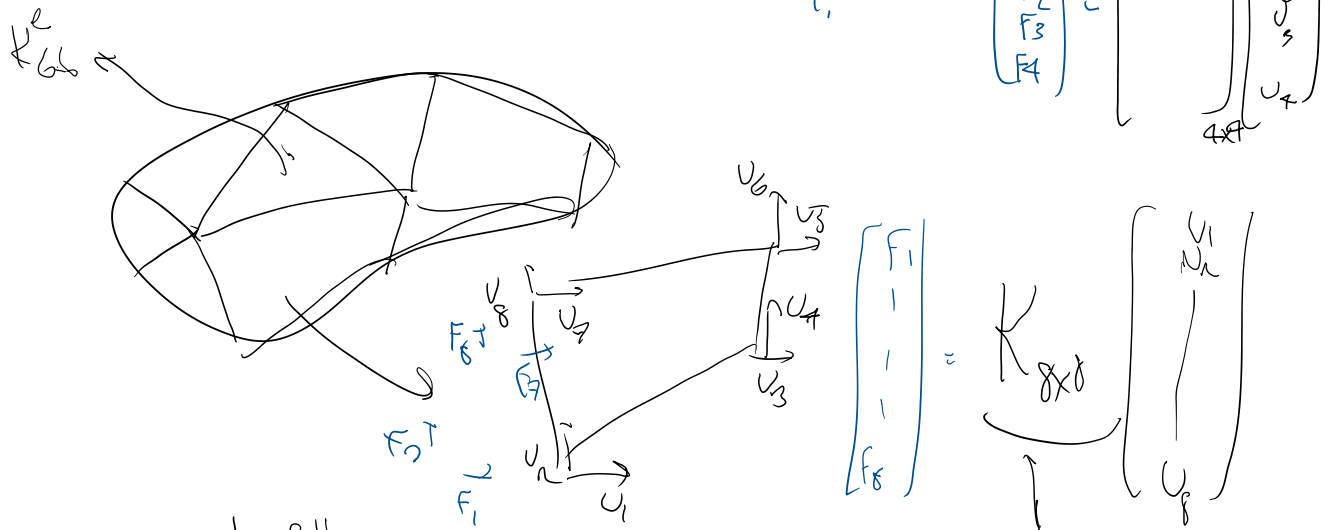
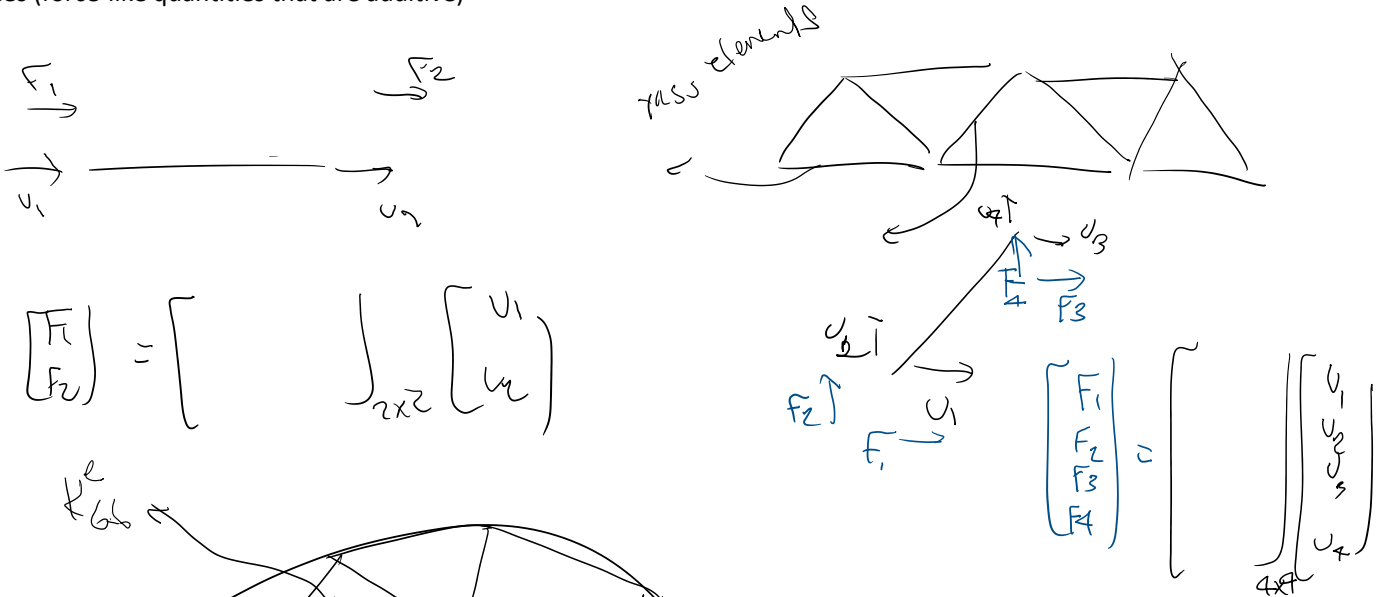


Finite element:

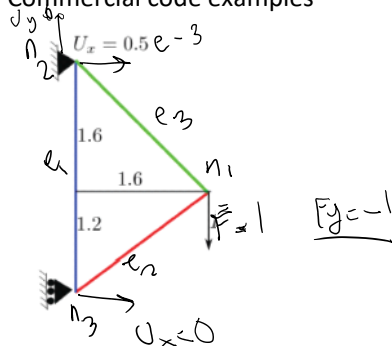
Engineering perspective: It relates nodal displacements (unknowns) to forces (force-like quantities that are additive)



the assembly follows the same process

one place we use can provide specific element stiffness matrix to a program.

Commercial code examples



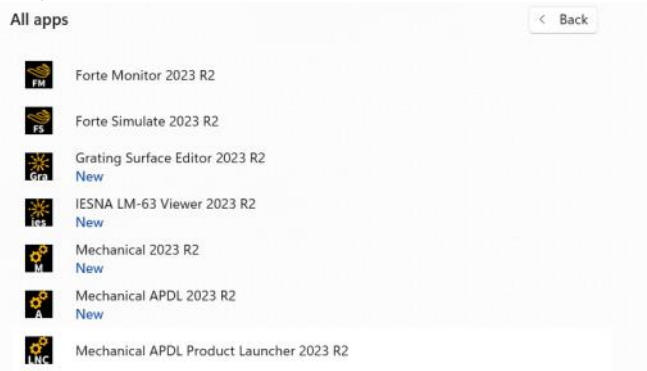
for each element we need Area(A) & elastic modulus E

	E	A
e1	$\frac{100}{E1}$	1 A1
e2	$\frac{100}{E1}$	1 A1
e3	$\frac{1000}{E2}$	10 A2

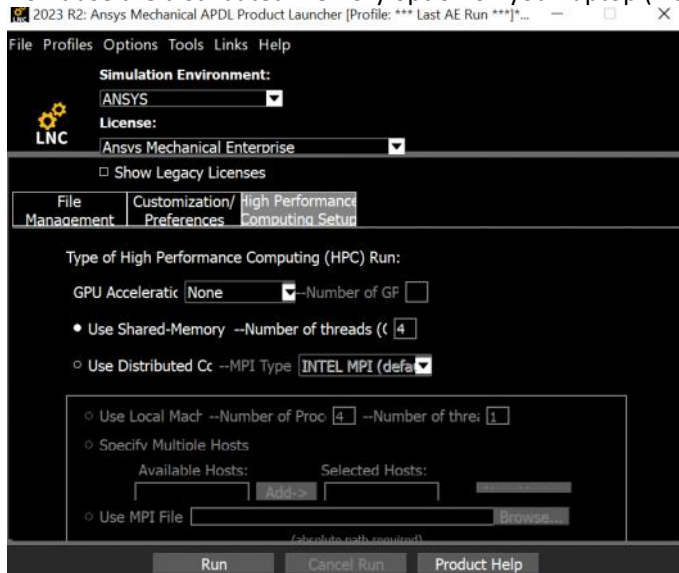
- Ansys:
 - Installation: (free academic version, [link](#)). While this is a limited version, it is sufficient for your project and is recommended due to the ease of installation
 - Make sure in ADPL Launcher you use "Shared Memory" under High Performance Computing Setup.
 - [Link](#) to command lines (Acknowledgment: Matthew Carter).

OR use the program of your choice

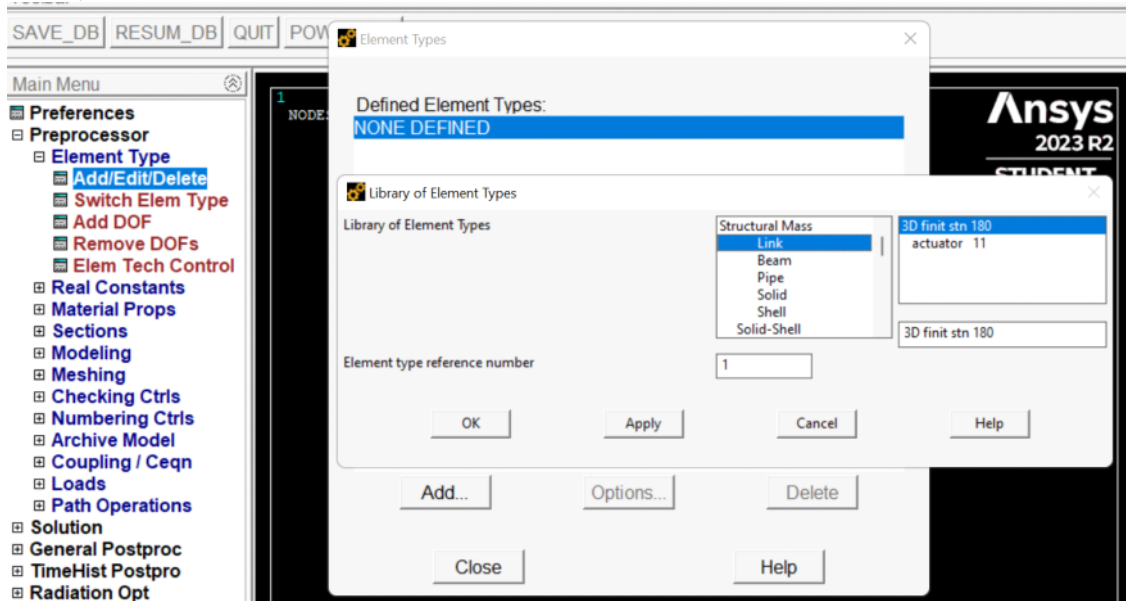
I use Ansys



Don't use the distributed memory option on your laptop (weird crashes)



