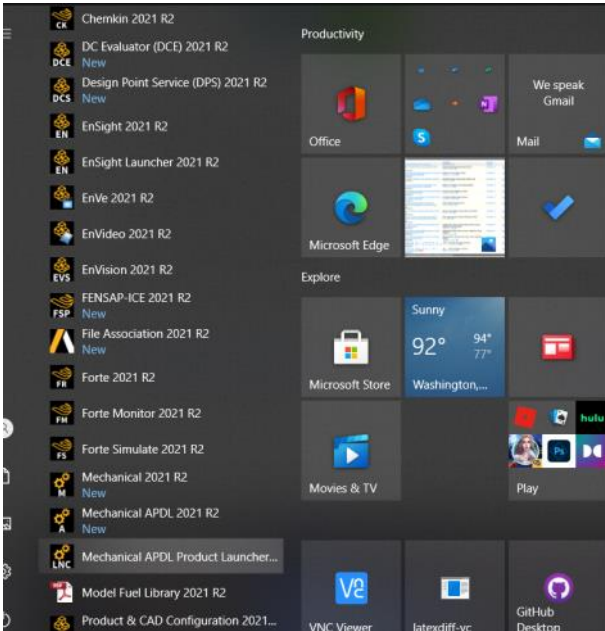


2021/08/24

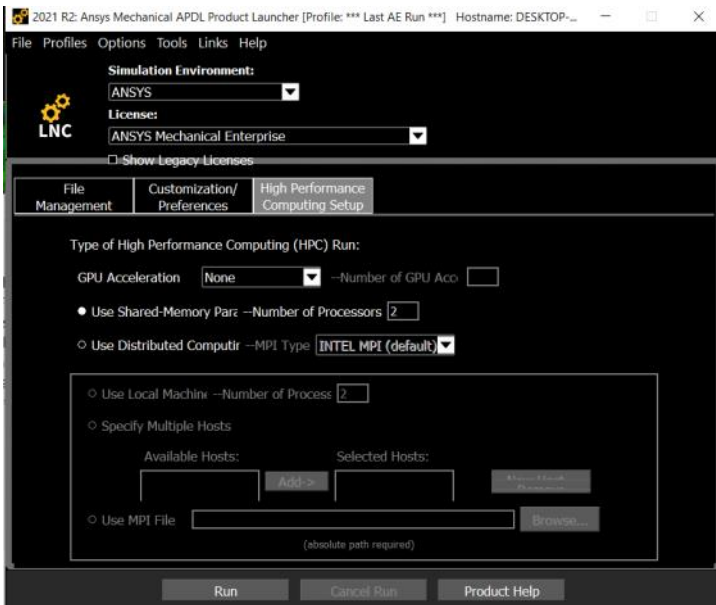
Tuesday, August 24, 2021 1:31 PM

If you need resources for learning C++, just contact me (for the second project)

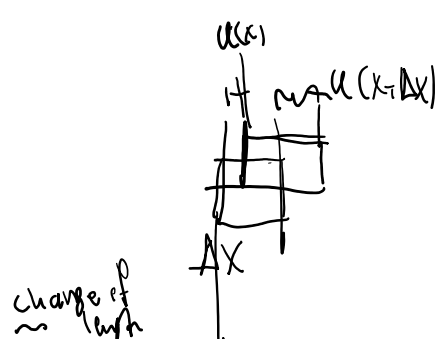
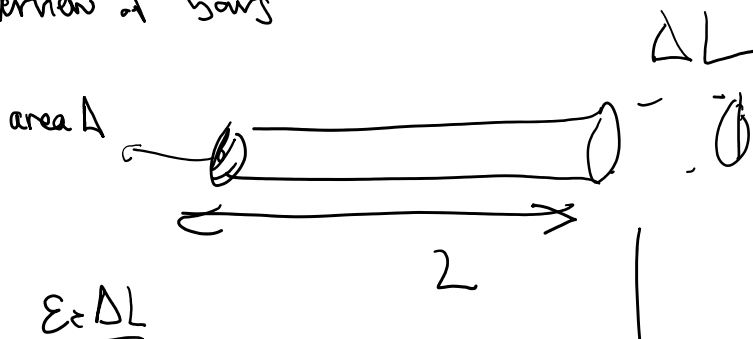
Running Ansys after the installation:



So choose shared memory option here before running it on your laptop:



overview of bars



$$\epsilon = \frac{\Delta L}{L}$$

$$\sigma = E \epsilon$$

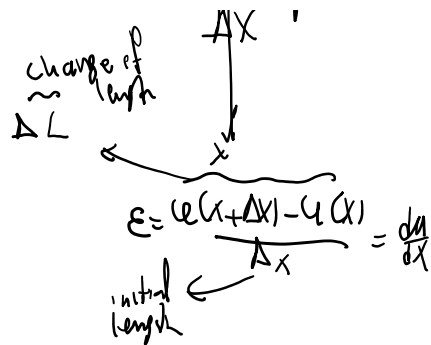
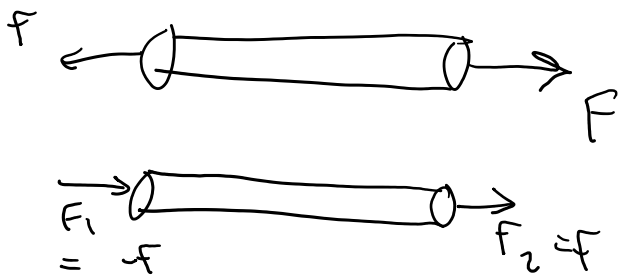
electric modulus

