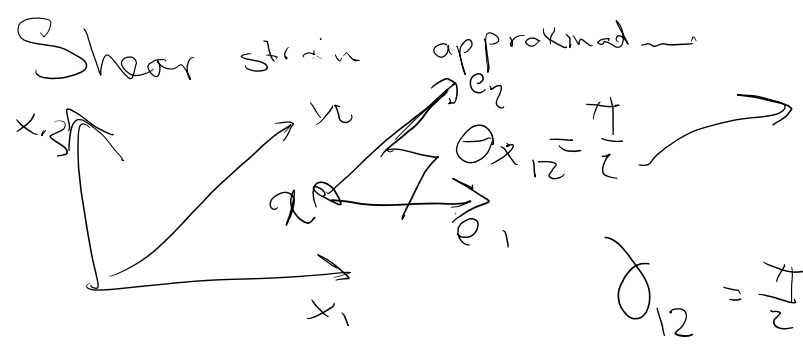
$$= E_{ii} + O(\epsilon^2)$$

①

$$\epsilon(x, e_i) = E_{ii} + O(\epsilon^2)$$

$\epsilon = \|H\|$       the norm we defined

no summation on  $i$



In general  $i \neq j$   $\sin \delta_{ij} = \frac{e_i \cdot C e_j}{\sqrt{e_i \cdot C e_i} \sqrt{e_j \cdot C e_j}}$

$$C = I + 2E + O(\epsilon^2)$$

$$e_i \cdot C e_j = C_{ij} = \left( \underset{0}{\delta_{ij}} + 2E_{ij} + O(\epsilon^2) \right) \quad i \neq j$$

$$= 2E_{ij} + O(\epsilon^2)$$

$$e_i \cdot C e_i = C_{ii} = 1 + \underbrace{2E_{ii}}_{O(\epsilon)} + \underbrace{O(\epsilon^2)}_{O(\epsilon^2)}$$

$$E = \frac{H + H^T}{2}$$

$$e_j \cdot C e_j = C_{jj} = 1 + O(\epsilon)$$

$$\sqrt{1 + O(\epsilon)} = 1 + \underbrace{\frac{1}{2}x}_{O(\epsilon)} - \frac{1}{4}x^2 + \dots \quad x = O(\epsilon^2)$$

$$\sin \delta_{ij} = \frac{2E_{ij} + O(\epsilon^2)}{\underbrace{1 + O(\epsilon)}_{C_{ij}}} \cdot \frac{1}{\sqrt{1 + O(\epsilon)}} \cdot \frac{1}{\sqrt{1 + O(\epsilon)}}$$

largest terms

$$= 2E_{ij} + O(\epsilon^2)$$

$$\sin \delta_{ij} = \underbrace{2E_{ij}}_{O(\epsilon)} + O(\epsilon^2) = O(\epsilon)$$

$$\sin \delta_{ij} = \underbrace{\epsilon_{ij}}_{O(\epsilon)} + O(\epsilon^2)$$

Small number

$$\sin \delta_{ij} = \underbrace{\delta_{ij}}_{O(\epsilon)} + \frac{1}{3} \underbrace{\delta_{ij}^3}_{O(\epsilon^3)}$$

since this is  $O(\epsilon)$

$$\delta_{ij} = 2 \epsilon_{ij} + O(\epsilon^2)$$

engineering shear strain

Tensorial strain  
half of  $\delta$

"Actual change of angle" between  $e_i$  &  $e_j$

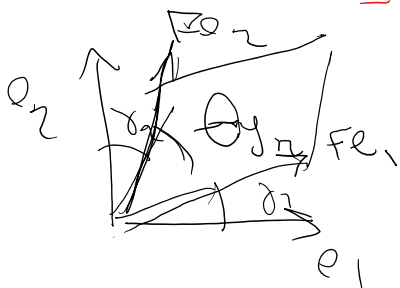
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### ★ Summary

$$\epsilon(x, e_i) = \sqrt{c_{ii}} - 1 = \epsilon_{ii} + O(\epsilon^2)$$

$$\epsilon(x, e_i) = \frac{|Fe_i|}{|e_i|} - 1$$

$$\sin \delta_{ij}(x) = \frac{c_{ij}}{\sqrt{c_{ii}} \sqrt{c_{jj}}} \rightarrow \delta_{ij} = 2 \epsilon_{ij} + O(\epsilon^2) \quad i \neq j$$