Define the element types we want to use.
In Ansys a truss element is called "link"

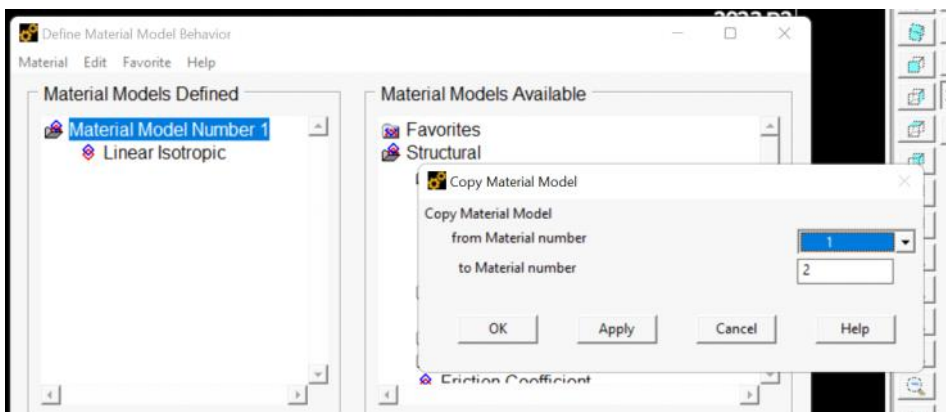
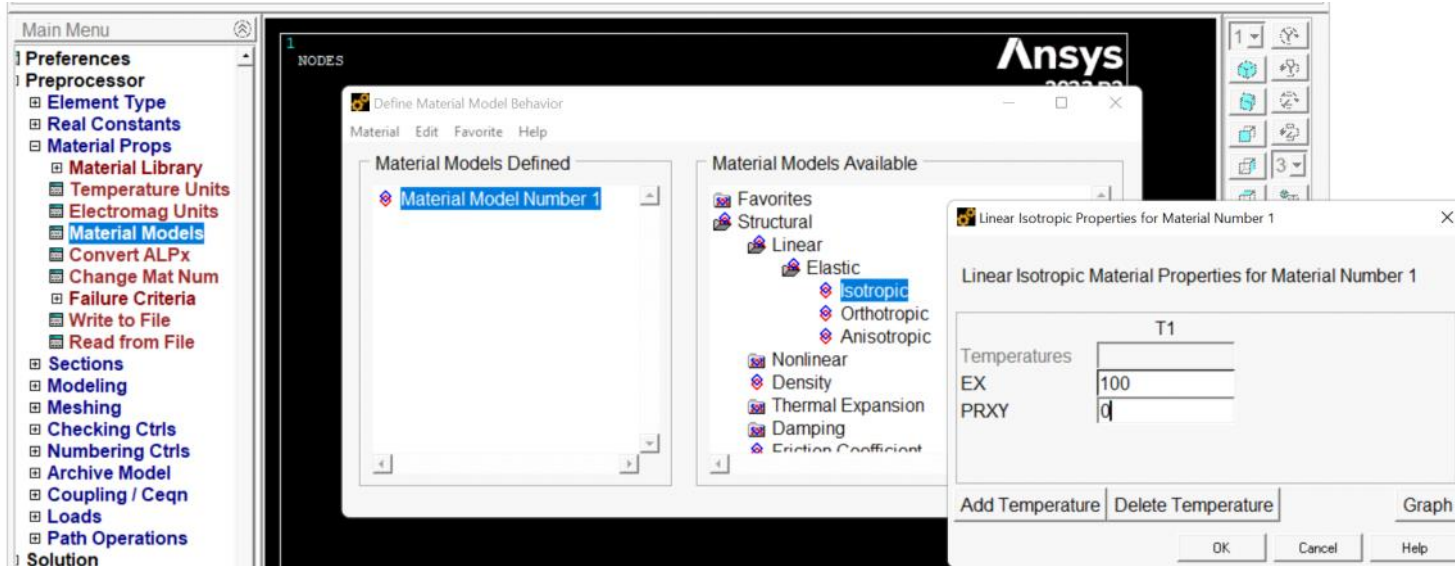


- ▣ General Postproc
- ▣ TimeHist Postpro
- ▣ Radiation Opt

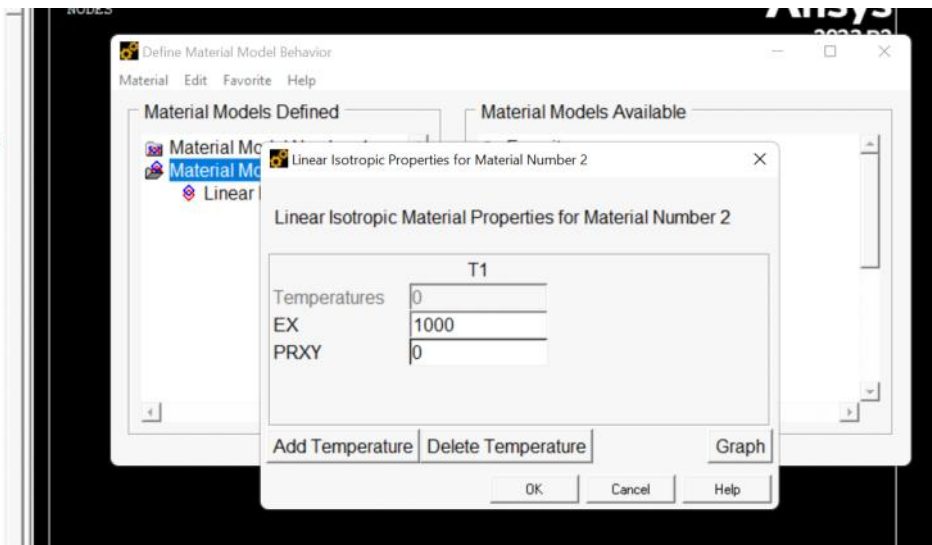


Let's define material properties

	E	A
E_1	$100 \frac{E_1}{E_1}$	$1 \quad A_1$
E_2	$100 \frac{E_1}{E_1}$	$1 \quad A_1$
E_3	$1000 \frac{E_2}{E_2}$	$10 \quad A_2$

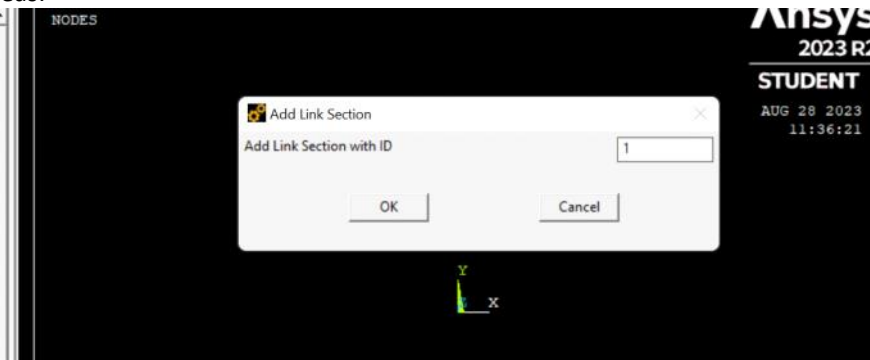


- Preferences
- Preprocessor
 - Element Type
 - Real Constants
 - Material Props
 - Material Library
 - Temperature Units
 - Electromag Units
 - Material Models
 - Convert ALPx
 - Change Mat Num
 - Failure Criteria
 - Write to File
 - Read from File
 - Sections
 - Modeling
 - Meshing
 - Checking Ctrl
 - Numbering Ctrl
 - Archive Model
 - Coupling / Ceqn
 - Loads
 - Path Operations
- Solution
- General Postproc

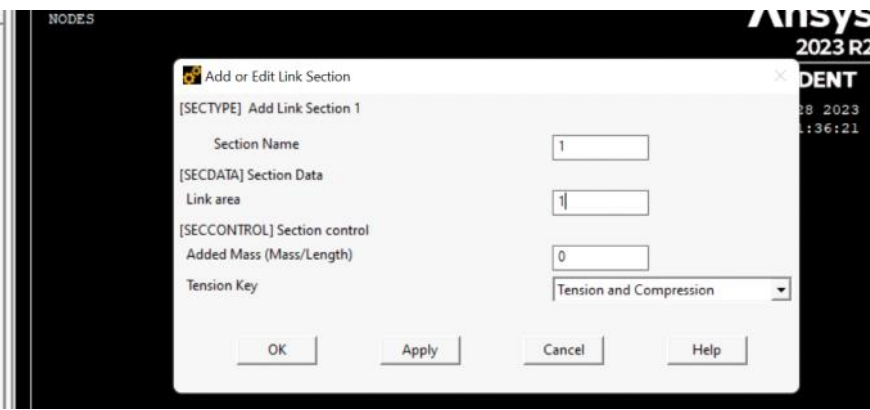


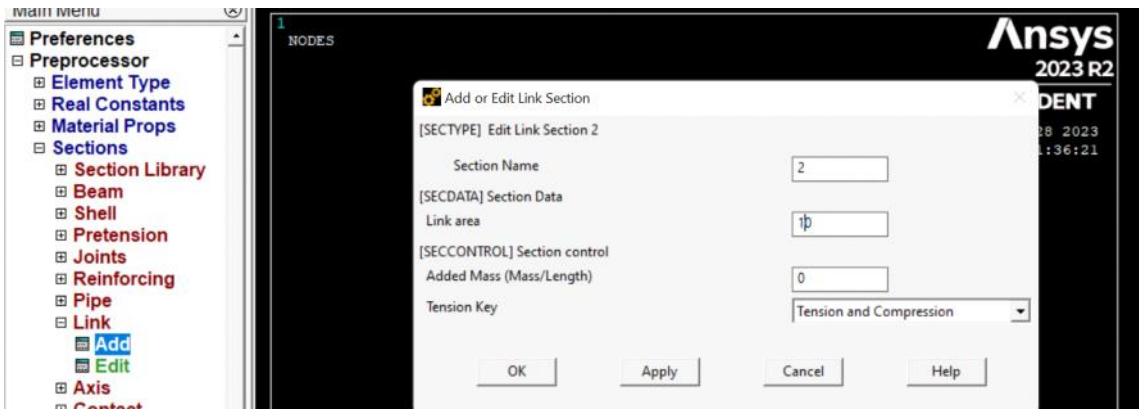
Define the section areas:

- Preferences
- Preprocessor
 - Element Type
 - Real Constants
 - Material Props
 - Sections
 - Section Library
 - Beam
 - Shell
 - Pretension
 - Joints
 - Reinforcing
 - Pipe
 - Link
 - Add
 - Edit

