

Size effect:

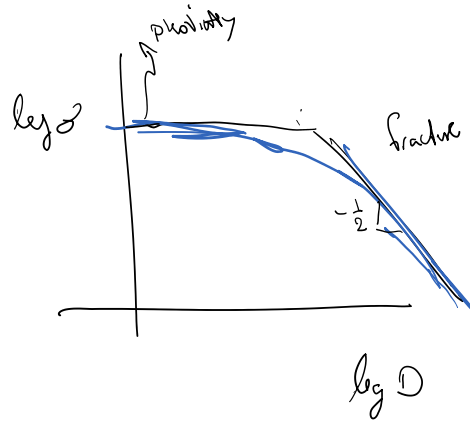
Bazant's size effect law

$$(\sigma_N)_u = Af_i \left(1 + \frac{D}{B}\right)^{-1/2} \approx \frac{Af_i}{\sqrt{1 + \frac{D}{B}}} \approx \frac{Af_i}{\sqrt{D/B}} \quad (14.8)$$

$D \gg B$

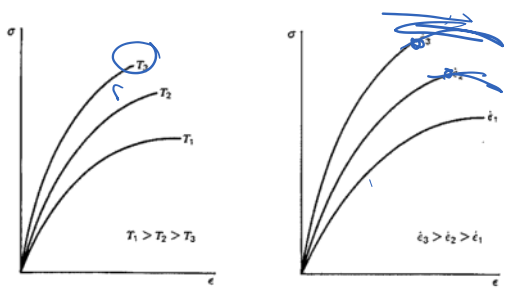
where

- $(\sigma_N)_u$  = Nominal stress at failure of a structure of specific shape and loading condition.
- $W$  = Characteristic length of the structure.
- $A, B$  = Positive constants that depend on the fracture properties of the material and on the shape of the structure, but not on the size of the structure.
- $f_i$  = Tensile strength of the material introduced for dimensional purposes.



## 7. Rate effects on ductility

- Same materials that show temperature toughness sensitivity (BCC metals) show high rate effect
- Polymers are highly sensitive to strain rate (especially for  $T >$  glass transition temperature)

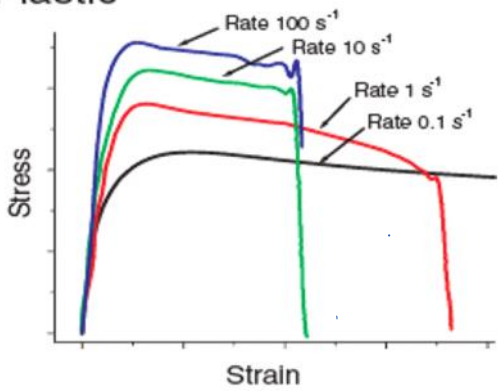


$\dot{\epsilon} \uparrow \rightarrow$  strength increase

Strain rate  $\nearrow$  similar to  $T \searrow$

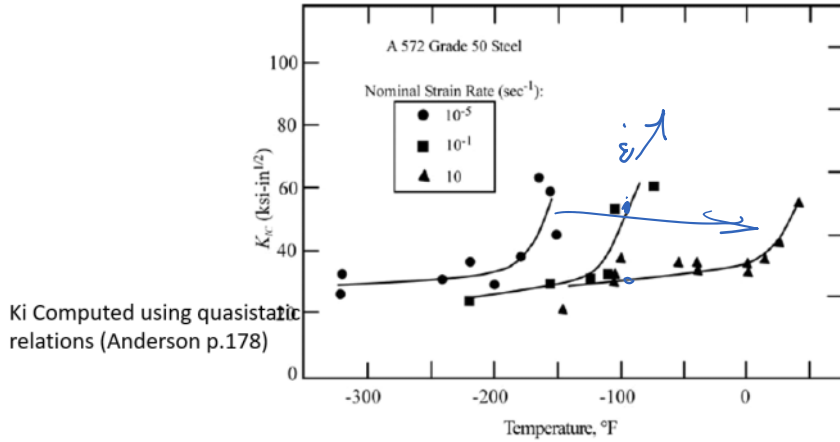
- Strain rate  $\nearrow$ 
  1. Strength  $\nearrow$
  2. Ductility  $\searrow$
  3. Toughness not clear

