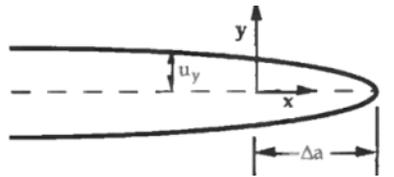


Relation between K and G

K: local behavior (tip stresses)
G: global behavior (energy)



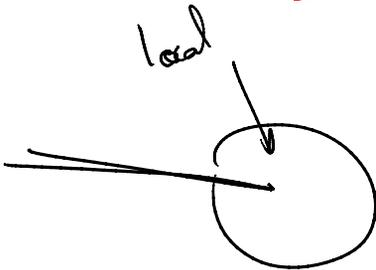
Physical dimension of K:

e.g. $K = \sigma \sqrt{\pi a}$ $[K] = [\sigma] \sqrt{L}$

physical meaning of K :

Local quantity

determines the scaling of stress, displacement, & strain around a crack tip



$$\sigma_{ij}^I = \frac{K^I}{\sqrt{2\pi r}} f_{ij}^I(\theta)$$

G: Energy release rate:

How much energy released per unit area of the crack:

$$[G] = \frac{\text{Energy}}{L^2} = \frac{F \cdot L}{L^2} = \left[\frac{F}{L} \right] L$$

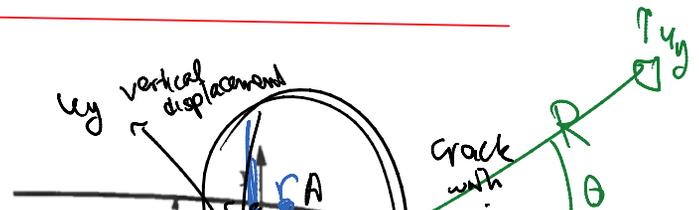
$$\begin{aligned} [G] &= [\sigma] L \\ [K] &= [\sigma] L^{1/2} \end{aligned}$$

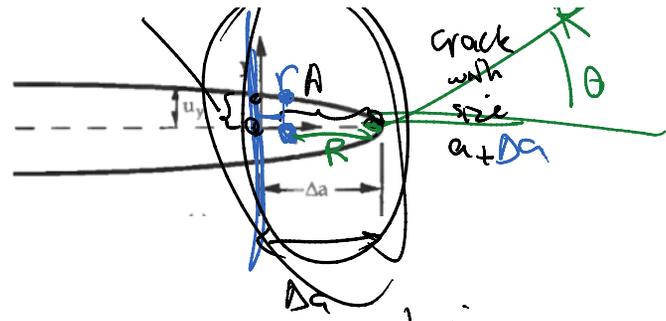
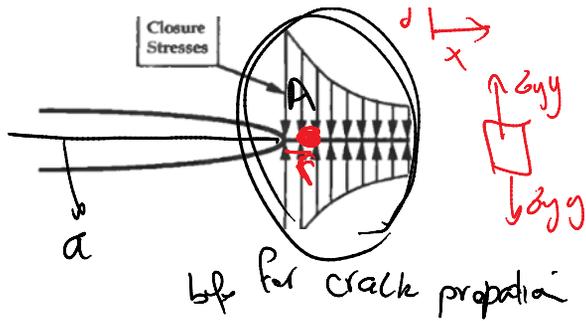
$$[\sigma] L$$

Irwin 1956 $G \equiv -\frac{d\Pi}{dA}$ →

G is a global quantity

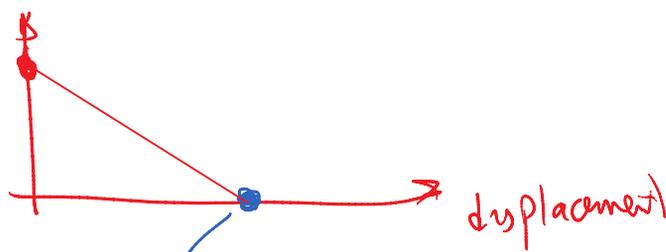
We want to relate G and K





σ_{yy} axis

stress $\frac{K \sqrt{a}}{\sqrt{2\pi r}}$



state 2 for point A (after crack has extended)

$$\frac{K \sqrt{a + \Delta a}}{\sqrt{2\pi}} \sqrt{\frac{\Delta a - r}{2\pi}}$$

crack extension

$$u_y = \frac{K \sqrt{a + \Delta a}}{2\mu} \sqrt{\frac{R}{2\pi}} \left(1 + 2C_0 \frac{\theta}{\pi} \right)$$