$$F = \sigma A = EA \epsilon$$



$$\Delta L = u_2 - u_1$$

$$F = EA \left( \frac{u_2 - u_1}{L} \right) = \frac{EA}{L} (u_2 - u_1)$$



$F = EA \frac{du}{dx}$

more general  $E(x) \cdot A(x)$   
we'll do it later

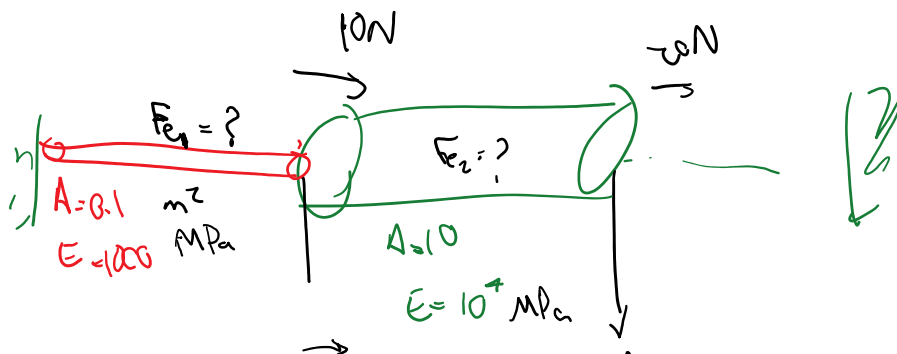
$$\begin{bmatrix} F_1 \\ F_2 \end{bmatrix} = \frac{EA}{L} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \end{bmatrix}$$

FEM

stiffness matrix for a bar  
for constant A constant E

Solid Mechanics version

$$F = \frac{EA}{L} \Delta L$$



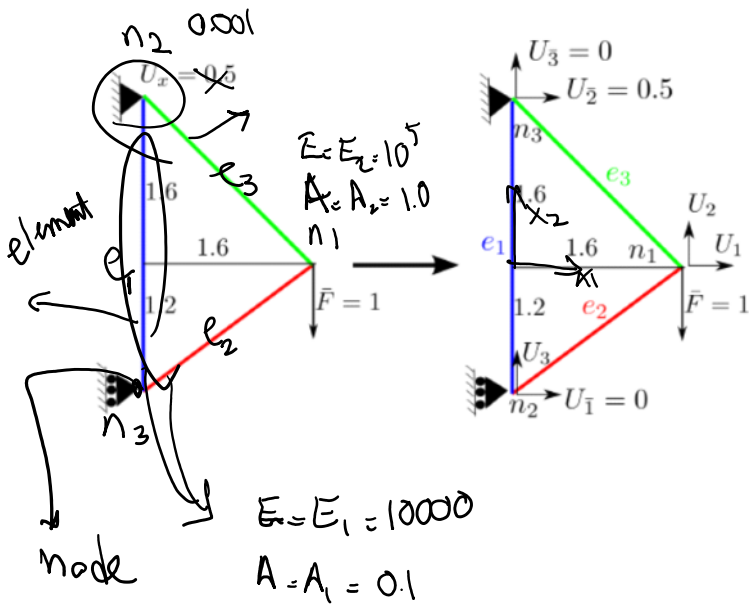
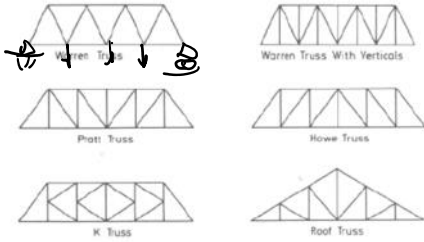
- equ

$$E = 10^7 \text{ MPa}$$

$$U_1 = ? \quad U_2 = ?$$

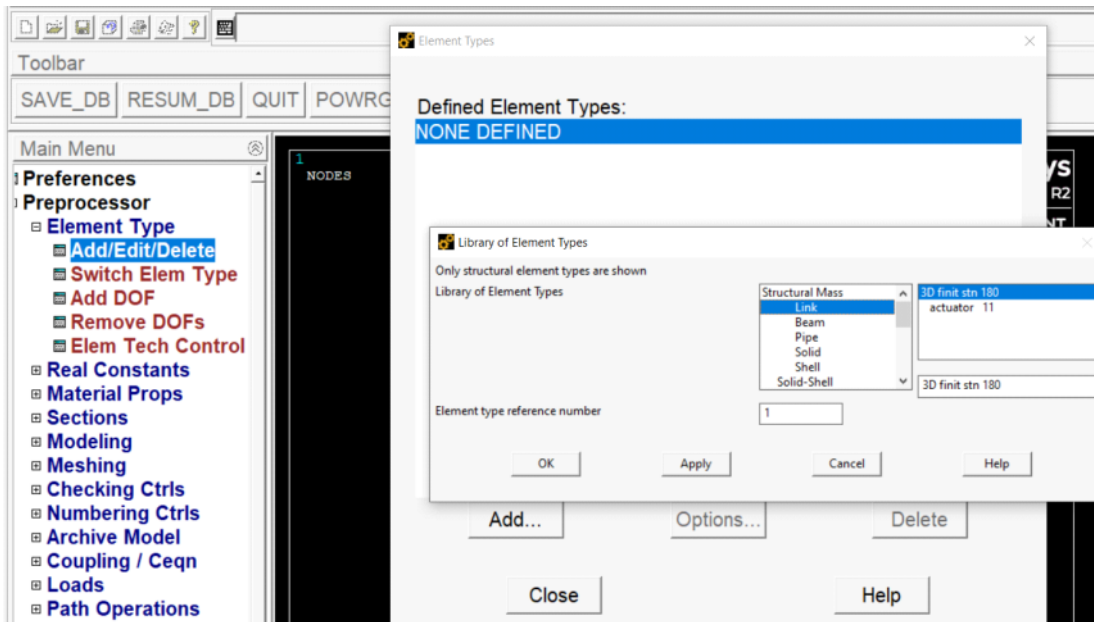
want to solve this to obtain  $U_1, U_2, F_1, F_2, \dots$