Plastic



# Strain rate effects on Impact toughness

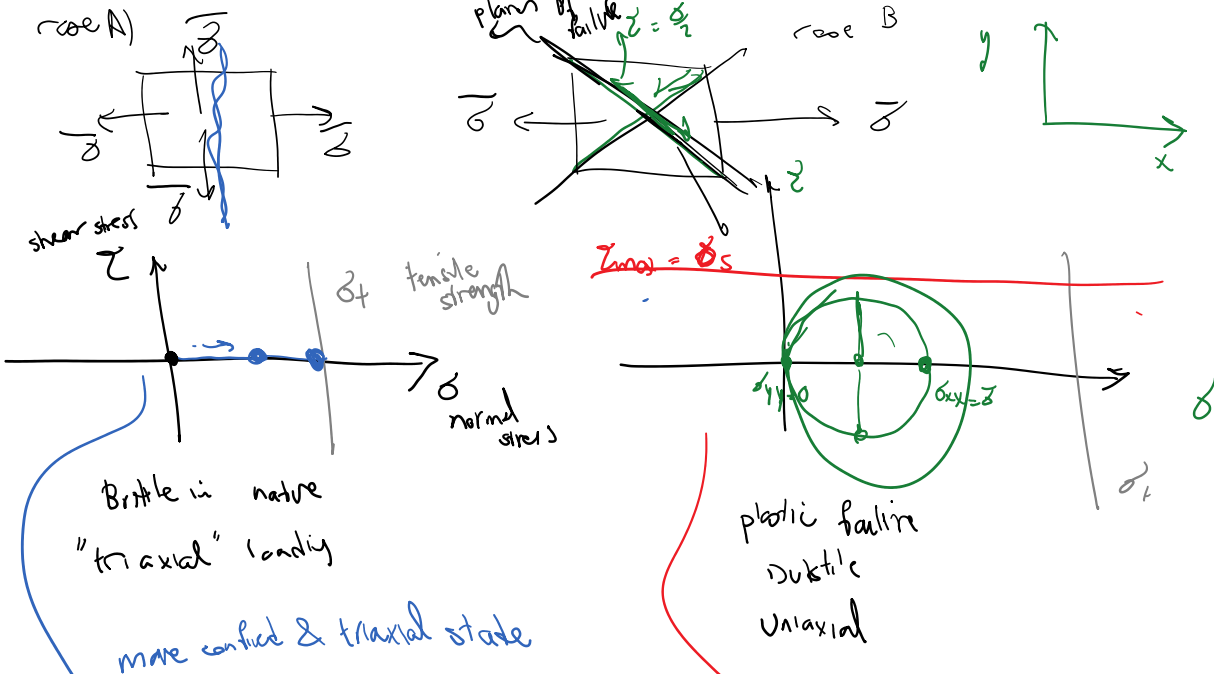
Strain rate  $\nearrow$   $\Rightarrow$   
 DBTT  $\nearrow$  (more brittle in impact)



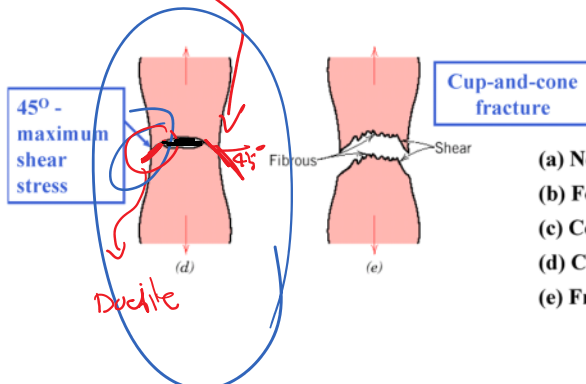
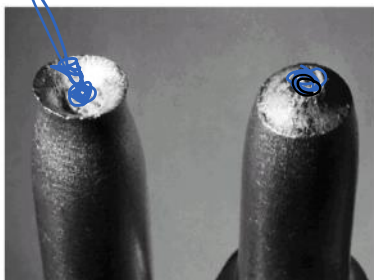
$K_{Ic}$  Computed using quasistatic relations (Anderson p.178)

FIGURE 4.5 Effect of loading rate on the cleavage fracture toughness of a structural steel. Taken from Barsom, J.M., "Development of the AASHTO Fracture Toughness Requirements for Bridge Steels." *Engineering Fracture Mechanics*, Vol. 7, 1975, pp. 605-618.