for point A

$$\theta_A = \pi$$

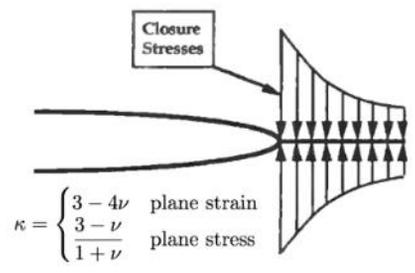
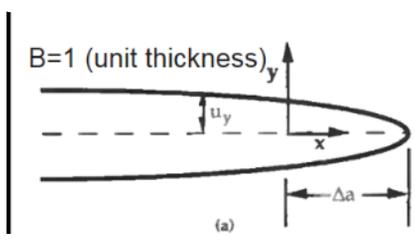
$$R = \Delta a - r$$

$$u_y = \frac{K \sqrt{a + \Delta a}}{2\mu} \sqrt{\frac{\Delta a - r}{2\pi}}$$

The work done at this point is the area under the triangle: 1/2 Height x base

$$1/2 u_y(\text{state 2}) * B dx \sigma_{yy}(\text{state 1}) * 2$$

Half of displacement is going up half is going down



$$\kappa = \begin{cases} \frac{3 - 4\nu}{3 - \nu} & \text{plane strain} \\ \frac{3 - \nu}{1 + \nu} & \text{plane stress} \end{cases}$$

$$dU(x) = 2 \frac{1}{2} \sigma_{yy}(x) u_y(x) dx$$

work of crack closure

$$\Delta U = \int_0^{\Delta a} dU(x)$$

$$u_y = \frac{K_I}{2\mu} \sqrt{\frac{r}{2\pi}} \sin \frac{\theta}{2} \left(\kappa + 1 - 2\cos^2 \frac{\theta}{2} \right), \theta = \pi \Rightarrow$$

$$u_y = \frac{(\kappa + 1)K_I(a + \Delta a)}{2\mu} \sqrt{\frac{\Delta a - x}{2\pi}}$$

$$\sigma_{yy} = \frac{K_I(a)}{\sqrt{2\pi x}}$$

$$G = \lim_{\Delta a \rightarrow 0} \frac{(\kappa + 1)K_I^2}{4\pi\mu\Delta a} \int_0^{\Delta a} \sqrt{\frac{\Delta a - x}{2\pi}} dx$$

In the denominator

After integration we get

$$G = \frac{(\kappa + 1)K_I^2}{8\mu}$$

$$\kappa = \begin{cases} \frac{3-4\nu}{1+\nu} & \text{plane strain} \\ \frac{3-\nu}{1+\nu} & \text{plane stress} \end{cases}$$

Shear modulus

$$\mu = \frac{E}{2(1+\nu)}$$

Scale of K: [stress] * sqrt(L) -> scale of K² = [stress]² L

Scale of mu: [stress]

What is the scale of the RHS: [stress] * L which is what we expected for G

K-G relationship (cont.)

Mode I

$$G_I = \begin{cases} \frac{K_I^2}{E} & \text{plane stress} \\ (1-\nu^2) \frac{K_I^2}{E} & \text{plane strain} \end{cases}$$

Mixed mode

$$G = \frac{K_I^2}{E'} + \frac{K_{II}^2}{E'} + \frac{K_{III}^2}{2\mu} \quad E' = \begin{cases} \frac{E}{1-\nu^2} & \text{for plane strain} \\ E & \text{for plane stress} \end{cases}$$

We know a crack can grow when G = G_c

At the same time for pure mode I we know

$$G = K^2 / E$$

Can we define a critical stress intensity factor?

G = G_c crack can grow

$$G = K^2 / E = G_c = K_c^2 / E \Rightarrow K_c = \sqrt{G_c E}$$

So if we have the value of K we can decide if the crack grows by

- 1) Directly comparing K with K_c (K_c = sqrt(G_c E))

2) Compute $G = K^2 / E'$ and then compare G with G_c ®

How does the evaluation of K come to use in practice?

K as a failure criterion

Failure criterion $K = K_c$ $f(a/W)\sigma\sqrt{\pi a} = K_c$

Basically we have stress, crack length a , and K_c
If we have 2 we can solve for the third one

In practice we have two available and want to find the third one

- **Problem 1:** given crack length a , compute the maximum allowable applied stress

$$\sigma_{\max} = \frac{K_c}{f(a/W)\sqrt{\pi a}}$$

K_c is known (material is given)
Crack length is known

Question: How much load can this specimen withstand before crack can propagate

How much load can we apply before the crack propagates



Material is known -> K_c

- **Problem 2:** for a specific applied stress, compute the maximum permissible crack length (critical crack length)

$$f(a_c/W)\sigma\sqrt{\pi a_c} = K_c \rightarrow a_c$$

This is the case that the loading is known (σ)
Material is known (K_c)

We are looking for a critical crack length a_c

If we replace / fix a part when the crack length is smaller than a_c then we are safe

Because a_c appears in f and $\sqrt{\pi a_c}$ we may need to solve a nonlinear equation

Problem 3: compute K_c provided crack length and stress at fracture

$$K_c = f(a_c/W)\sigma\sqrt{\pi a_c}$$

This case load and a crack length (for example the resolution of monitoring system) are known -> Looking for a material than can take such loading.