Trusses are simply bars assembled in 2D & 3D  
rotated

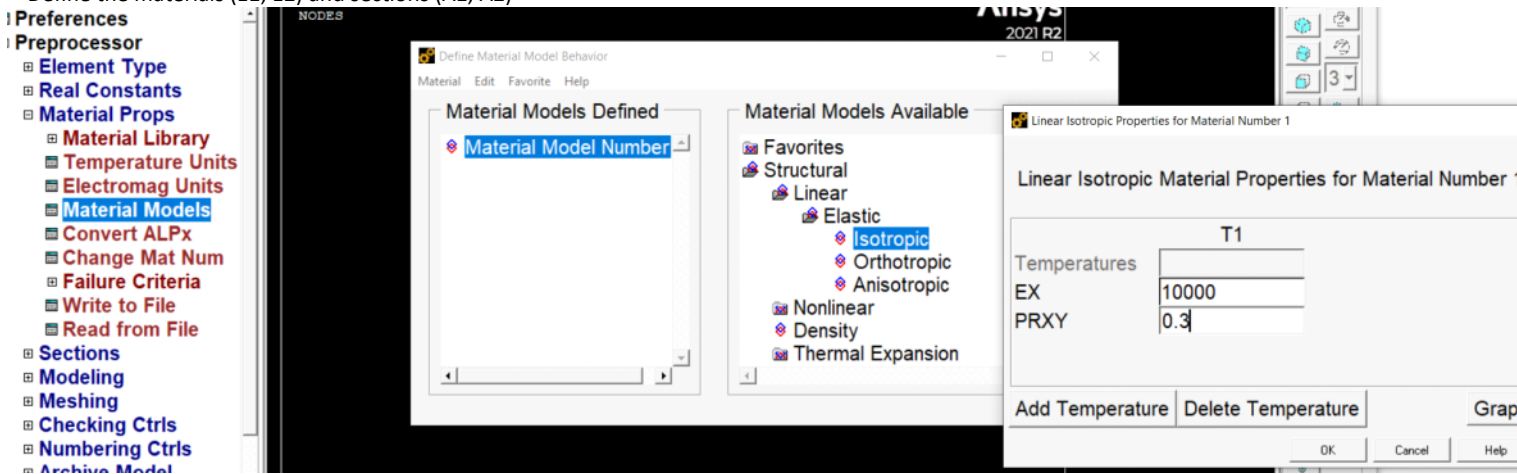


In Ansys truss elements are called links.

- Specify that we are using truss (link) elements.

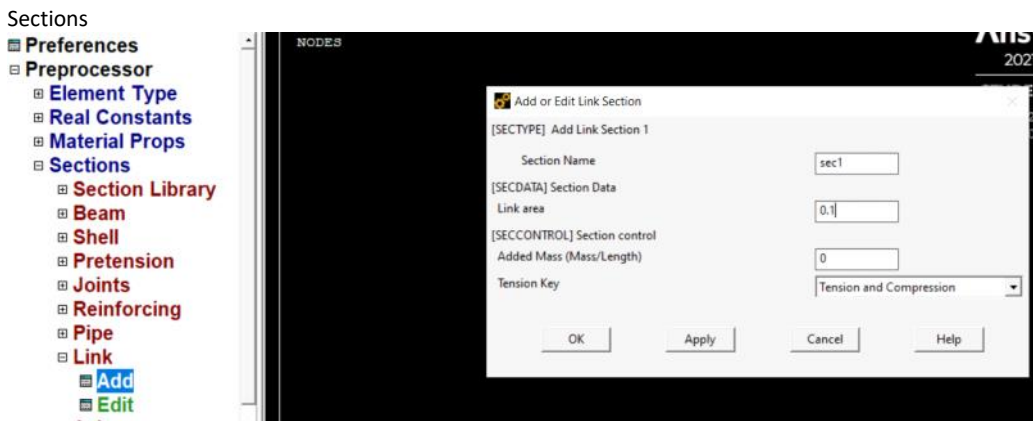


-- Define the materials (E1, E2) and sections (A1, A2)

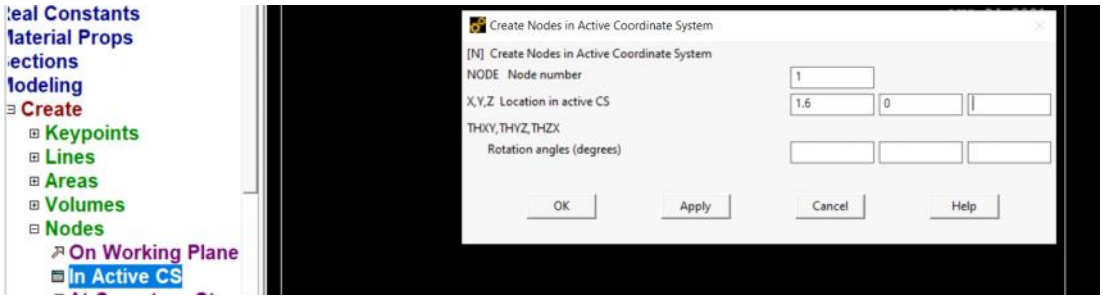


Copy and paste, and edit E = 10000 for material 2 (E2)