## 8. Triaxial stress and confinement

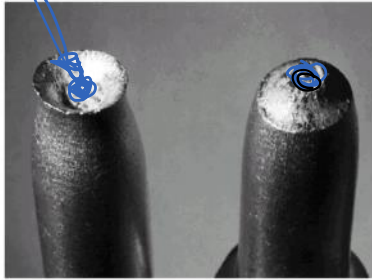


### Ductile Fracture

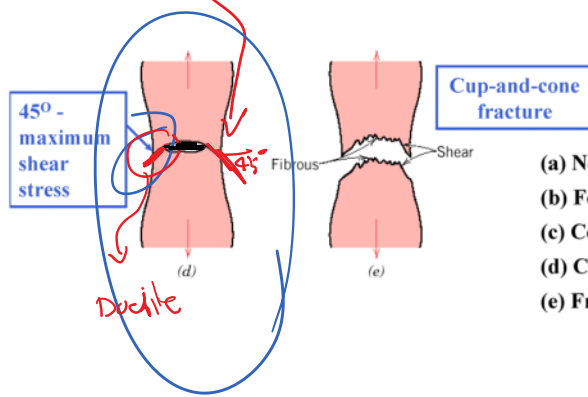


- (a) Necking
- (b) Formation of microvoids
- (c) Coalescence of microvoids to form a crack
- (d) Crack propagation by shear deformation
- (e) Fracture

## Ductile Fracture



(Cup-and-cone fracture in Al)



- (a) Necking
- (b) Formation of microvoids
- (c) Coalescence of microvoids to form a crack
- (d) Crack propagation by shear deformation
- (e) Fracture

Side note we can use other failure criteria  
 Mohr - Coulomb (m.c.)

$$\sigma_v = \dots$$

Larger specimens have less of surface regions even for uniaxial loading and tend to have higher relative volume of "triaxial" stress state -> more prone to brittle fracture (this is another reason beside having higher probability of larger defects for brittle fracture)

## 8. Triaxial stress and confinement

**Effect of specimen thickness**

- **Larger specimen size** (in-service components) provides higher constraint -> **more brittle**.

Effect of section thickness on transition temperature

If large size specimens are used, the transition temperature will increase.

**Large scale tests**

Suranaree University of Technology | Tapany Udomphol | May-Aug 2007

source: Tapany Udomphol, Suranaree University of Technology  
[http://eng.sut.ac.th/metal/images/stories/pdf/14\\_Brittle fracture and impact testing 1-6.pdf](http://eng.sut.ac.th/metal/images/stories/pdf/14_Brittle%20fracture%20and%20impact%20testing%201-6.pdf)

Often hardening (increasing strength) reduces ductility

Phenomena affecting ductile/brittle response

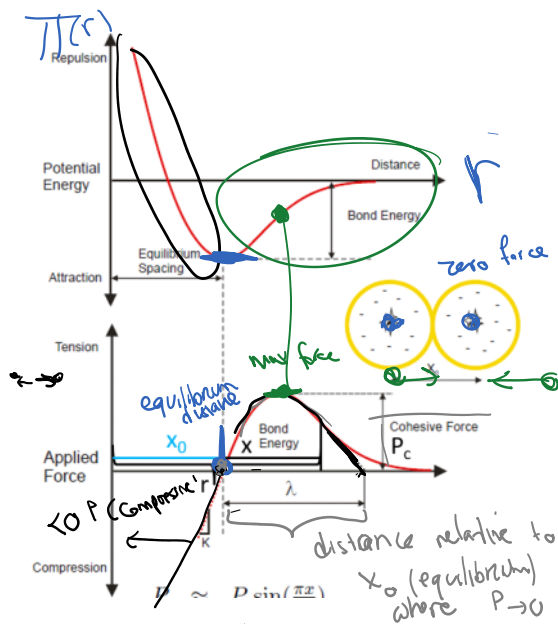
1. T (especially for BCC metals and ceramics)
2. Impurities and alloying
3. Radiation
4. Hydrogen embrittlement
5. Grain size
6. Size effect
7. Rate effect
8. Confinement and triaxial stress state



Decreasing grain size is the only mechanism that hardens and promotes toughness

## 4. Linear Elastic Fracture Mechanics (LEFM)

### 4.1 Griffith energy approach



Handwritten notes and diagrams for the Griffith energy approach:

- Diagram of a crack of length  $2r$  under tensile force  $P$ .
- Equation:  $P = -\frac{dT}{dr}$
- Equation:  $P = \frac{dT}{dr}$  (tensile force)
- Equation:  $P(\alpha) = P_c \sin\left(\frac{\pi\alpha}{\lambda}\right)$
- Equation:  $\alpha = r - x_0$  (displacement beyond equilibrium)

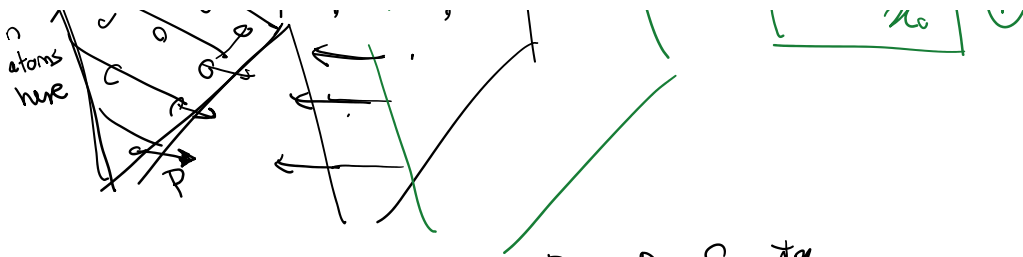
Handwritten notes:

- step 1  $X \rightarrow \epsilon$  (strain)
- step 2  $P \rightarrow \sigma$  (stress)
- original length  $x_0$
- displacement  $x$



strain  $\epsilon = \frac{\text{change of length}}{\text{original length}}$

$$\epsilon = \frac{x}{x_0} \quad (1)$$



step 2:  $P \rightarrow \delta$

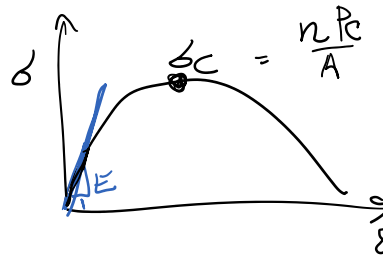
$P = P_c \sin \frac{\pi x}{\lambda}$   $\rightarrow$  cut-off distance  
 of 1 atom

$$F = \sum P = n \cdot P_c \cdot \lambda \cdot \sin \frac{\pi x}{\lambda}$$

$$\delta = \frac{F}{A} = \underbrace{\left(\frac{n}{A}\right)}_{\text{density of atoms}} P_c \frac{\sin \pi x}{\lambda}$$

$$x = \epsilon \lambda_0 \quad (1)$$

$$\delta = \left(\frac{n P_c}{A}\right) \sin \frac{\epsilon \lambda_0 \pi}{\lambda}$$



$$\delta = \delta_c \sin\left(\frac{\epsilon \lambda_0 \pi}{\lambda}\right) \quad (2)$$

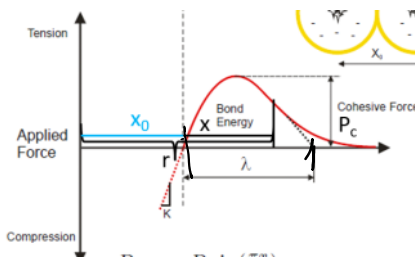
$$E = \left. \frac{d\delta}{d\epsilon} \right|_{\epsilon=0}$$

strength

what is the elastic modulus

$$E = \left. \frac{d\delta}{d\epsilon} \right|_{\epsilon=0} = \delta_c \frac{\pi \lambda_0 \pi}{\lambda} \cos\left(\frac{\epsilon \lambda_0 \pi}{\lambda}\right) \Big|_{\epsilon=0} \rightarrow \boxed{E = \delta_c \frac{\lambda_0 \pi}{\lambda}} \quad (3)$$

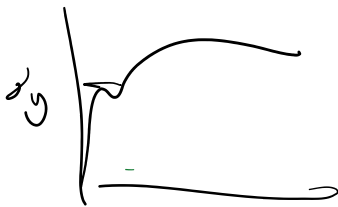
$$\delta_c \leftrightarrow E$$



$$\frac{\lambda}{\lambda_0} = O(1)$$

$$E \approx \delta_c \pi \rightarrow \boxed{\delta_c \approx \frac{E}{\pi}} \quad (4)$$

Steel  $E = 200 \text{ GPa}$ , (4) predicts strength  $\delta_c \approx 60 \text{ GPa}$ !  
theoretical estimate



$\sigma_y = 250 \text{ MPa}$   
practici.