--- In the following example right now we are not going to talk about yield stress part (plasticity) And will solve it only for steel (first row)

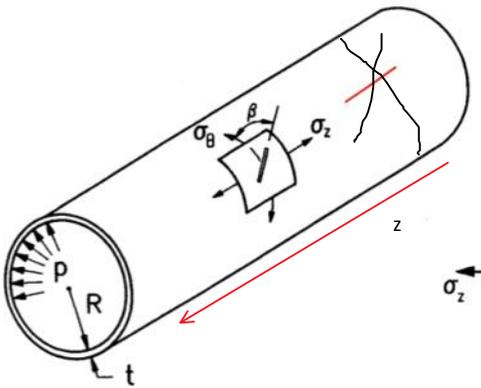
Example

A cylindrical pressure vessel with closed ends has a radius $R = 1$ m and thickness $t = 40$ mm and is subjected to internal pressure p . The vessel must be designed safely against failure by yielding (according to the von Mises yield criterion) and fracture. Three steels with the following values of yield stress σ_Y and fracture toughness K_{Ic} are available for constructing the vessel.

Steel	σ_Y (MPa)	K_{Ic} (MPa \sqrt{m})
A: 4340	860	100
B: 4335	1300	70
C: 350 Maraging	1550	55

Fracture of the vessel is caused by a long axial surface crack of depth a . The vessel should be designed with a factor of safety $S = 2$ against yielding and fracture. For each steel:

- R = 1 m
- t = 40 mm
- p = internal pressure
- $K_{Ic} = 100$ Mpa sqrt(m) critical K for mode I



$$K_I = \frac{pR}{2t} \sqrt{\pi a} (1 + \sin^2 \beta)$$

$$K_{II} = \frac{pR}{2t} \sqrt{\pi a} \sin \beta \cos \beta$$

$$\sigma_z = \frac{pR}{2t}$$

$$\sigma_\theta = \frac{pR}{t}$$

If we only restrict ourselves to mode I fracture (not worry about KII) beta = 90 degrees (e.g. the red crack in the figure) has the most critical angle

$K = pR/t \sqrt{\pi a}$ for a crack that this along the z direction

(a) Plot the maximum permissible pressure p_c versus crack depth a_c ;

$$K = pR / t \sqrt{\pi a}$$

$$K = K_{Ic}$$

$K_{Ic} = p_c R / t \sqrt{\pi a_c}$ K_{Ic}, R, t are known \rightarrow we want to obtain a relation between a_c (critical crack length) and p_c (critical pressure)

$$K_{Ic} = 100 \text{ Mpa} \sqrt{\text{m}}$$

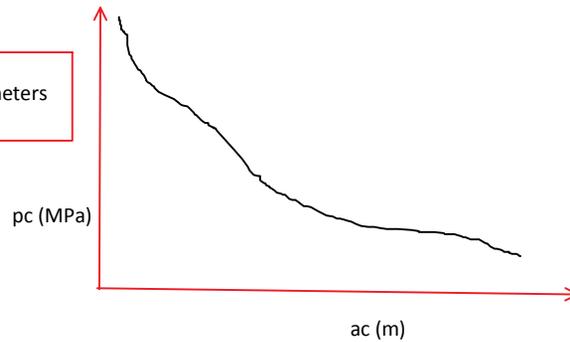
$$R = 1 \text{ m}$$

$$t = 40 \text{ mm} = 0.04 \text{ m}$$

$$\pi = 3.141592653589793$$

$$100 \text{ Mpa} \sqrt{\text{m}} = p_c (1\text{m}) / (0.04 \text{ m}) \sqrt{3.14159 a_c}$$

$$p_c \text{ (Pa)} = 4e6 / \sqrt{a_c} \quad - a_c \text{ in meters}$$



For very small crack lengths for this problem yielding will be the mechanism that causes the failure of this pressure vessel

In general we should design against all failure modes:

Fracture, ultimate stress, buckling, ...

(b) Calculate the maximum permissible crack depth a_c for an operating pressure $p = 12 \text{ MPa}$;

$$p_c \text{ (Pa)} = 4e6 / \sqrt{a_c} \rightarrow p_c \cdot \sqrt{a_c} = 4e6 \quad \text{put } p = 12e6 \rightarrow$$

$$\sqrt{a} = 1 / 3 \rightarrow a = 1/9 = 0.1111 \text{ m} = 11\text{cm}$$

When the crack length gets close to this value we must fix it

(c) Calculate the failure pressure p_c for a minimum detectable crack depth $a = 1 \text{ mm}$.

161

$$P = 4e6 / \sqrt{1e-3\text{m}} \Rightarrow p = 126 \text{ Mpa}$$

This 1 mm can be the resolution of monitoring system so we use that to calculate how much load (pressure) the pressure vessel can take safely