G in terms of compliance **Fixed** grips **Fixed** loads $G = \frac{1}{2B} \frac{u^2}{C^2} \frac{dC}{da} = \frac{1}{2B} P^2 \frac{dC}{da}$ $G = \frac{1}{2B}P^2 \frac{dC}{da}$ for general case $G = \frac{1}{2B}P^2 \frac{dC}{dG}$ C: compliance = U 20 planar crack growth P $\begin{array}{c} \Delta_{1}, \quad \frac{PA^{3}}{3EI} \rightarrow C = \frac{\Delta_{1}}{P} = \frac{2a^{3}}{3EI} \end{array}$ N $G = \frac{1}{2B} p^2 \frac{dC}{d} \rightarrow$ G = P2a2 ET for this poolier 9 Andor example P. CRCP3 a G = 72 3 2 1= 63 61 8 6,26262 G(OotBa)>R(a+ba), G(Pini) Grache grovus Crack growth for the cantilever geometry (load control) D R 26 Ç R. R(9+13a) plane strain Do $- \begin{bmatrix} G(a_0 + Da) & > R_1(a_z Da) \\ G(a_0) & = R_1(a_0) \end{bmatrix}$

 $\lim_{N \to 0} \frac{G(\alpha_{\circ} + D\alpha) G(\alpha_{\circ})}{D\alpha} \geq R_{1} \frac{(\alpha_{\circ} + D\alpha) - R_{1}(\alpha_{\circ})}{D\alpha}$ $\frac{dC_{e}(a_{o})}{dc_{o}} \geq \frac{dR_{e}(a_{o})}{dc_{o}}$ then the crack gions GrR G(a) = R(a)G(a) SR(a) Crack Grows G-R G17R Oracic unstably grows R ~ Plane $G_{i}(o_{r}) = R(o_{r})$ G(R) Pi>P, G(P) OrR Gloging Riazy $G(\alpha_1) \in R(\circ)$ R(a)processione forms doesn't _ Cilaik phnessiress grow usta) ?! $C_1(\sigma_0) \langle R'(\sigma_0) \rangle$ and at early stages of cerack growl Э_с 00 resistance mechanisms are activated -> are need to increase the load for the crack grow stably Post P2 Grack grows unstably - for constant P2 crack grows unbouldably in a dynamic fashion Displacement control:

Instead of applying a known load at the end points, we pull it with specified displacement







For displacement-control loading, even if R is constant (e.g. plane strain condition), G' < R' so the crack does not grow in an unstable manner. We need to keep increasing Δ to have a continued crack growth.

4.2. Stress solutions, Stress Intensity Factor K (SIF)

Solving elastodynamic / elastostatic problem:

1. Displacement-based approach

V.6 \$ p520

Constitutive eqn d=



3 ther elosticity densor

Example: For isotropic material
$$\delta_{ij} = (\lambda \delta_{ij} \delta_{kl} + M(\delta_{il} \delta_{jk} + \delta_{ik} \delta_{jl})) \epsilon_{kl}$$

Lame's parameters

Isotropic
3D
$$T_{ij} = \lambda \delta_{ij} E_{kk} + 2\mu E_{ij} \quad \text{or} \quad \mathbf{T} = \lambda \mathbf{I}_{E} + 2\mu \mathbf{E}$$
2D (plane strain) $\begin{pmatrix} \sigma_{ij} \\ \sigma_{ij}$

$$(\lambda + \mu)\nabla\nabla \cdot \mathbf{u} + \mu\nabla \cdot \nabla \mathbf{u} = \mathbf{0} \quad (\lambda + \mu)u_{j,ji} + \mu u_{i,jj} = \mathbf{0}$$

$$1 \quad \cup_{\text{sths}} \quad \text{BC} \Rightarrow \quad \lambda \quad \bigoplus \quad \text{PDE} \quad \text{ale solve} \quad u : \begin{bmatrix} U_{1} \\ U_{2} \\ U_{3} \end{bmatrix}$$

$$2 \quad \varepsilon \quad = \quad \frac{1}{2} (\nabla u + \nabla u^{\dagger})$$

$$3 \quad \delta \quad = \quad \varepsilon \quad \varepsilon$$

_

Today, we cover a method that we first compute stress, then strain, and finally displacement.

Stress function approach What are Airy stress function approach?