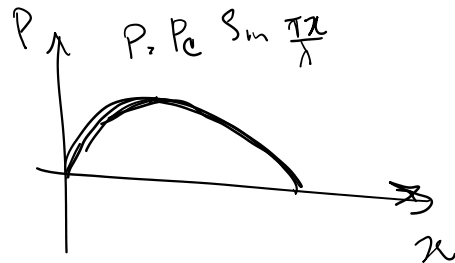
$E$  über  $\text{GPa}$

Strengths  $\text{MPa}$   
 über

/ why equation (4) does not hold?

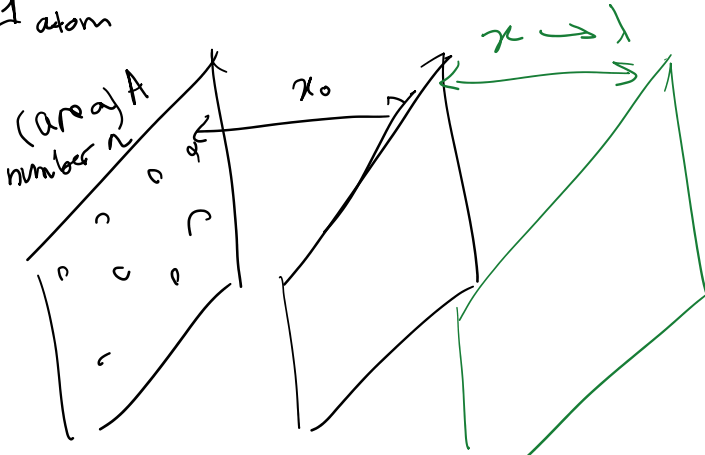
$E \leftrightarrow \sigma_c$

let's also relate these to toughness



1 atom

$$W_1 = \int_{x=0}^{\lambda} P(x) dx = \int_0^{\lambda} P_c \sin \frac{\pi x}{\lambda} dx = \frac{2\lambda}{\pi} P_c$$

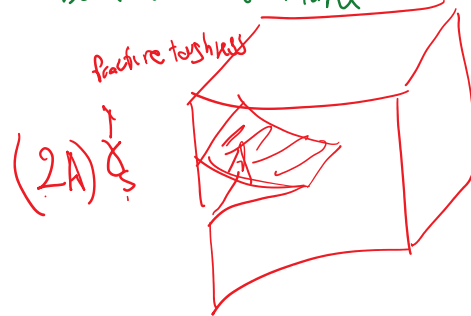


$W$  for this plane =  $n W_1 = \frac{2n\lambda}{\pi} P_c$

energy needed to break the bond on this plane

$$\frac{W}{A} = \frac{2n\lambda P_c}{\pi A} = 2\gamma_s$$

energy per unit area needed to cause fracture



$$\frac{n P_c}{A} = \sigma_c$$

$\sigma_s \leftrightarrow \delta_s$

$$\sigma_s = \frac{\lambda}{\pi} \sigma_c \rightarrow \left[ \sigma_c = \frac{\pi}{\lambda} \delta_s \right] \text{ (5)}$$

multiply these two

we also had  $\sigma_c \approx \frac{E}{\pi} \frac{\lambda}{\lambda_0}$  (3)

$E \leftrightarrow \sigma_c$

$$\sigma_c^2 = \frac{\pi}{\lambda} \delta_s \frac{E}{\pi} \frac{\lambda}{\lambda_0} \rightarrow \sigma_c = \sqrt{\frac{E \delta_s}{\lambda_0}}$$

Summary

$$\sigma_c = \frac{E}{\pi} \frac{\lambda}{\lambda_0} \quad \& \quad \sigma_c = \sqrt{\frac{E \delta_s}{\lambda_0}} \quad \text{(6)}$$