$$\delta_{xy} = \delta_1 + \delta_2 = 2 \epsilon_{xy}$$

$$\epsilon_{xy} = \text{Approximately } (O(\epsilon^2))$$

half of change of angle

half of change of angle

$$\begin{aligned}
 \underline{E} &= \begin{pmatrix} E_{11} & E_{12} & E_{13} \\ E_{21} & E_{22} & E_{23} \\ E_{31} & E_{32} & E_{33} \end{pmatrix} \\
 &= \frac{H + H^T}{2} \\
 &= \frac{(\nabla u + \nabla u^T)}{2} \quad H = \nabla u
 \end{aligned}$$

shear strains = half of change of angle  
 normal strains along  $e_1, e_2, e_3$  directions

$u(x)$  displacement vector  $\rightarrow \nabla u$   
 matrix  
 2nd order tensor  
 so is  $\nabla u^T \Rightarrow$

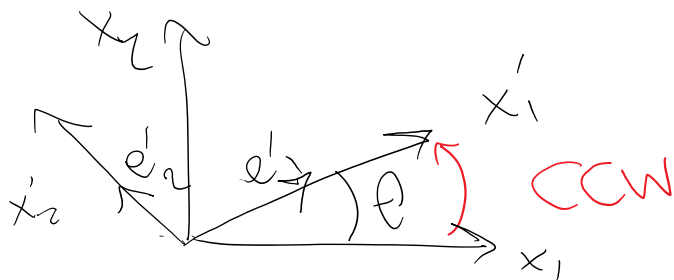
$E = \frac{\nabla u + \nabla u^T}{2}$  is a second order tensor!

likewise  $\omega = \frac{\nabla u - \nabla u^T}{2}$  Also a tensor

Now that we know it's a tensor, we know that it follows coordinate transformation rules:

$$E'_{ij} = Q_{im} Q_{jn} E_{mn}$$

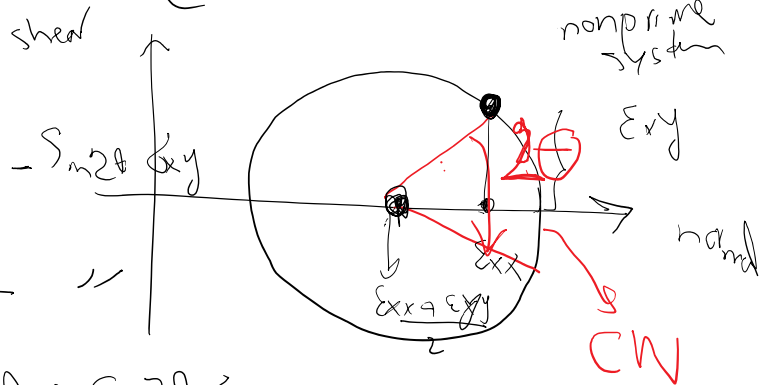
2D



$$Q = \begin{bmatrix} e'_1 \\ e'_2 \end{bmatrix} = \begin{bmatrix} c & s \\ -s & c \end{bmatrix} \quad \begin{matrix} c = \cos\theta \\ s = \sin\theta \end{matrix}$$

⇒ Show that

$$\begin{aligned} \epsilon'_{xx} &= \left( \frac{\epsilon_{xx} + \epsilon_{yy}}{2} \right) + \left( \frac{\epsilon_{xx} - \epsilon_{yy}}{2} \right) \cos 2\theta - \sin 2\theta \epsilon_{xy} \\ \epsilon'_{yy} &= \left( \frac{\epsilon_{xx} + \epsilon_{yy}}{2} \right) - \left( \frac{\epsilon_{xx} - \epsilon_{yy}}{2} \right) \cos 2\theta + \sin 2\theta \epsilon_{xy} \\ \epsilon'_{xy} &= \left( \frac{\epsilon_{xx} - \epsilon_{yy}}{2} \right) \sin 2\theta + \cos 2\theta \epsilon_{xy} \end{aligned}$$



E is a tensor, but components of normal and shear strains do not transfer like tensor for large deformation theory