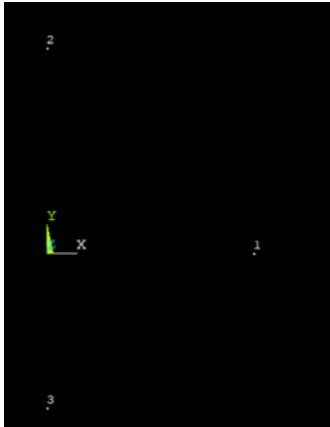
----



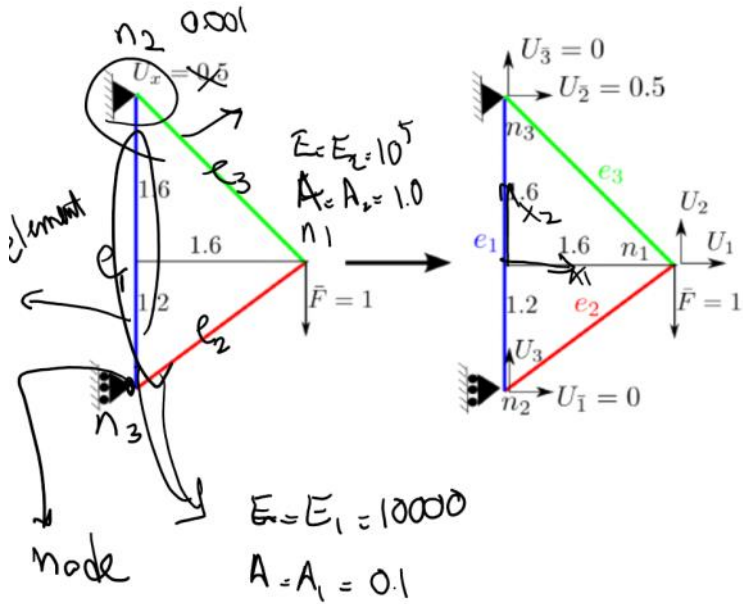
Create nodes



Apply and create nodes 2, 3

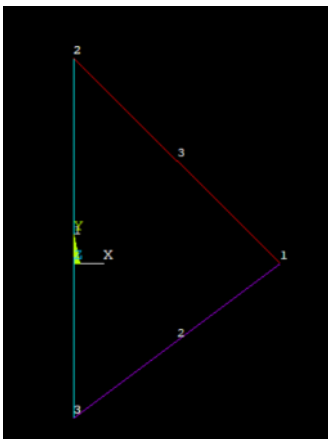
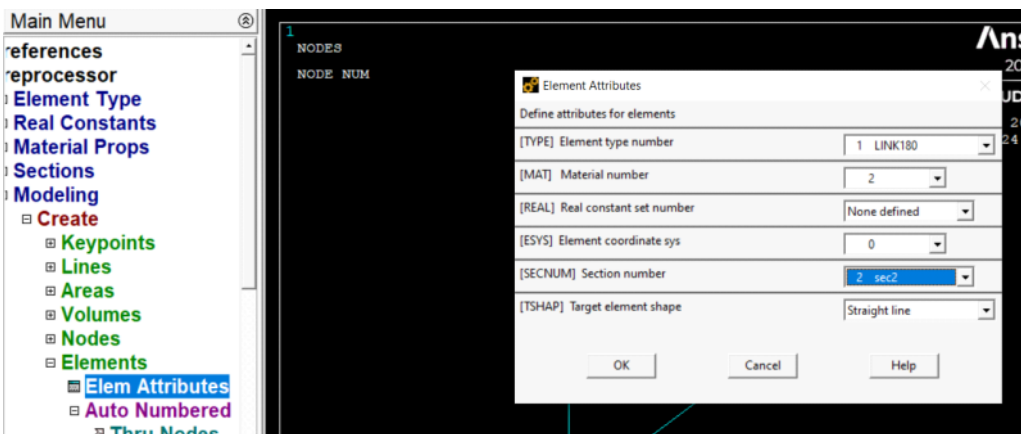


- Create elements

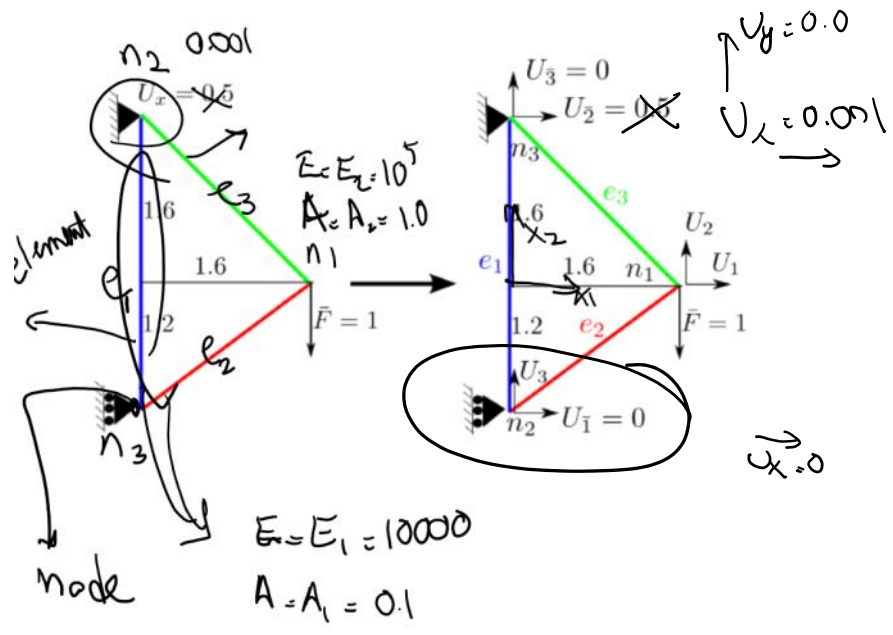




After creating element 2, we change to E2 and A2 to create element e3

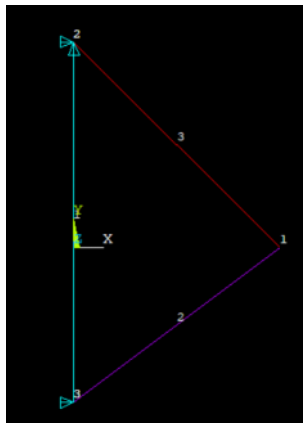
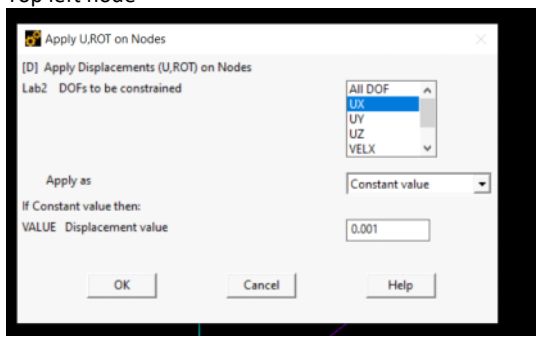


-----  
 We apply supports (prescribed displacements)