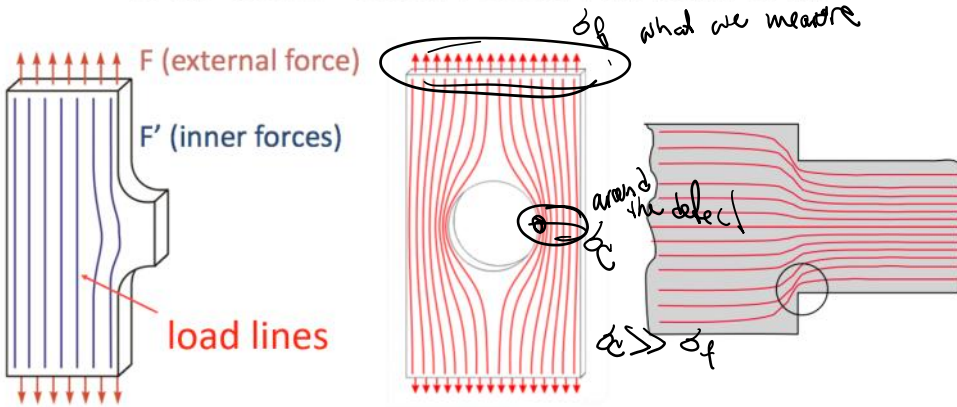
Summary  $\sigma_c = \frac{E}{r} \frac{\lambda}{\lambda_0}$  &  $\sigma_c = \sqrt{\frac{E \delta s}{\lambda_0}}$  (6)

The main reason why the stress is 100s to <1000 smaller than theoretical estimate is the presence of defects

We'll use two explanations for this discrepancy  
Stress-based and energy-based

1. Stress-based (stress concentration around defects, ...)

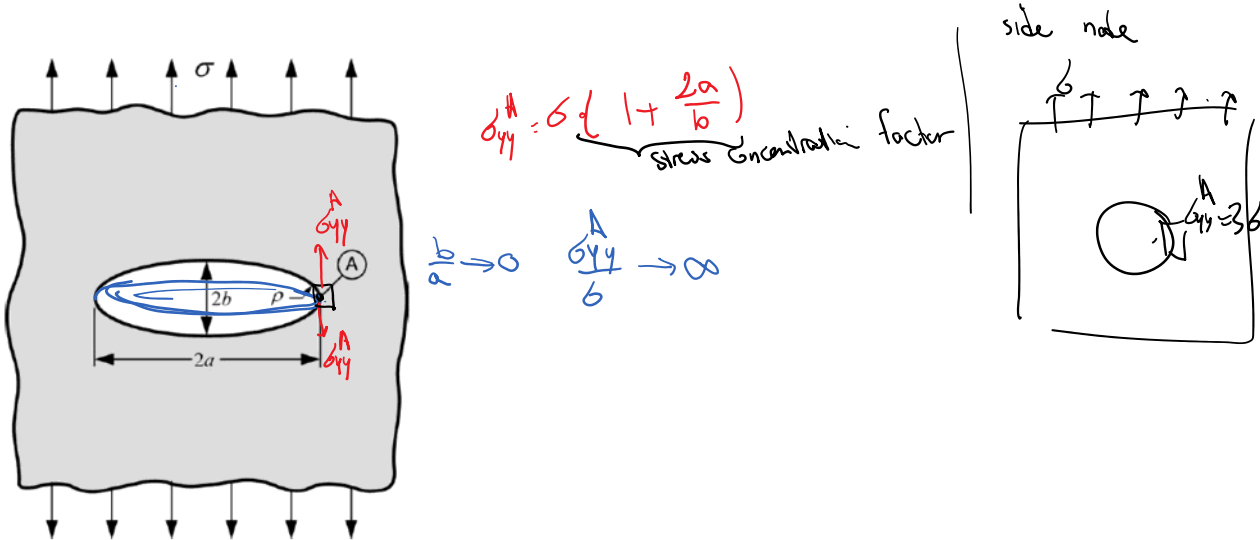
# Stress concentration

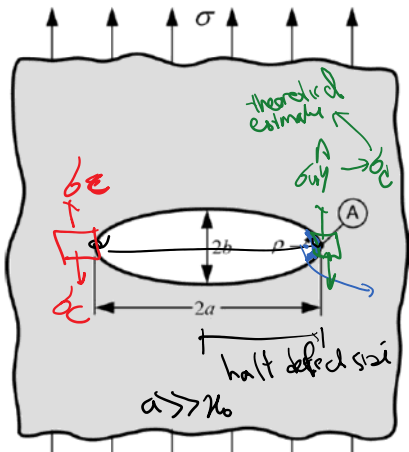


**Geometry discontinuities:** holes, corners, notches, cracks  
etc: stress concentrators/risers

## Elliptic hole

Inglis, 1913, theory of elasticity





$$\rho = \frac{b^2}{a} \quad \left. \begin{array}{l} \sigma_{yy}^A = \delta \left(1 + 2\sqrt{\frac{a}{\rho}}\right) \end{array} \right\} \rightarrow \sigma_{yy}^A = \delta \left(1 + 2\sqrt{\frac{a}{\rho}}\right)$$