2D

$$E = \begin{bmatrix} E_{11} & E_{12} \\ E_{12} & E_{22} \end{bmatrix}$$

3D

$$E = \begin{bmatrix} E_{11} & E_{12} & E_{13} \\ E_{12} & E_{22} & E_{23} \\ E_{13} & E_{23} & E_{33} \end{bmatrix}$$

$$\gamma = \begin{bmatrix} E_{11} \\ E_{22} \\ 2E_{12} \end{bmatrix} \quad \gamma_{12}$$

$$\gamma = \begin{bmatrix} E_{11} \\ E_{22} \\ E_{33} \\ 2E_{12} \\ 2E_{23} \\ 2E_{31} \end{bmatrix} \quad \begin{matrix} \text{normal} \\ \text{shear components} \end{matrix}$$

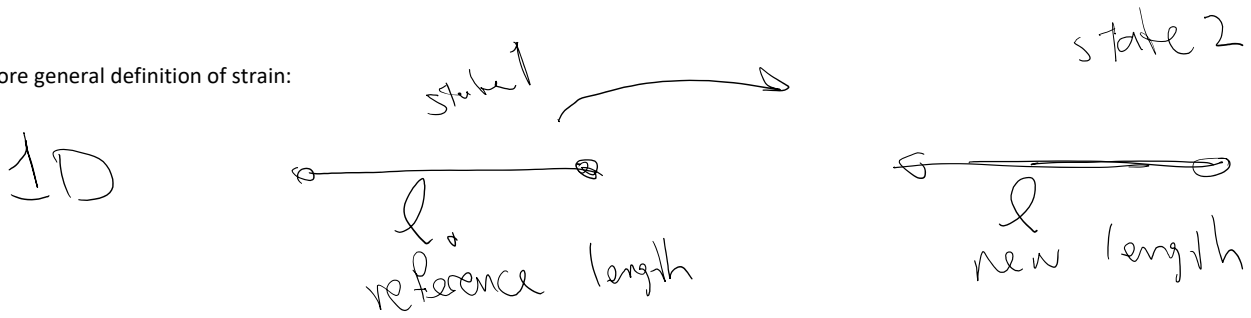
redundant & not included in Voigt notation

engineering shear strains

Voigt notation

1-array (not a vector)

A more general definition of strain:



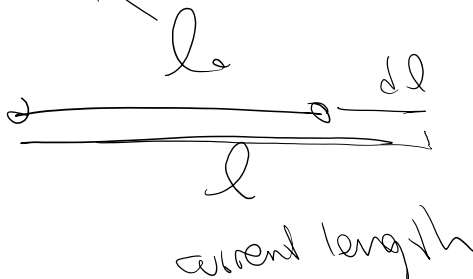
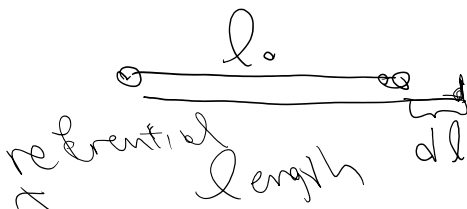
$$\epsilon = \frac{\Delta l}{l_0} = \frac{l - l_0}{l_0}$$

$$\Delta l = l - l_0$$

How about this

$$\epsilon = \int_{l_0}^l d\epsilon = \int_{l_0}^l \frac{dl}{l_0} = \frac{1}{l_0} \int_{l_0}^l dl = \frac{\Delta l}{l_0}$$

$$d\epsilon = \frac{\text{change of length}}{\text{original length}}$$



$$d\epsilon = \frac{dl}{l} = \frac{dl}{\text{current length}}$$

$$\epsilon = \int_{l_0}^l \frac{dl}{l} = \ln\left(\frac{l}{l_0}\right) \quad \text{logarithmic strain}$$