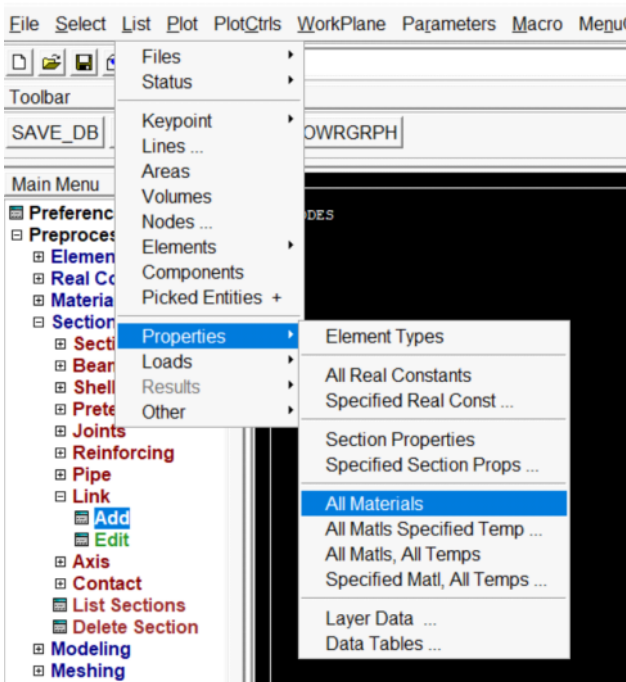


- Preferences
- Preprocessor
 - Element Type
 - Real Constants
 - Material Props
 - Sections
 - Section Library
 - Beam
 - Shell
 - Pretension
 - Joints
 - Reinforcing
 - Pipe
 - Link
 - Add
 - Edit
 - Axis
 - Contact
 - Link Sections





We can check our material and section properties (not necessary)



EVALUATE MATERIAL PROPERTIES FOR MATERIALS 1 TO 2
IN INCREMENTS OF 1

MATERIAL NUMBER = 1 EVALUATED AT TEMPERATURE OF
0.0000

EX = 100.00
PRXY = 0.0000

MATERIAL NUMBER = 2 EVALUATED AT TEMPERATURE OF
0.0000

EX = 1000.0
PRXY = 0.0000

----- same with sections

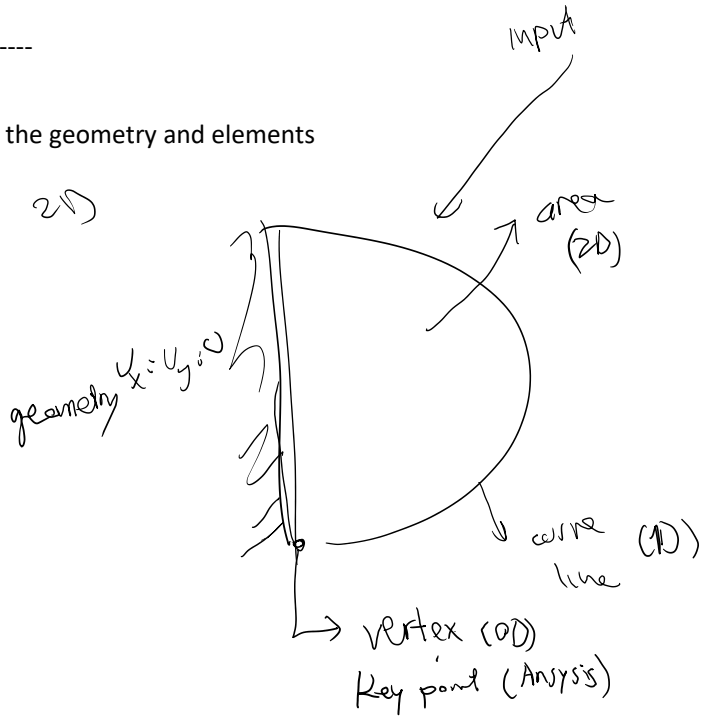
LIST SECTION ID SETS 1 TO 2 BY 1

SECTION ID NUMBER: 1
LINK SECTION NAME IS: 1
Link Area = 1.0000

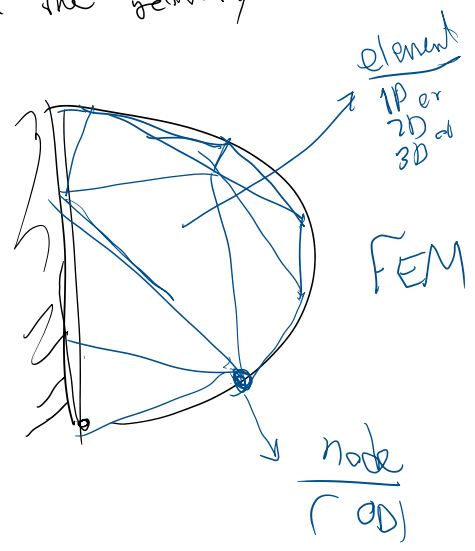
SECTION ID NUMBER: 2
 LINK SECTION NAME IS: 2
 Link Area = 10.000

Create the geometry and elements

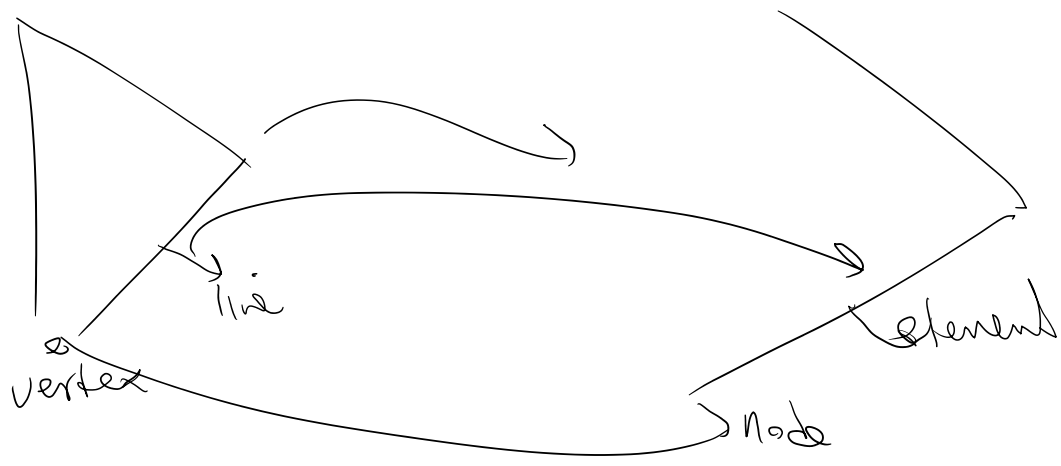
eg 2D



Software "mesh" the geometry

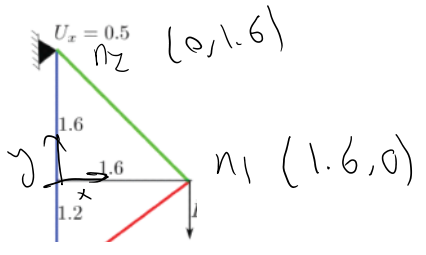


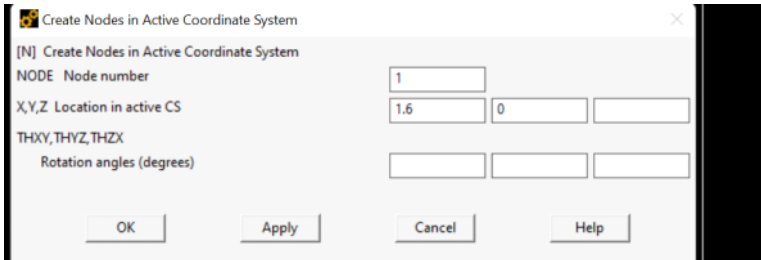
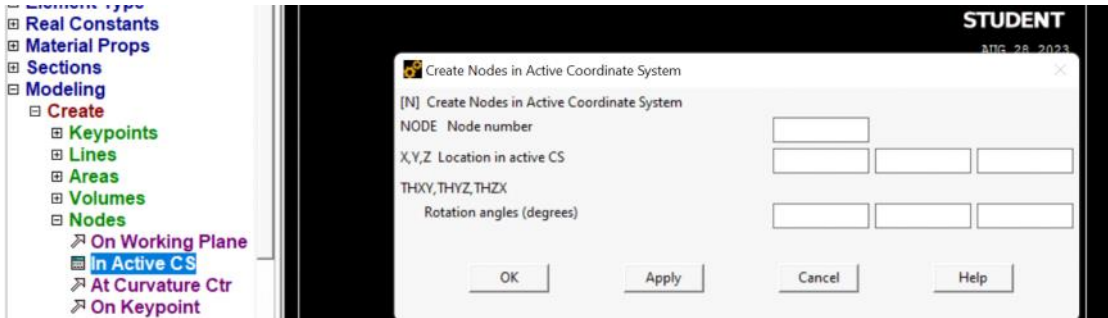
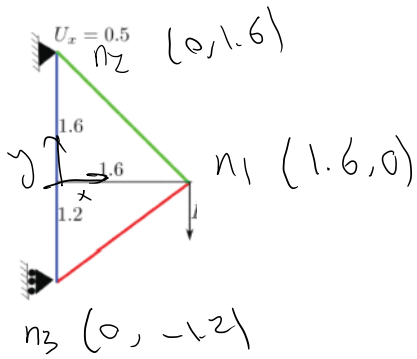
we prefer to provide geometry to the program & ask it to create FEM elements & nodes (mesh it) — we can get closer to exact geometry (2D & 3D)



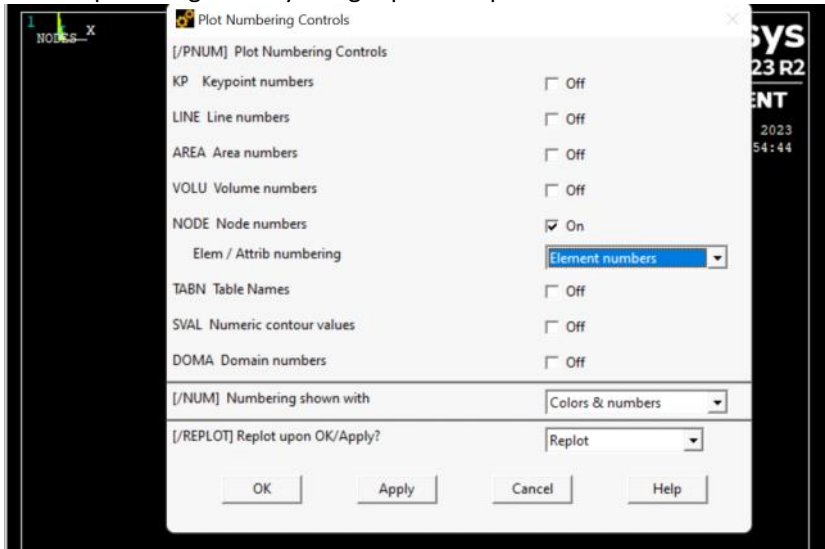
1D elements just directly create FEM objects (nodes & elements)