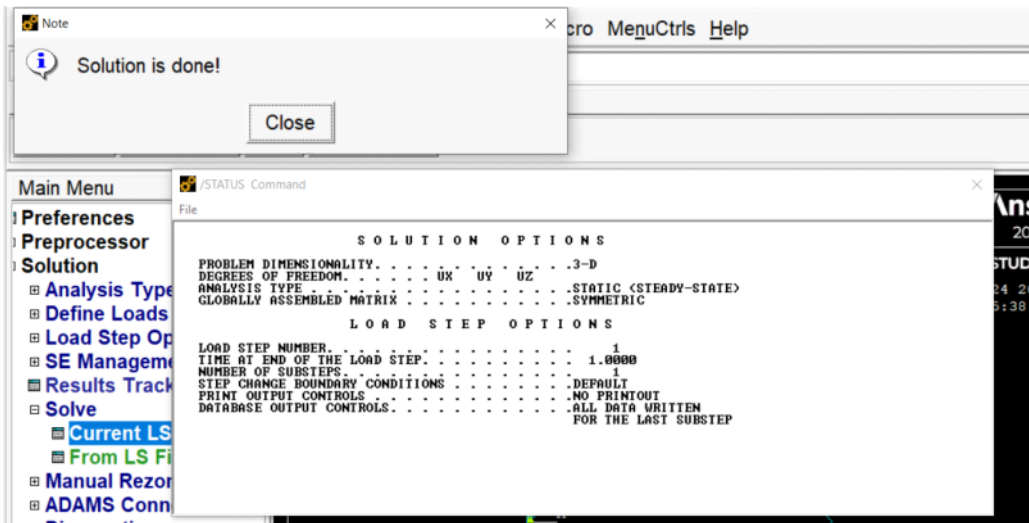
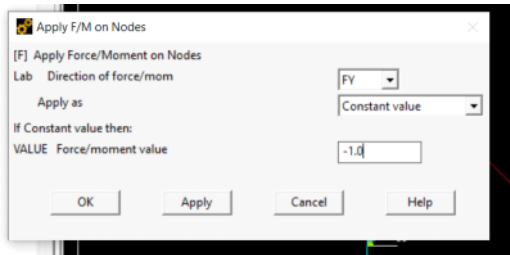
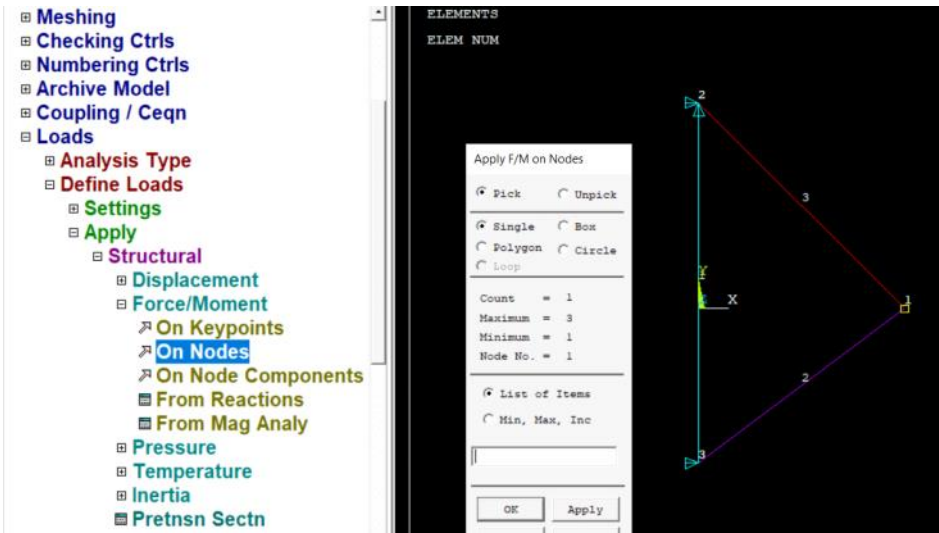


- ▣ Numbering Ctrl
- ▣ Archive Model
- ▣ Coupling / Ceqn
- ▣ Loads
  - ▣ Analysis Type
  - ▣ Define Loads
    - ▣ Settings
    - ▣ Apply
      - ▣ Structural
        - ▣ Displacement
          - ▣ On Lines
          - ▣ On Areas
          - ▣ On Keypoints
          - ▣ On Nodes