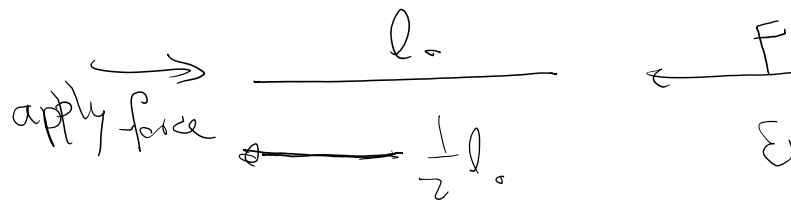
$$\epsilon = \frac{\Delta l}{l_0} \quad \text{change of length normalized by initial length}$$

$$\epsilon = \ln\left(\frac{l}{l_0}\right) = \ln\left(\frac{\Delta l}{l_0} + 1\right) \approx \frac{\Delta l}{l_0} + o\left(\left(\frac{\Delta l}{l_0}\right)^2\right)$$

$$\epsilon = \ln\left(\frac{l}{l_0}\right) = \ln\left(\frac{\Delta l}{l_0} + 1\right) \approx \frac{\Delta l}{l_0} + o\left(\left(\frac{\Delta l}{l_0}\right)^2\right)$$

at every stage of deformation increment of strain is computed using the current length  $\square$

Example

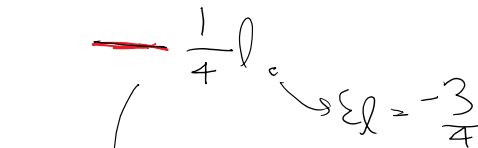


$$\epsilon_{lin} = \frac{\Delta l}{l_0} = \frac{-\frac{1}{2}l_0}{l_0} = -\frac{1}{2}$$

$$\epsilon_{log} = \ln\left(\frac{\frac{1}{2}l_0}{l_0}\right) = \ln\left(\frac{1}{2}\right)$$

$$F = k \epsilon$$

for  $\epsilon_{log}$   $F = k \epsilon$  is OK as  $l \rightarrow 0$



$$\epsilon_{log} = \ln\left(\frac{1}{4}\right)$$

$$\epsilon_{lin} = \frac{\Delta l}{l_0} \text{ as } l \rightarrow 0 = -1$$

$$\epsilon_{log} = \log\left(\frac{l}{l_0}\right) = -\infty$$

But physically

$$F = k \epsilon$$

for  $\epsilon_{lin}$  is a very good

model (need to use a better  $\epsilon \rightarrow F$  const. eqn)

Other strain definitions:

Strain = stretch - 1

rigid  
matrix

0

1

∞ - √

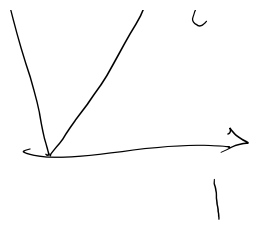
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$$\varepsilon_i = \sqrt{C_{ii}} - 1 = \underbrace{U_{ii} - 1}_{\text{stretch}}$$


$$= \sqrt{\underbrace{1 + 2G_{ii}}_{C_{ii}}} - 1 \approx 1 + \frac{1}{2} 2G_{ii} - 1 = G_{ii}$$

$$\sqrt{1+x} \approx 1 + \frac{1}{2}x + O(x^2)$$

$$\approx G_{ii}$$

$$G = \frac{1}{2}(C - 1)$$

$$= \frac{1}{2}(U^2 - 1)$$

$U - 1$  Exact strain

$G = \frac{1}{2}(U^2 - 1)$  approximate strain

$\ln U$  logarithmic strain

The various measures of Lagrangian strain used in the literature are all related to the stretch  $U$  in a one-to-one manner. Examples include,