Create the nodes

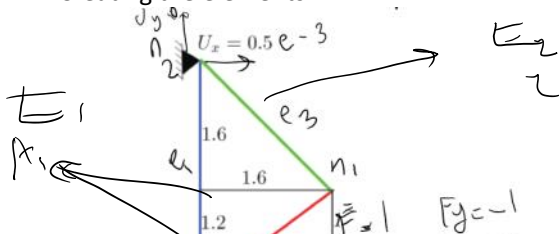




We can plot the geometry using replot and plot controls

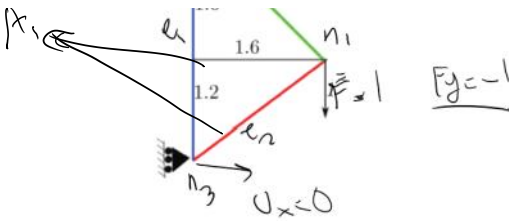


Creating the elements



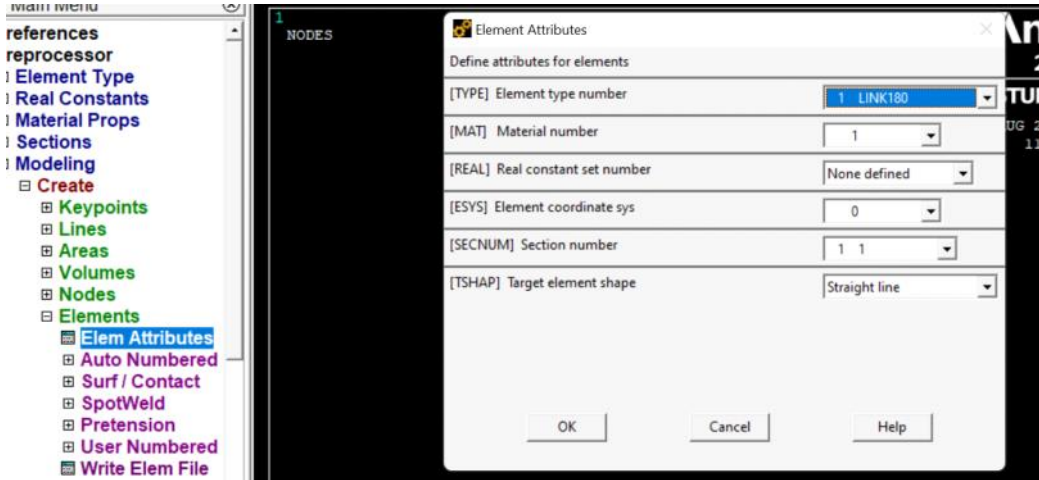
& elastic modulus E

	E	A
e_1	$100 \frac{E_1}{E_1}$	$1 \ A_1$
e_2	$100 \frac{E_1}{E_1}$	$1 \ A_1$

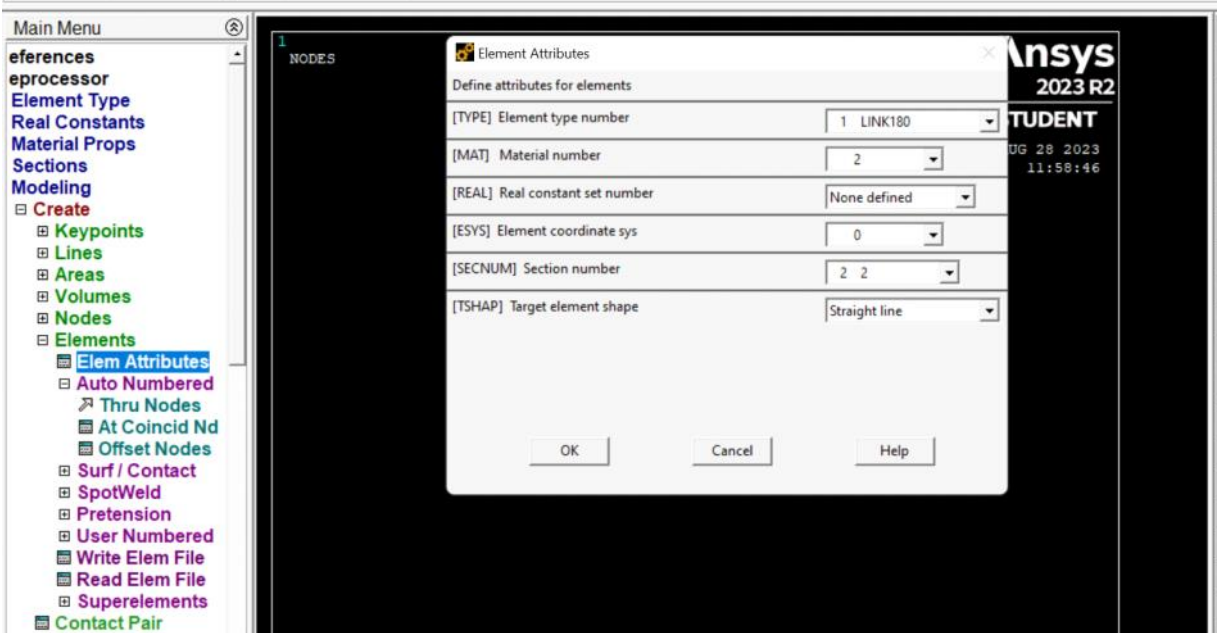


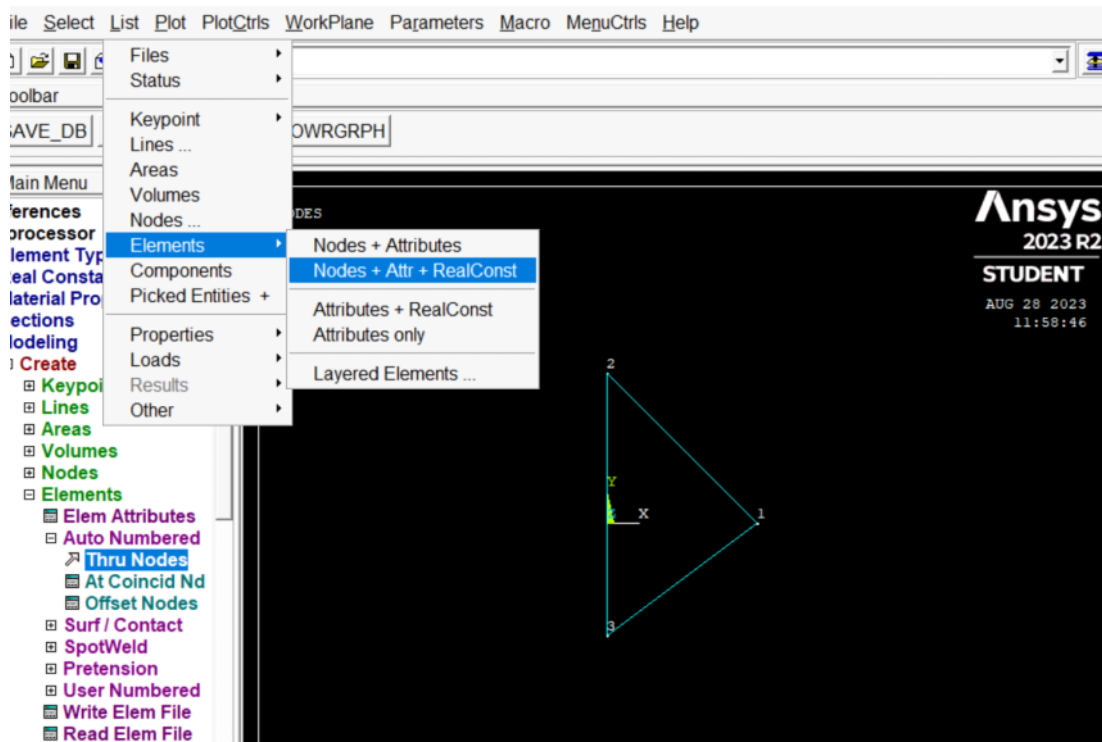
e_2	$100 \frac{E_1}{E_2}$	1	A1
e_3	$1000 \frac{E_2}{E_3}$	10	A2

Check material and section number:



Change E and A for element 3:





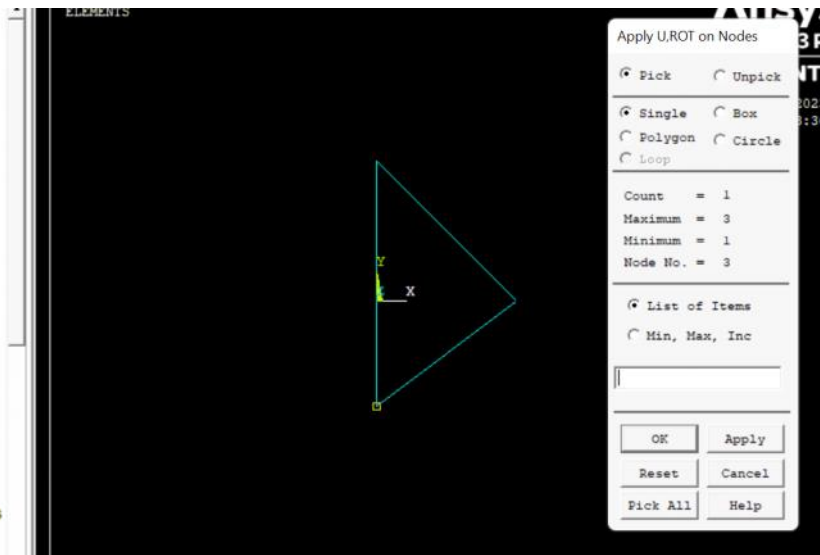
LIST ALL SELECTED ELEMENTS. (LIST NODES)