Top left node



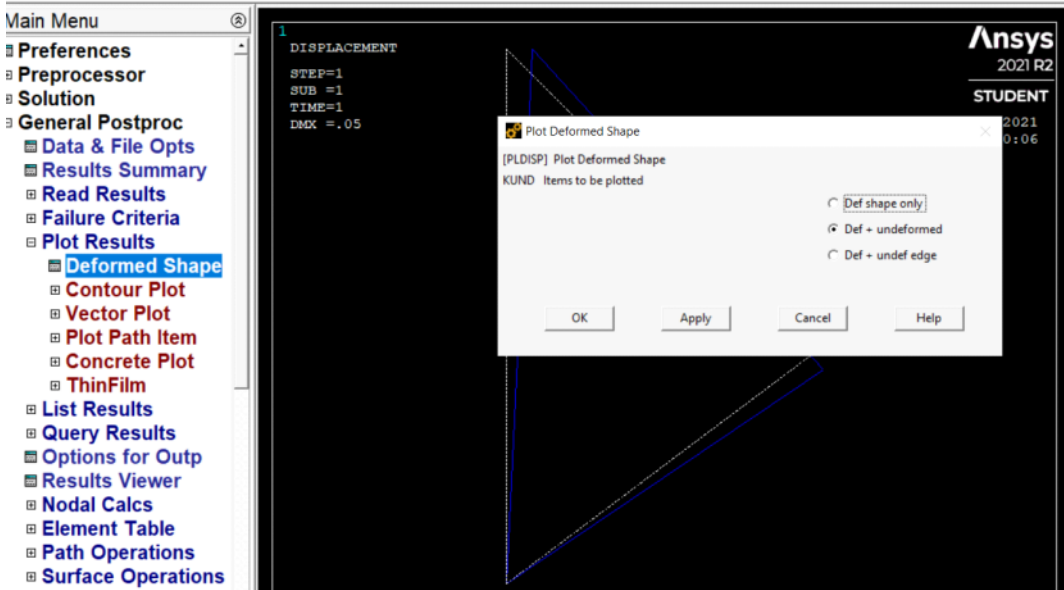
--- apply the force on node 1



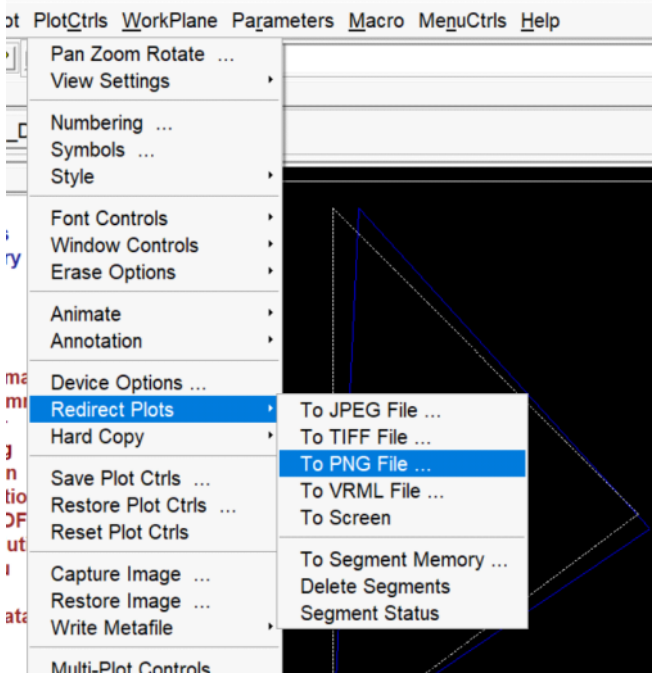
-----  
Postprocessing

Show the deformed shape:



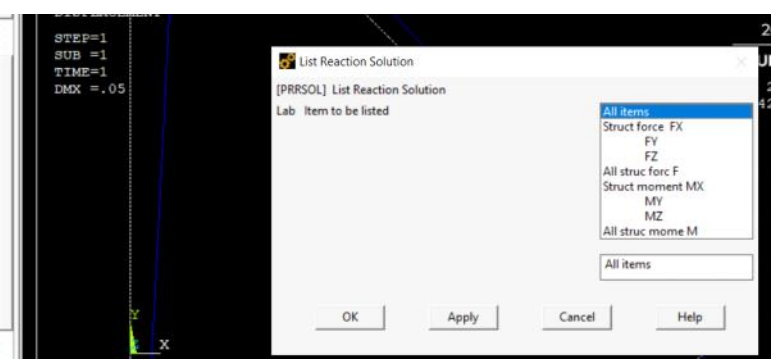


If you want to save the deformed shape or any other thing do the following



For 1D elements (bars, trusses, ...) we list the following:

- A. Reaction forces
- General Postproc
- Data & File Opts
- Results Summary
- Read Results
- Failure Criteria
- Plot Results
- List Results
  - Detailed Summary
  - Iteration Summary
  - Percent Error
  - Sorted Listing
  - Nodal Solution
  - Element Solution
  - Superelem DOF
  - SpotWeld Solution



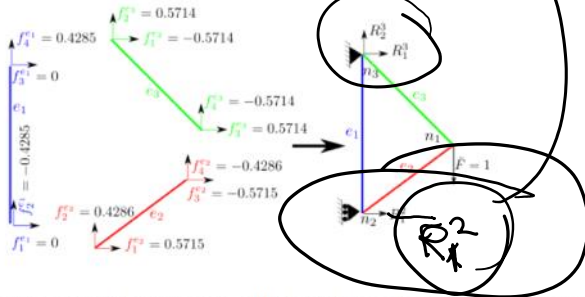
```

NODE   FX   FY   FZ
2     0 0000  0.71511E-016
  
```

NODE	FX	FY	FZ
2	0.0000	0.71511E-016	
3	0.94369E-016		

TOTAL VALUES  
 VALUE 0.94369E-016 0.71511E-016 0.0000

When you do it at home, you get these values



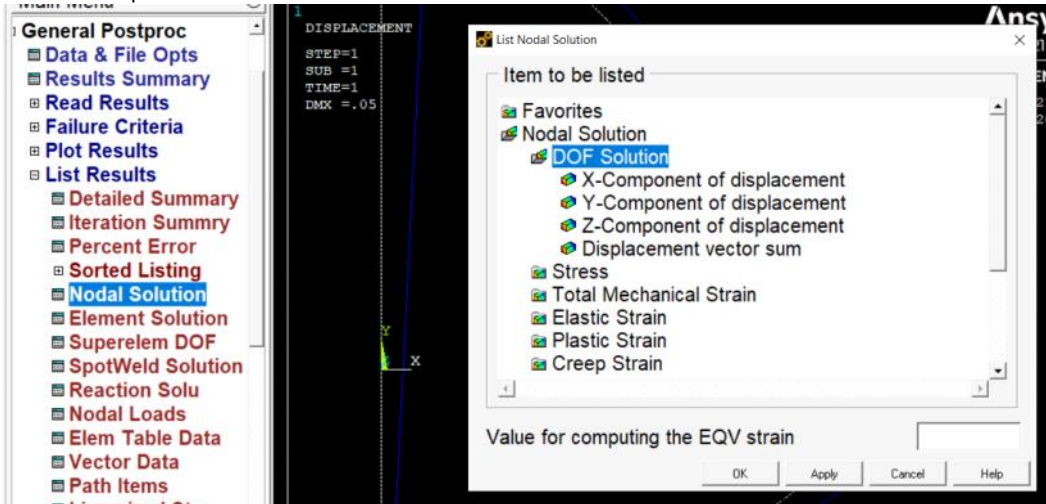
- First, we compute reaction forces by adding up forces from individual elements that contribute to reaction forces:

$$R_1^2 = f_1^{e1} + f_1^{e2} = 0 + 0.5715 = 0.5715 \quad (397a)$$

$$R_1^3 = f_3^{e1} + f_3^{e2} = 0 + -0.5714 = -0.5714 \quad (397b)$$

$$R_2^3 = f_4^{e1} + f_2^{e3} = 0.4285 + 0.5714 = 0.9999 \quad (397c)$$

Get nodal displacements



PRINT U NODAL SOLUTION PER NODE

\*\*\*\*\* POST1 NODAL DEGREE OF FREEDOM LISTING \*\*\*\*\*

LOAD STEP= 1 SUBSTEP= 1  
 TIME= 1.0000 LOAD CASE= 0

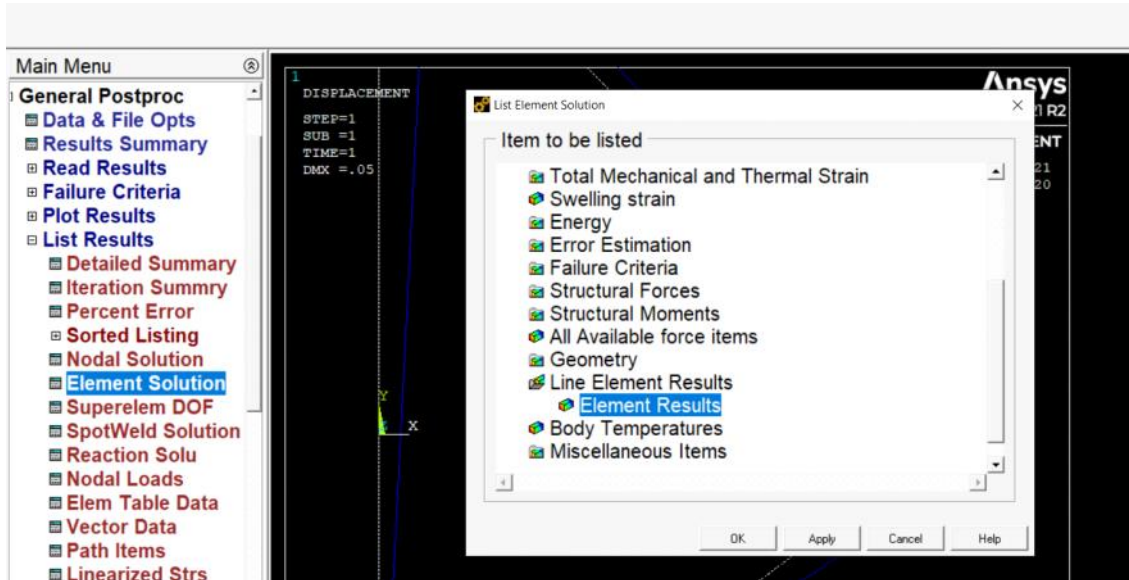
THE FOLLOWING DEGREE OF FREEDOM RESULTS ARE IN THE GLOBAL COORDINATE SYSTEM

NODE	UX	UY	UZ	USUM
1	0.21429E-001	-0.28571E-001	0.0000	0.35714E-001
2	0.50000E-001	0.0000	0.0000	0.50000E-001
3	0.0000	-0.20023E-015	0.0000	0.20023E-015

MAXIMUM ABSOLUTE VALUES

NODE 2 1 0 2  
 VALUE 0.50000E-001-0.28571E-001 0.0000 0.50000E-001

To list element forces use



PRINT ELEM ELEMENT SOLUTION PER ELEMENT

\*\*\*\*\* POST1 ELEMENT SOLUTION LISTING \*\*\*\*\*

LOAD STEP 1 SUBSTEP= 1  
 TIME= 1.0000 LOAD CASE= 0

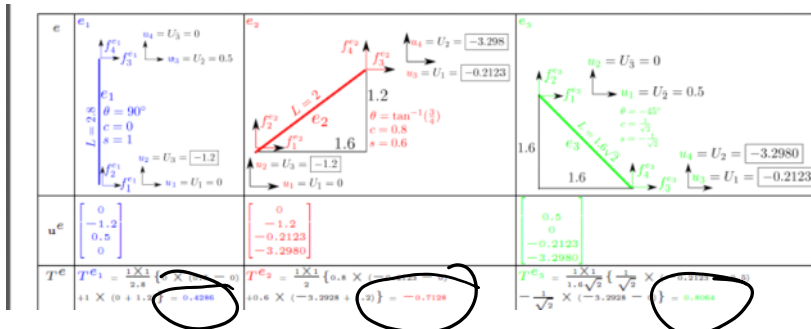
EL= 1 NODES= 2 3 MAT= 1 XC,YC,ZC= 0.000 0.2000 0.000 AREA= 0.10000E-01 LINK180  
 FORCE= 0.71511E-16 STRESS= 0.71511E-14 EPEL= 0.71511E-16  
 TEMP= 0.00 0.00 EPTH= 0.0000

Incorrect forces

EL= 2 NODES= 3 1 MAT= 1 XC,YC,ZC= 0.8000 -0.6000 0.000 AREA= 0.10000E-01 LINK180  
 FORCE=-0.11796E-15 STRESS=-0.11796E-13 EPEL=-0.11796E-15  
 TEMP= 0.00 0.00 EPTH= 0.0000

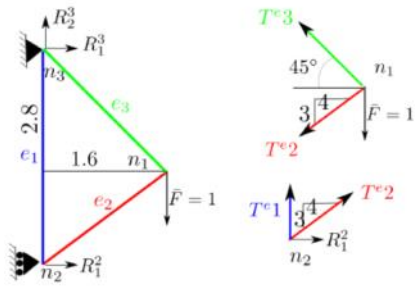
EL= 3 NODES= 2 1 MAT= 2 XC,YC,ZC= 0.8000 0.8000 0.000 AREA= 0.10000 LINK180  
 FORCE= 0.0000 STRESS= 0.0000 EPEL= 0.0000  
 TEMP= 0.00 0.00 EPTH= 0.0000

These are the correct forces



The exact solution of this statistically-determinant structure is obtained here (forces do not depend on E and A)

### Truss Example: Direct solution method



• Since this is a statically determinate structure, we can easily solve the forces and verify our FEM forces.