$\rho =$  radius of curvature



$\rho_{min} \sim \lambda_0$   
 dense core schematic  
 "sharpest crack"

$$\sigma_{yy}^A = \delta \left(1 + 2\sqrt{\frac{a}{\lambda_0}}\right)$$

for the sharpest crack

$\delta \rightarrow \sigma_f$   
 macroscopic strength

$$\sigma_{yy}^A \approx 2\delta \sqrt{\frac{a}{\lambda_0}}$$

Increase  $\delta$  until  
 $\sigma_{yy}^A \rightarrow \sigma_c$

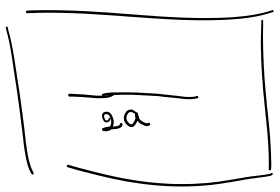
then we'll experience macroscopic strength ( $\delta \rightarrow \sigma_f$ )

$$\sigma_c = 2 \sigma_f \sqrt{\frac{a}{\lambda_0}}$$

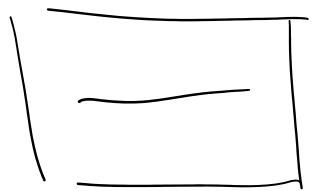
$$\sigma_f = \sqrt{\frac{\lambda_0}{4a}} \sigma_c \quad (9)$$

$$\sigma_f \propto \frac{\sigma_c}{\sqrt{a}}$$

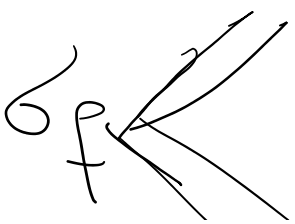
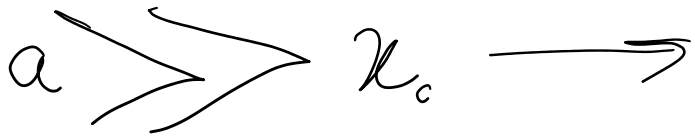
Stress Concentration  
 argument



$\sigma_f$  high




$\sigma_f \downarrow$  as  $\frac{1}{\sqrt{a}}$



$\sigma_c$

$\propto F$



$\sim 100 \text{ MPa}$    $\sigma \propto \sqrt{E/\rho}$

# Griffith's work (brittle materials)

FM was developed during WWI by English aeronautical engineer A. A. Griffith to explain the following observations:



- The stress needed to fracture **bulk glass** is around **100 MPa**
- The **theoretical stress needed for breaking atomic bonds** is approximately **10,000 MPa**
- experiments on glass fibers that Griffith himself conducted: the fracture stress increases as the fiber diameter decreases => Hence the uniaxial tensile strength, which had been used extensively to predict material failure before Griffith, could not be a specimen-independent material property.


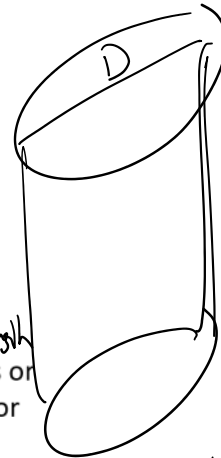
 Griffith suggested that the low fracture strength observed in experiments, as well as the size-dependence of strength, was due to the presence of **microscopic flaws** in the bulk material.

TABLE I.1. Strength of glass fibers according to Griffith's experiments.

Diameter (10 <sup>-3</sup> in)	Breaking stress (lb/in <sup>2</sup> )	Diameter (10 <sup>-3</sup> in)	Breaking stress (lb/in <sup>2</sup> )
40.00	24 900	0.95	117 000
4.20	42 300	0.75	134 000
2.78	50 800	0.70	164 000
2.25	64 100	0.60	185 000
2.00	79 600	0.56	154 000
1.85	88 500	0.50	195 000
1.75	82 600	0.38	232 000
1.40	85 200	0.26	332 000
1.32	99 500	0.165	498 000
1.15	88 700	0.130	491 000

$$\sigma_f = \sqrt{\frac{2\gamma_s}{4a}} \sigma_c$$



$\sim 20\times$  increase in strength

"the weakness of isotropic solids... is due to the presence of discontinuities or flaws... The effective strength of technical materials could be increased 10 or 20 times at least if these flaws could be eliminated."

the larger D the larger potential  $a$ 's