ELEM MAT TYP REL ESY SEC NODES

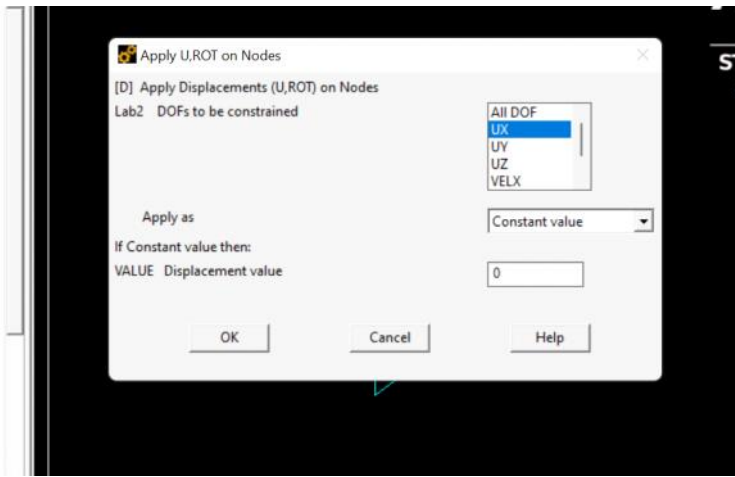
1	1	1	1	0	1	2	3
2	1	1	1	0	1	3	1
3	2	1	1	0	2	2	1

Apply the BCs:

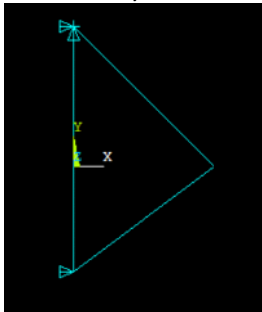
- Preferences
- Preprocessor
 - Element Type
 - Real Constants
 - Material Props
 - Sections
 - Modeling
 - Meshing
 - Checking Ctrls
 - Numbering Ctrls
 - Archive Model
 - Coupling / Ceqn
 - Loads
 - Analysis Type
 - Define Loads
 - Settings
 - Apply
 - Structural
 - Displacement
 - On Lines
 - On Areas
 - On Keypoints
 - On Nodes
 - On Node Components
 - Symmetry B.C.
 - Antisymm B.C.



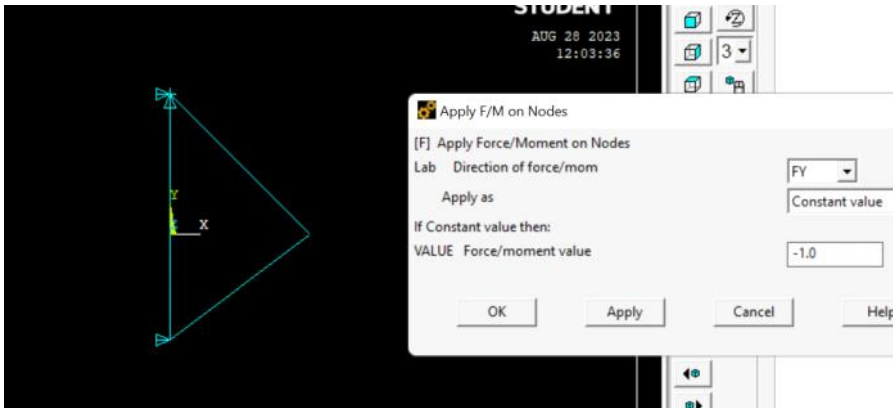
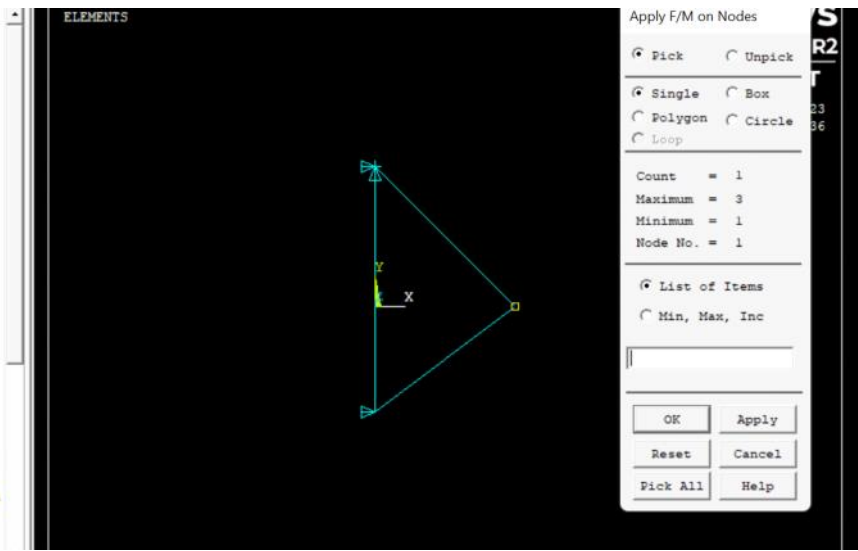
- Preprocessor
 - Element Type
 - Real Constants
 - Material Props
 - Sections
 - Modeling
 - Meshing
 - Checking Ctrl
 - Numbering Ctrl
 - Archive Model
 - Coupling / Ceqn
 - Loads
 - Analysis Type
 - Define Loads
 - Settings
 - Apply
 - Structural
 - Displacement
 - On Lines
 - On Areas
 - On Keypoints



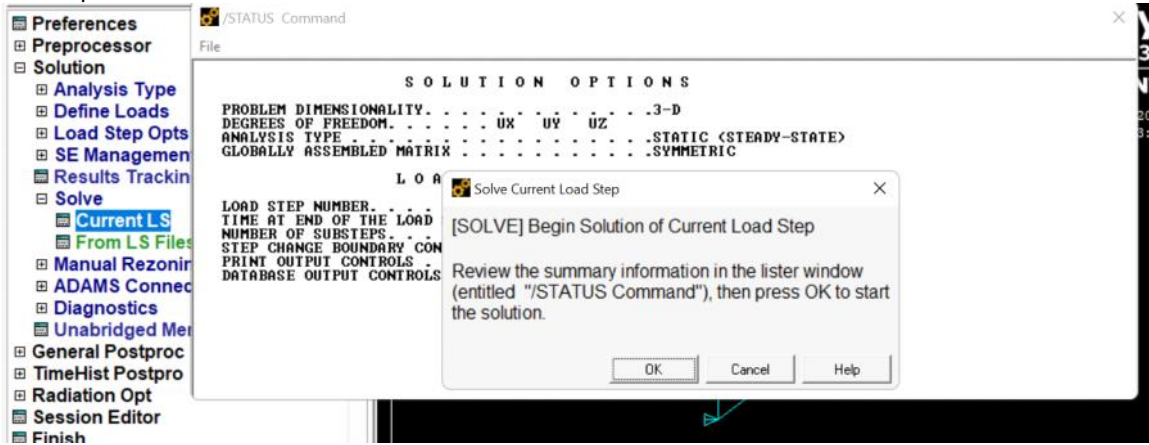
I use $u_x = u_y = 0$ for the top node



- Preferences
 - Preprocessor
 - Element Type
 - Real Constants
 - Material Props
 - Sections
 - Modeling
 - Meshing
 - Checking Ctrl
 - Numbering Ctrl
 - Archive Model
 - Coupling / Ceqn
 - Loads
 - Analysis Type
 - Define Loads
 - Settings
 - Apply
 - Structural
 - Displacement
 - Force/Moment
 - On Keypoints
 - On Nodes
 - On Node Components
 - From Reactions
 - From Mag Analy

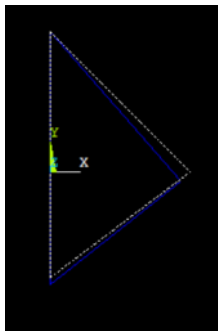
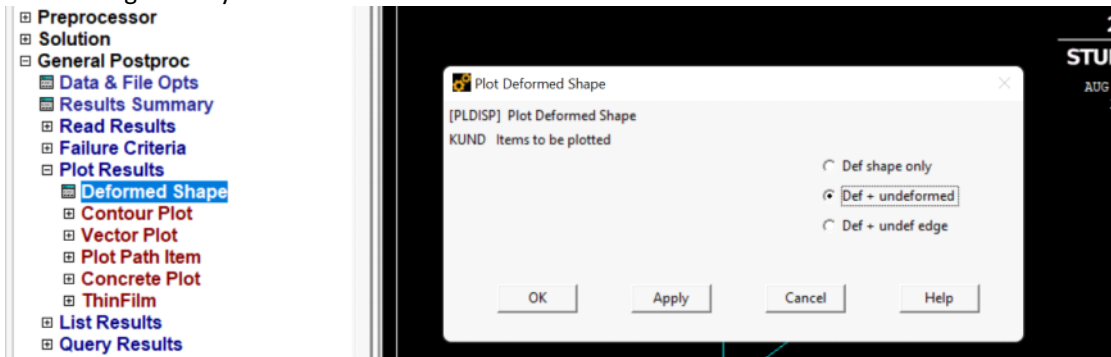


At this point we can solve it

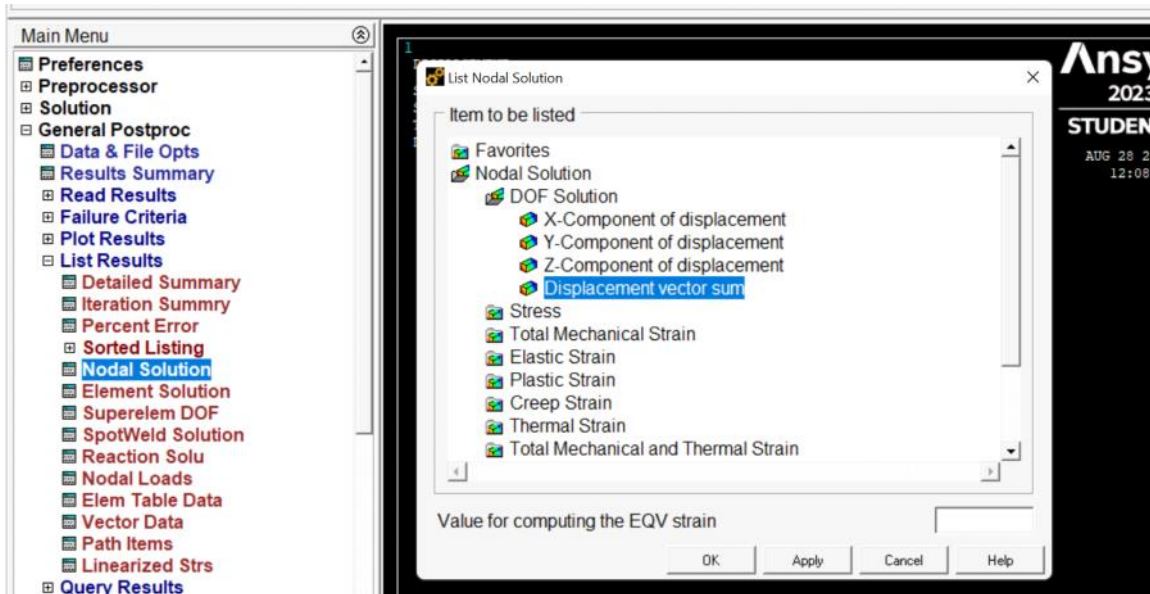


Postprocess (viewing / outputting the results):