$$\Sigma F_2 = 0 \Rightarrow R_2^3 - F = 0 \Rightarrow R_2^3 = 1 \quad (399a)$$

$$\Sigma M_{n_3} = 0 \Rightarrow 2.8R_1^2 - 1.6F = 0 \Rightarrow R_1^2 = \frac{4}{7} = 0.5714 \quad (399b)$$

$$\Sigma F_1 = 0 \Rightarrow R_1^2 + R_1^3 = 0 \Rightarrow R_1^3 = -\frac{4}{7} = -0.5714 \quad (399c)$$

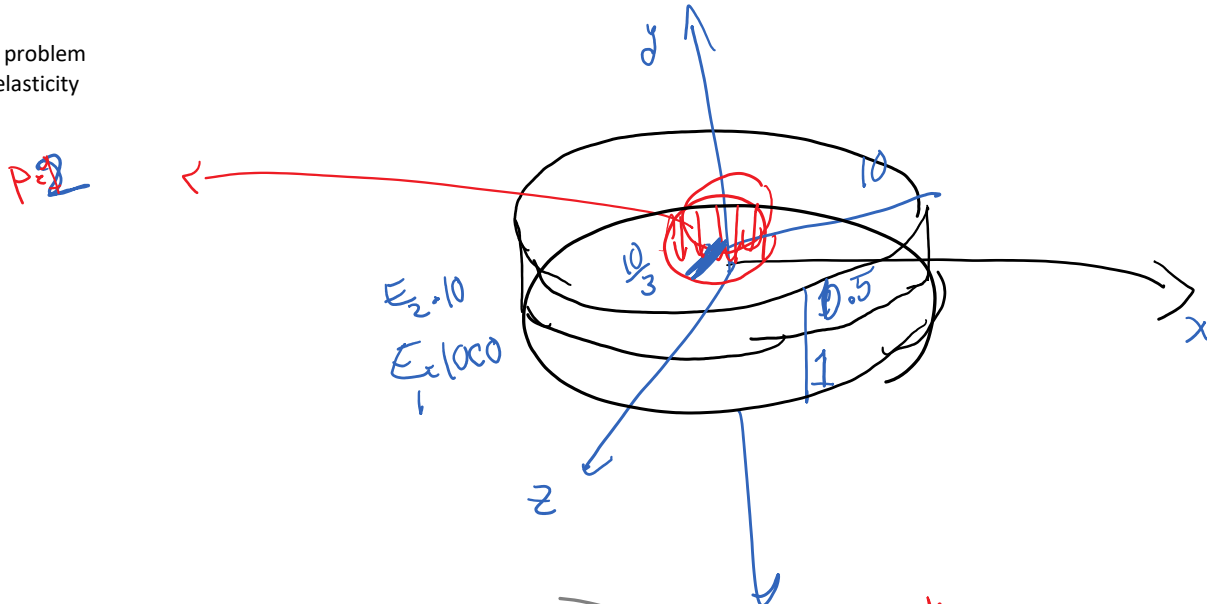
$$\Sigma F_1 = 0(\text{@}n_2) \Rightarrow R_1^2 + \frac{4}{5}T^{e2} = 0 \Rightarrow T^{e2} = -\frac{5}{7} = -0.7143 \quad (399d)$$

$$\Sigma F_2 = 0(\text{@}n_2) \Rightarrow T^{e1} + \frac{3}{5}T^{e2} = 0 \Rightarrow T^{e1} = \frac{3}{7} = 0.4286 \quad (399e)$$

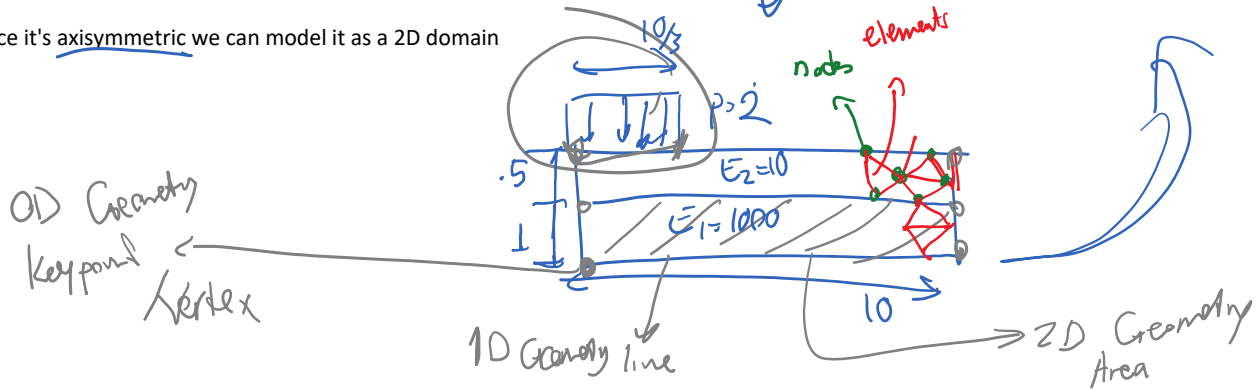
$$\Sigma F_1 = 0(\text{@}n_1) \Rightarrow -\frac{4}{5}T^{e2} - \frac{1}{\sqrt{2}}T^{e3} = 0 \Rightarrow T^{e3} = \frac{4}{7}\sqrt{2} = 0.8081 \quad (399f)$$

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2nd problem  
2D elasticity



Since it's axisymmetric we can model it as a 2D domain



Problems with 2D / 3D elements:

We create the geometry and apply BC (supports, loads), ... on geometry not on element nodes

Define the element

Defined Element Types:  
NONE DEFINED

Library of Element Types

Element type reference number: 1

1st order accurate

2nd order accurate

Defined Element Types:  
Type 1 PLANE183

PLANE183 element type options

Options for PLANE183, Element Type Ref. No. 1

Element shape: K1

Element behavior: K3

Element formulation: K6

(NOTE: Mixed formulation is not valid with plane stress)

Asymmetric

Choose this