$U - I$  (exact) linear strain

$G = \frac{1}{2}(U^2 - I)$  Green strain

$$\frac{1}{m}(U^m - I)$$

generalized Green strain  
 $m$ : nonzero integer

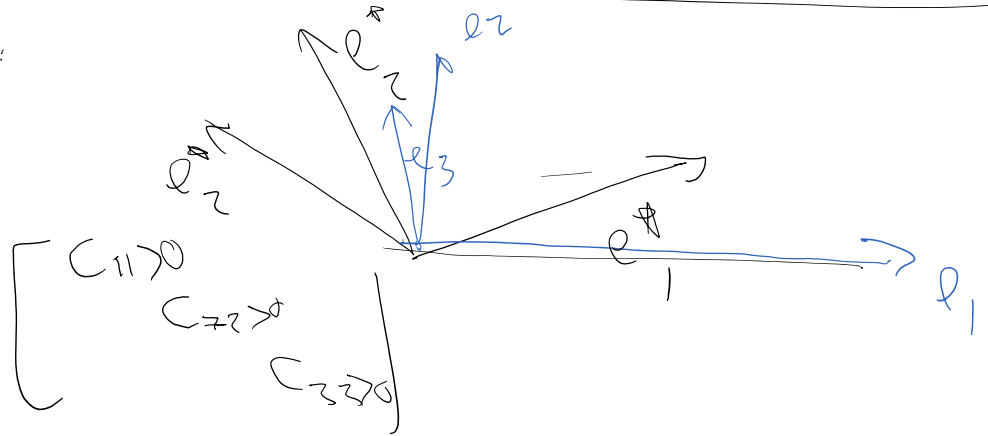
$$\ln U$$

Logarithmic strain  
 (Hencky strain)

Defining  $\ln U$ :

$$C = F^T F$$

$[C]^{\#} =$   
 principal axes of  $C$



$$\begin{bmatrix} C_{11} > 0 & & \\ & C_{22} > 0 & \\ & & C_{33} > 0 \end{bmatrix}$$

$$U = \sqrt{C}$$

$$[U]^{\#} = \begin{bmatrix} \sqrt{\frac{C_{11}}{U_{11}}} & & \\ & \sqrt{\frac{C_{22}}{U_{22}}} & \\ & & \sqrt{\frac{C_{33}}{U_{33}}} \end{bmatrix}$$

$$[\ln U]^{\#} = \begin{bmatrix} \ln U_{11} & 0 & 0 \\ 0 & \ln U_{22} & 0 \\ 0 & 0 & \ln U_{33} \end{bmatrix}$$

Other definitions of strain

$$e(\lambda)$$

$\lambda$  eigenvalues of  $U$

$$e(\lambda) : [\text{strain}]^{\#} = \begin{bmatrix} e(U_{11}) & & \\ & e(U_{22}) & \\ & & e(U_{33}) \end{bmatrix}$$

$$L \quad e(U_{33})$$

Examples

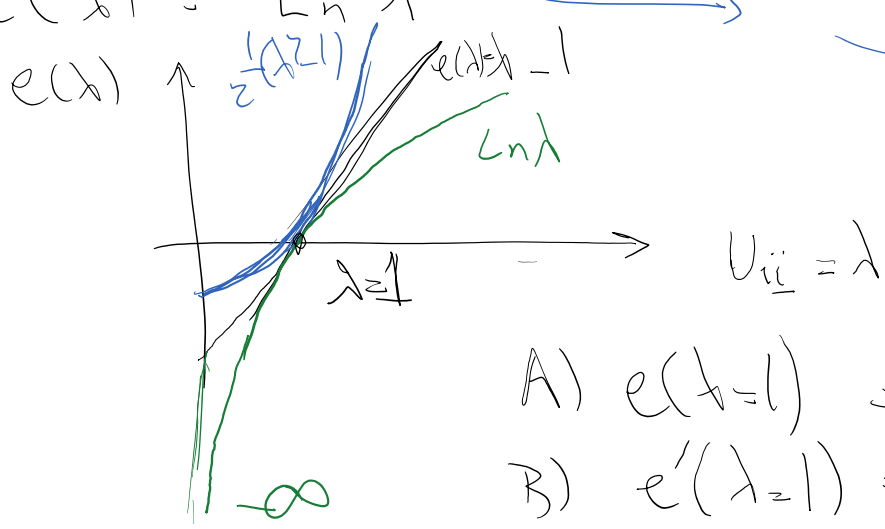
$$e(\lambda) = \lambda - 1 \longrightarrow U - 1$$

$$e(\lambda) = \frac{1}{2}(\lambda^2 - 1) \longrightarrow \frac{1}{2}(U^2 - 1) = G$$

$$e(\lambda) = \frac{1}{m}(\lambda^m - 1) \longrightarrow \frac{1}{m}(U^m - 1)$$

$$e(\lambda) = \ln \lambda \longrightarrow \ln U$$

Corresponding strain tensors



- A)  $e(\lambda=1) = 0$
- B)  $e'(\lambda=1) = 1$
- C) all monotonically increasing  
 $\implies e'(\lambda) > 0$  for all  $\lambda > 0$

$$\underline{\underline{\epsilon}}_r = \underline{\underline{E}} + O(\epsilon^2)$$

next time