1. Deformed geometry



2. Nodal displacements



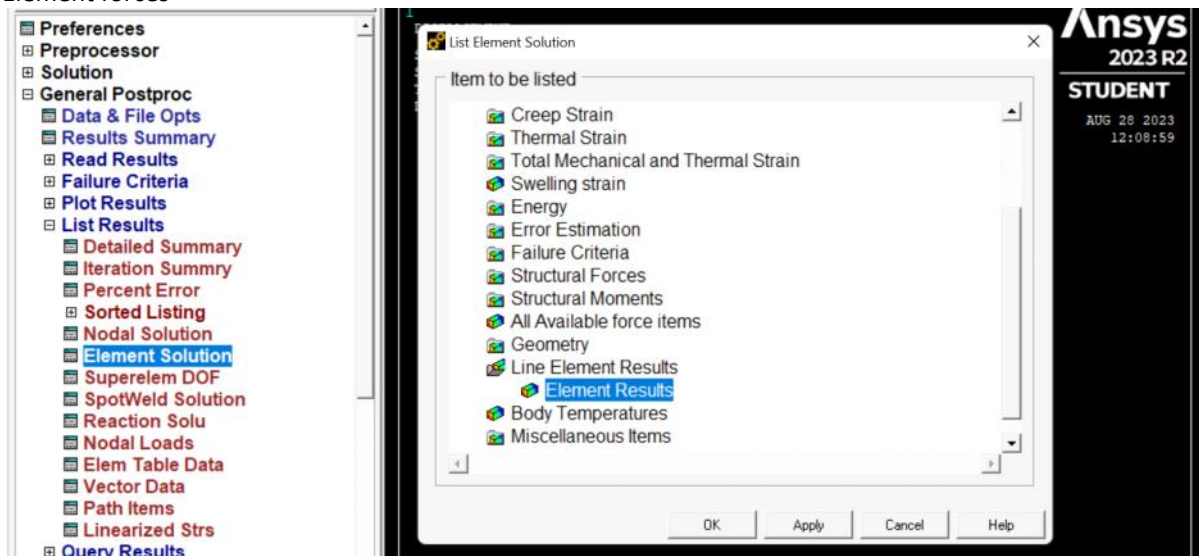
THE FOLLOWING DEGREE OF FREEDOM RESULTS ARE IN THE GLOBAL COORDINATE SYSTEM

NODE	UX	UY	UZ	USUM
1	-0.15236E-001	-0.15495E-001	0.0000	0.21731E-001
2	0.0000	0.0000	0.0000	0.0000
3	0.0000	-0.12000E-001	0.0000	0.12000E-001

MAXIMUM ABSOLUTE VALUES

NODE	1	1	0	1
VALUE	-0.15236E-001	-0.15495E-001	0.0000	0.21731E-001

3. Element forces



LOAD STEP 1 SUBSTEP= 1
 TIME= 1.0000 LOAD CASE= 0

EL= 1 NODES= 2 3 MAT= 1 XC,YC,ZC= 0.000 0.2000 0.000 AREA= 1.0000
 LINK180
 FORCE= 0.42857 STRESS= 0.42857 EPEL= 0.42857E-02
 TEMP= 0.00 0.00 EPTH= 0.0000

EL= 2 NODES= 3 1 MAT= 1 XC,YC,ZC= 0.8000 -0.6000 0.000 AREA= 1.0000

TEMP= 0.00 0.00 EPTH= 0.0000

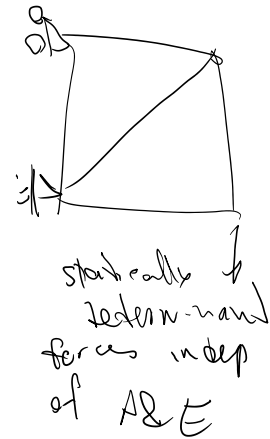
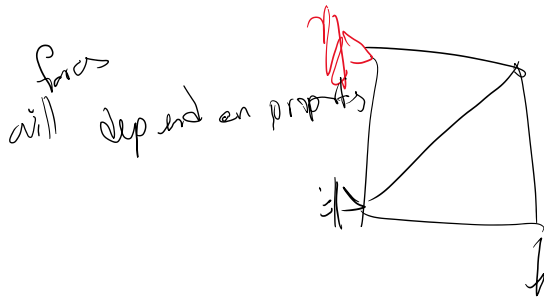
EL= 2 NODES= 3 1 MAT= 1 XC,YC,ZC= 0.8000 -0.6000 0.000 AREA= 1.0000
LINK180

FORCE=-0.71429 STRESS=-0.71429 EPEL=-0.71429E-02
TEMP= 0.00 0.00 EPTH= 0.0000

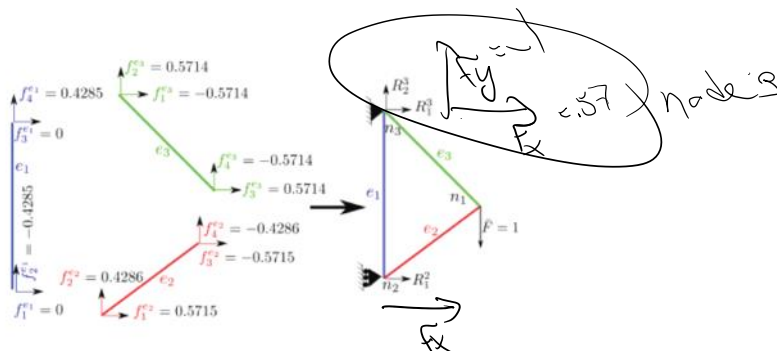
EL= 3 NODES= 2 1 MAT= 2 XC,YC,ZC= 0.8000 0.8000 0.000 AREA= 10.000
LINK180

FORCE= 0.80812 STRESS= 0.80812E-01 EPEL= 0.80812E-04
TEMP= 0.00 0.00 EPTH= 0.0000

T^e	$T^{e1} = \frac{1 \times 1}{2.8} \{ 0 \times (0.5 - 0) + 1 \times (0 + 1.2) \} = 0.4286$	$T^{e2} = \frac{1 \times 1}{2} \{ 0.8 \times (-0.2123 - 0) + 0.6 \times (-3.2928 + 1.2) \} = -0.7128$	$T^{e3} = \frac{1 \times 1}{1.6\sqrt{2}} \{ \frac{1}{\sqrt{2}} \times (-0.2123 - 0.5) - \frac{1}{\sqrt{2}} \times (-3.2928 - 0) \} = 0.8084$
-------	--	---	--



4. Reaction forces

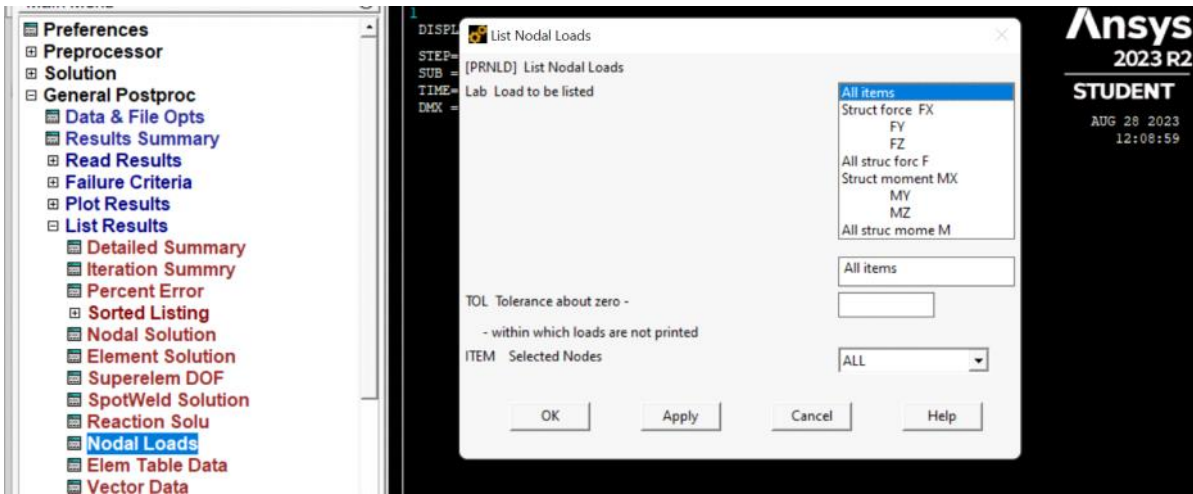


- Also, if we want to double-check our calculations on free dofs. This step is not needed and it may be done as a verification for hand calculations:

$$F_1^1 = f_3^{e2} + f_3^{e3} = -0.5715 + 0.5714 = -0.0001 \quad (398a)$$

$$F_2^1 = f_4^{e2} + f_4^{e3} = -0.4286 + -0.5714 = -1 = \bar{F} \quad (398b)$$

$$R_2^2 = f_2^{e1} + f_2^{e2} = -0.4285 + 0.4286 = 0.0001 \quad (398c)$$



LOAD STEP= 1 SUBSTEP= 1
 TIME= 1.0000 LOAD CASE= 0

THE FOLLOWING X,Y,Z SOLUTIONS ARE IN THE GLOBAL COORDINATE SYSTEM

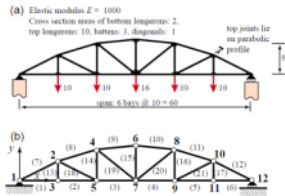
NODE	FX	FY	FZ
1		1.0000	
2	0.57143	-1.0000	
3	-0.57143		

TOTAL VALUES

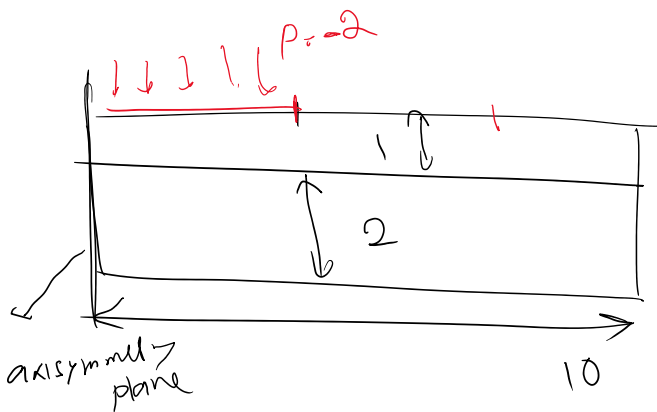
VALUE -0.28866E-014 0.11102E-015 0.0000

References:

[1]. Introduction to Finite Element Methods (ASEN 5007), Fall 2013, Department of Aerospace Engineering Sciences, University of Colorado at Boulder; Part III: Computer Implementation of Finite Elements: Chapter 22 Index. FEM Programs for Trusses and Frames.