Use of stress function \Rightarrow

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Balance of linear momentum is automatically satisfied (no body force, static)

$$\begin{array}{c} g_{11} = \Psi_{722} \\ g_{22} = \Psi_{11} \\ g_{21} = G_{12} = -\Psi_{712} \end{array} \qquad \begin{array}{c} \nabla_{0}G = \begin{pmatrix} G_{11}S_{1} + G_{12}, z \\ G_{11} + G_{22,12} \end{pmatrix} \\ \hline \nabla_{0}G = \begin{pmatrix} (\Psi_{722})_{,1} + (-\Psi_{712})_{,2} \\ (-\Psi_{122})_{,1} + (-\Psi_{711})_{,2} \end{pmatrix} = \begin{pmatrix} \Psi_{7221} - \Psi_{122} \\ -\Psi_{7221} + \Psi_{112} \end{pmatrix} \\ \hline \hline \nabla_{0}G = \begin{pmatrix} 0 \\ -\Psi_{722} + \Psi_{112} \end{pmatrix} \\ \hline \hline & 0 \end{pmatrix} \\ \hline \hline & \nabla_{0}G = \begin{pmatrix} 0 \\ -\Psi_{722} + \Psi_{112} \end{pmatrix} \\ \hline & 0 \\ \hline &$$

Is it too good to be true that we don't need to solve a PDE? Are there any potential problems?

Example
$$Y(X_{1}, X_{2}) = X_{1} \rightarrow$$

 $G_{11} = Y_{22} = 0$
 $G_{22} = Y_{31} = 12 \times 1$
 $G_{22} = -Y_{312} = 0$
 $G_{22} = -Y_{312} = 0$
 $U \rightarrow E = (T_{22} - V)$
 $U \rightarrow U \rightarrow U$
 $U \rightarrow U \rightarrow U$



The problem is that we have 3 equations and only 2 unknowns. That's not we cannot always solve for u



duplacements 4 can be integrated from E.

3D compatibility equations:

Can we always obtain u by integration? No

3 displacements (unknowns) 6 strains (equations)

Need to satisfy strain compatibility condition(s)

.

$$\frac{\partial^2 \varepsilon_{ik}}{\partial x_j \partial x_j} + \frac{\partial^2 \varepsilon_{jj}}{\partial x_i \partial x_k} - \frac{\partial^2 \varepsilon_{jk}}{\partial x_i \partial x_j} - \frac{\partial^2 \varepsilon_{ij}}{\partial x_j \partial x_k} = 0.$$

2D (I just derived them)

$$2\varepsilon_{12,12} - \varepsilon_{11,22} - \varepsilon_{22,11} = 0$$

Strains are obtained from stresses by using the inverse of stiffness matrix (compliance matrix) and stresses are obtained from the Airy stress function

$$2\psi_{,1122} + \psi_{,2222} + \psi_{,1111} = 0 \rightarrow$$

$$(\psi_{,11} + \psi_{,22})_{,11} + (\psi_{,11} + \psi_{,22})_{,22} = 0 \quad \text{comptability equal :}$$

$$(\Delta \psi) = \psi_{,11} + \psi_{,22} = 0 \quad \text{comptability equal :}$$

$$(\Delta \psi)_{,11} + (\Delta \psi)_{,722} = 0 \quad \text{the Ath order}$$

$$(\Delta \psi)_{,11} + (\Delta \psi)_{,722} = 0 \quad \text{the Ath order}$$

$$PDE \quad for \text{ stress first is a biharmonic fundi}$$

$$stress approach \quad duplacement \\ \Delta(\Delta \psi) = 0 \quad I \quad \text{true}$$

$$U(\Delta \psi) = 0 \quad I \quad \text{true}$$



Although the PDE of stress function is more difficult there are a lot of biharmonic functions available





Any Complex function does this, meaning that its real & imaging parts are harmonic Hny

$$f(z) = U(x_{1y}) + i V(x_{1y})$$

$$\Delta U = 0$$

$$\Delta V = 0$$

$$\Delta A U = 0 (A = billion A = billion$$

• Any biharmonic solution can be expressed by Kolonov-Muskhelishvili complex potentials, ϕ, χ :

$$\frac{\partial f}{\partial x} = 0, x + i V, x$$

$$\frac{\partial^2 f}{\partial y^2} = 0, y + i V, y$$

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$$\frac{\partial^2 f}{\partial y^2} = 0, y + i V, y$$

$$\frac{\partial^2 f}$$

$$\frac{\lambda^{4}}{\lambda^{4}} = \frac{f + 1}{2} = \frac{f}{2} \qquad f + 1 = \frac{f}{2} \qquad f + \frac{\lambda^{4}}{2} = \frac{\lambda^{4}}{2} \qquad \frac{\lambda^{2}}{2} = \frac{\lambda^{2}}{2} \qquad \frac{\lambda^{$$

Use for us:

Real and imaginary parts of a complex function satisfy

$$\nabla^2 \cup 0 \quad \& \quad \nabla^2 V_2 0$$

We were looking for a bi-harmonic function =>

We were looking for Airy stress function satisfying $\nabla^2 \nabla^2 \psi > 0$

Obviously if a function is harmonic

Then it is automatically a bi-harmonic funciton

The real and imaginary parts of a complex function are both also bi-harmonic functions