A 2D plate example



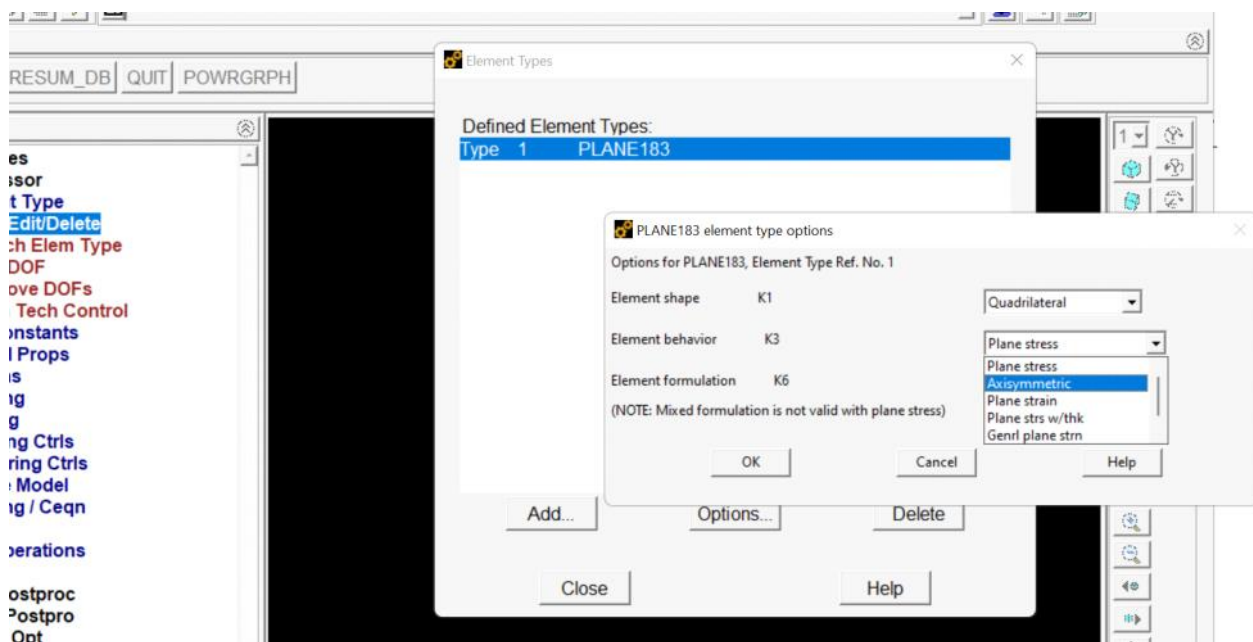
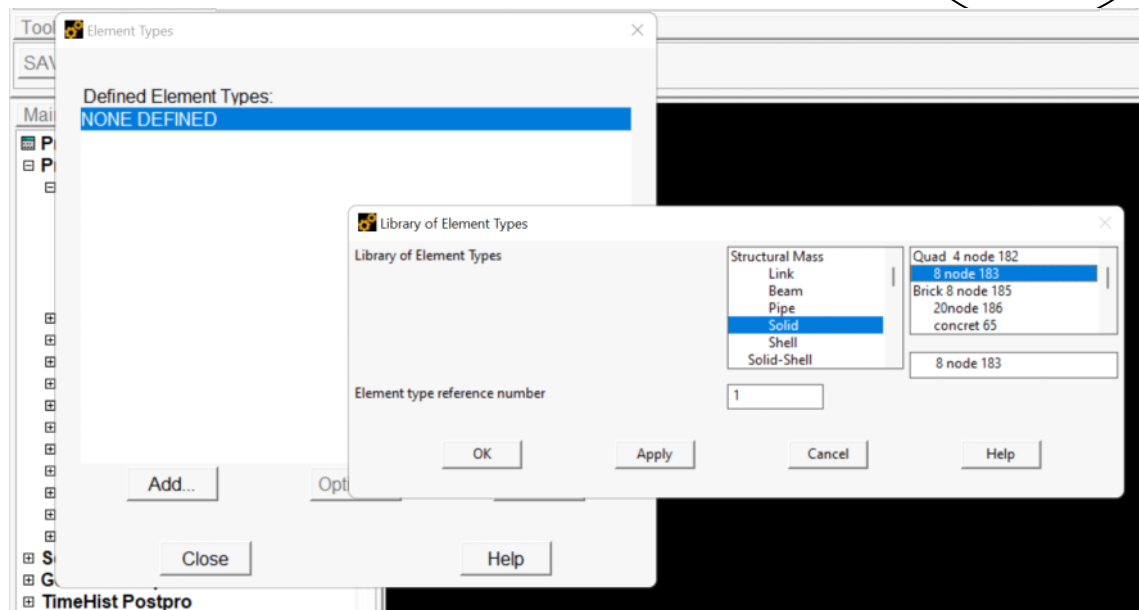
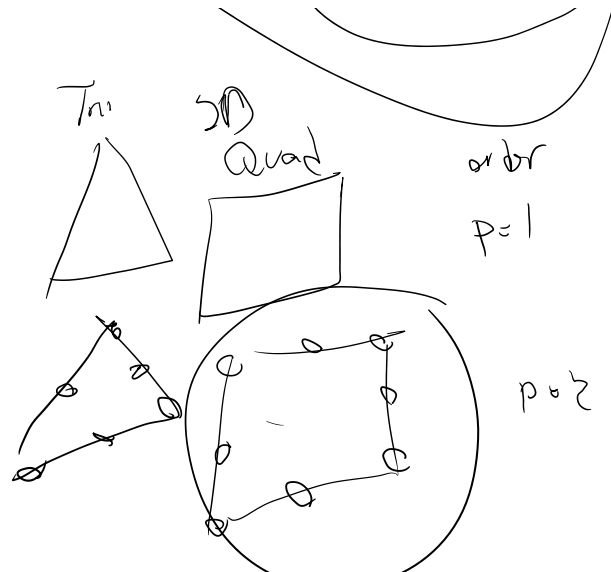
$$E = 100$$

$$E = 1000$$

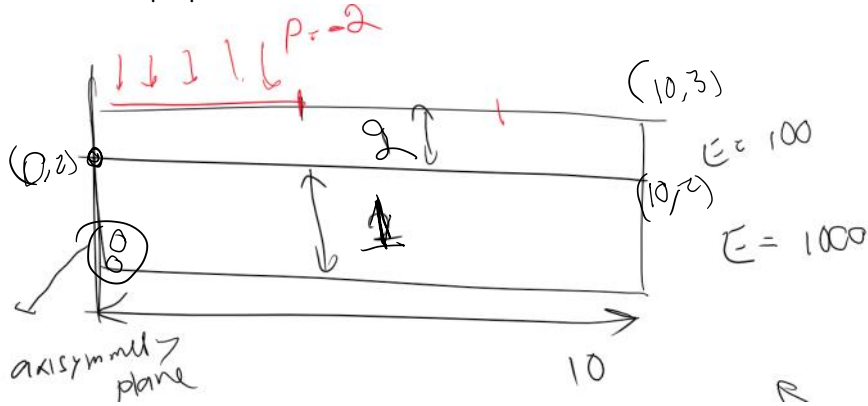
$$U = 0.3$$

T...

Define the element type

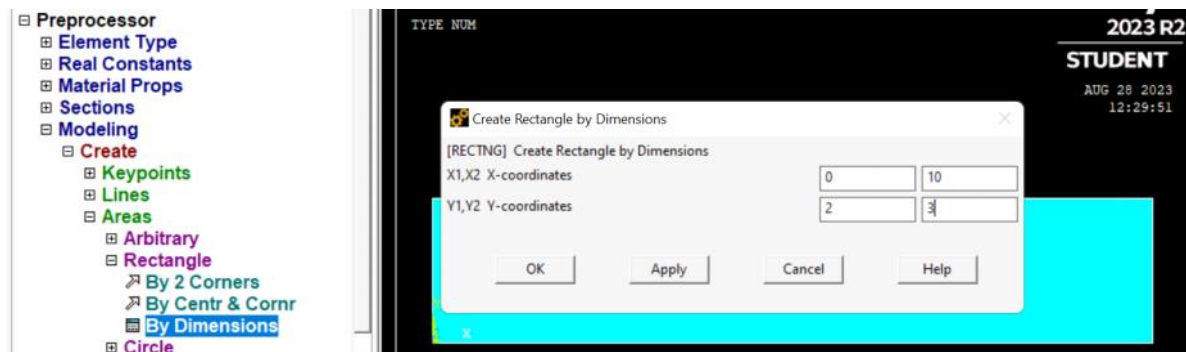
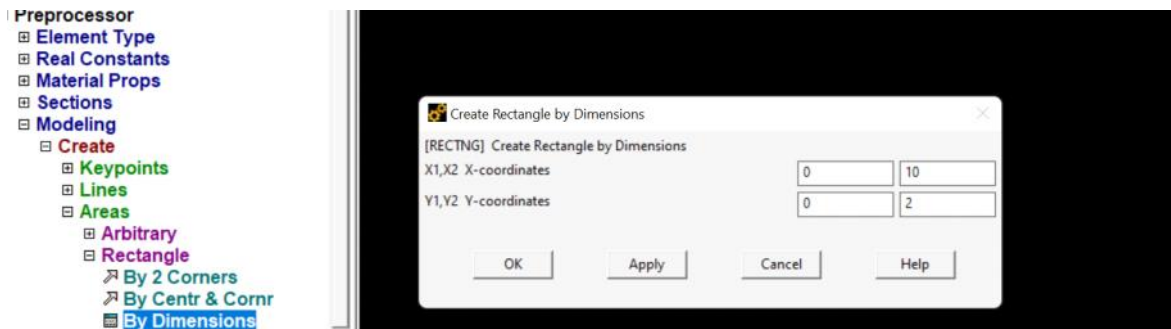
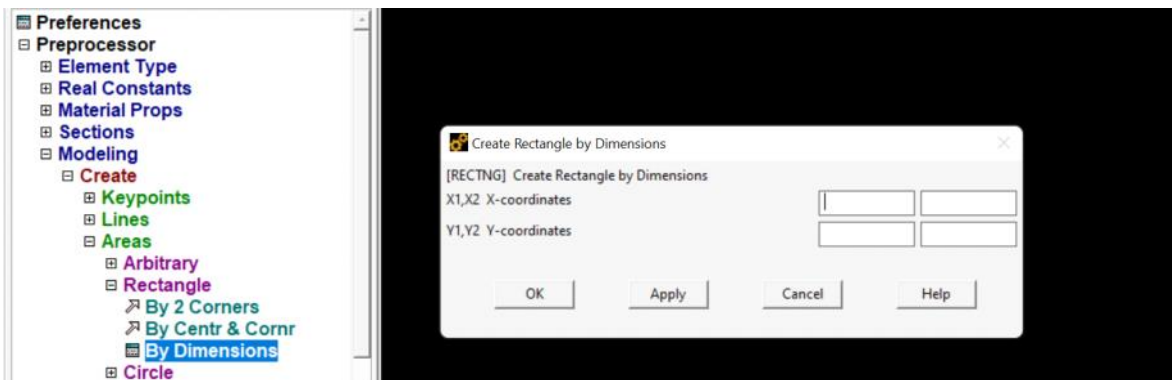


Define material properties

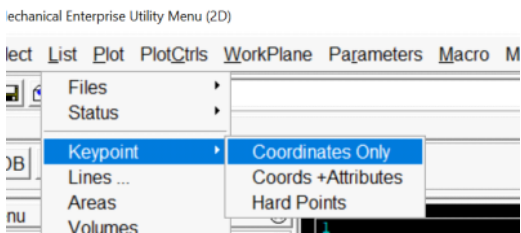
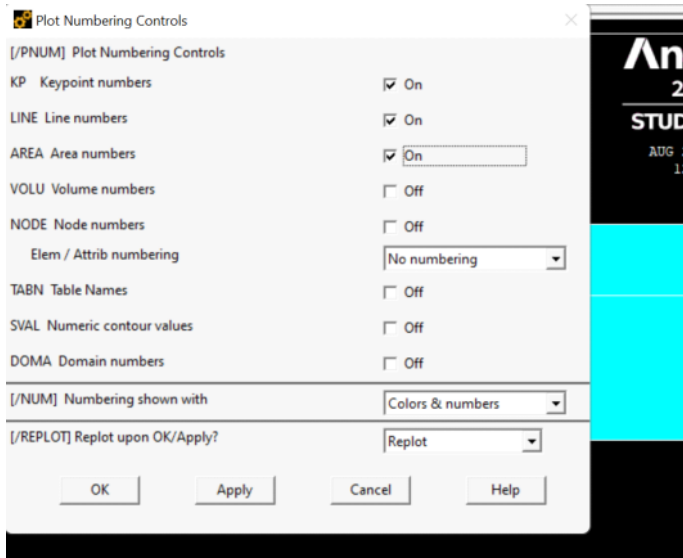
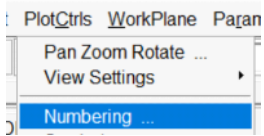


Define the geometry

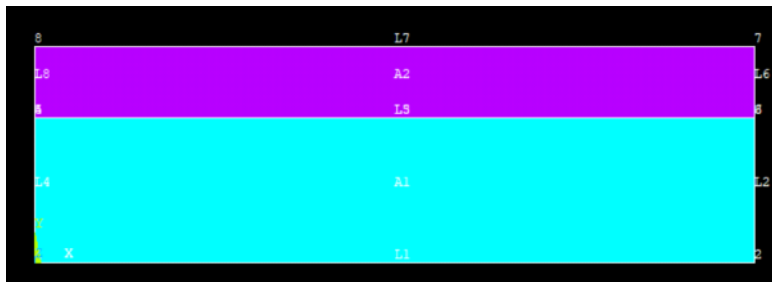
I'm going to create areas and areas create lines and vertices (keypoints) associated with them



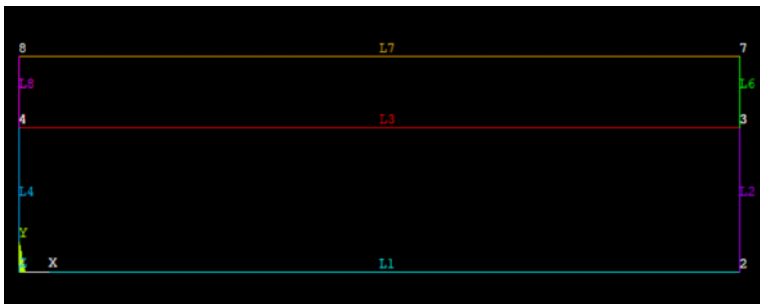
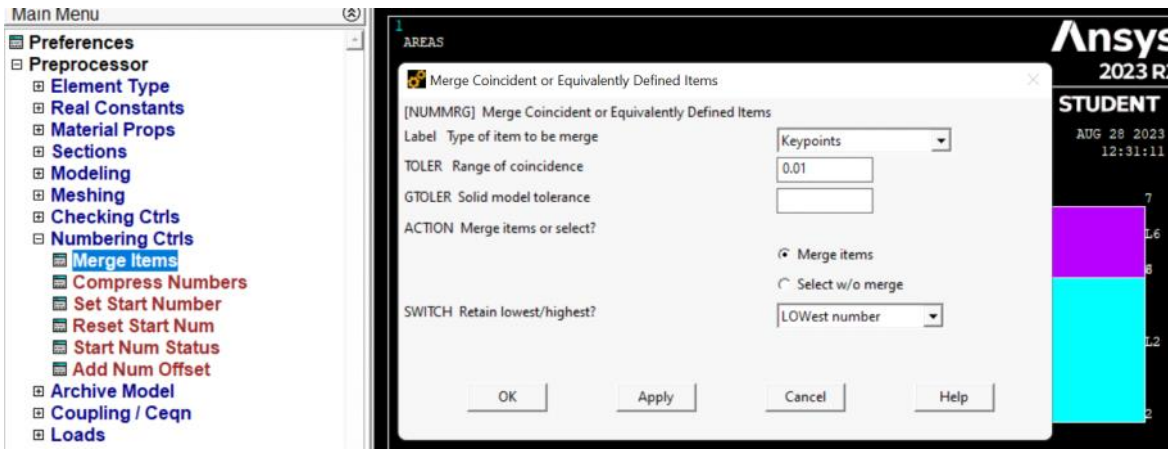
Show all geometry IDs



NO.	X,Y,Z LOCATION			THXY,THYZ,THZX ANGLES		
1	0.000000	0.000000	0.000000	0.0000	0.0000	0.0000
2	10.00000	0.000000	0.000000	0.0000	0.0000	0.0000
3	10.00000	2.000000	0.000000	0.0000	0.0000	0.0000
4	0.000000	2.000000	0.000000	0.0000	0.0000	0.0000
5	0.000000	2.000000	0.000000	0.0000	0.0000	0.0000
6	10.00000	2.000000	0.000000	0.0000	0.0000	0.0000
7	10.00000	3.000000	0.000000	0.0000	0.0000	0.0000
8	0.000000	3.000000	0.000000	0.0000	0.0000	0.0000

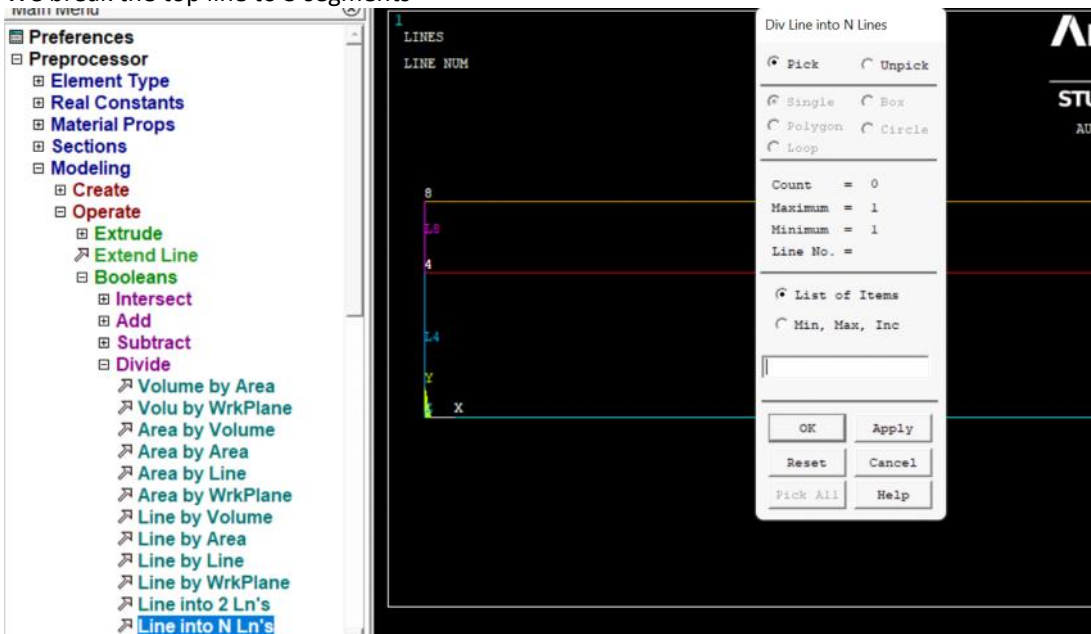


We need to merge the keypoints between the two rectangles so the rectangles are attached



NO.	X,Y,Z LOCATION	THXY,THYZ,THZX ANGLES
1	0.000000 0.000000 0.000000	0.0000 0.0000 0.0000
2	10.00000 0.000000 0.000000	0.0000 0.0000 0.0000
3	10.00000 2.000000 0.000000	0.0000 0.0000 0.0000
4	0.000000 2.000000 0.000000	0.0000 0.0000 0.0000
7	10.00000 3.000000 0.000000	0.0000 0.0000 0.0000
8	0.000000 3.000000 0.000000	0.0000 0.0000 0.0000

 We break the top line to 3 segments



- Menu
- ferences
- processor
- lement Type
- leal Constants
- aterial Props
- ections
- odeling
- 3 Create
- 3 Operate
 - Extrude
 - Extend Line
 - Booleans
 - Intersect
 - Add
 - Subtract
 - Divide
 - Volume by Area
 - Volu by WrkPlane
 - Area by Volume
 - Area by Area

1

ANSYS
2023 R2
STUDENT
AUG 28 2023
12:34:02

LINES

LINE NUM

Divide Line into N Lines

[LDIV] Divide Line into N Lines

NL1 Line to be divided

NDIV No. of lines to create

KEEP Existing line will

OK Apply Cancel Help