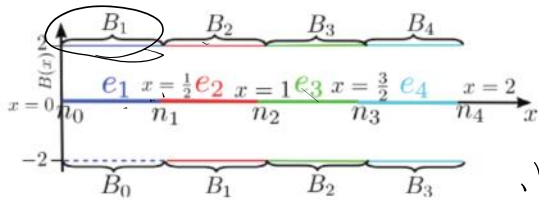


From last time

Local approach (element-centered)

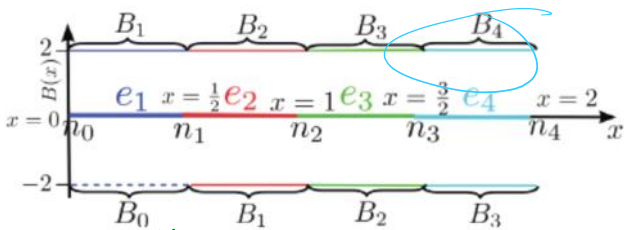


$$K^{e_1} = \int_{e_1} \begin{bmatrix} B_1 \\ B_2 \\ B_3 \\ B_4 \end{bmatrix} [B_1 \ B_2 \ B_3 \ B_4] dx$$

$$= \int_0^{0.5} \begin{bmatrix} 2 \\ 0 \\ 0 \\ 0 \end{bmatrix} [2 \ 0 \ 0 \ 0] dx = \begin{bmatrix} 2 & -2 & 0 & 0 \\ -2 & 2 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix}$$

$$K^{e_2} = \int_{e_2} \begin{bmatrix} B_1 \\ B_2 \\ B_3 \\ B_4 \end{bmatrix} [B_1 \ B_2 \ B_3 \ B_4] dx$$

$$= \int_{0.5}^1 \begin{bmatrix} -2 \\ 2 \\ 0 \\ 0 \end{bmatrix} [-2 \ 2 \ 0 \ 0] dx = \begin{bmatrix} 2 & -2 & 0 & 0 \\ -2 & 2 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix}$$



$$K^{e_3} = \int_{e_3} \begin{bmatrix} 0 \\ -2 \\ 2 \\ 0 \end{bmatrix} [0 \ -2 \ 2 \ 0] dx$$

$$= \begin{bmatrix} 0 & 0 & 0 & 0 \\ 0 & 2 & -2 & 0 \\ 0 & -2 & 2 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix}$$

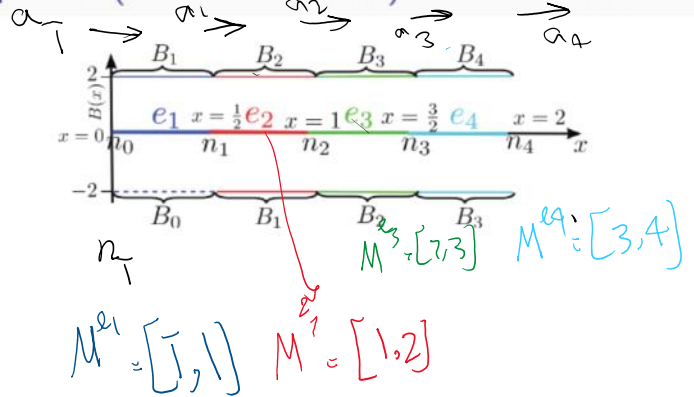
$$K^4 = \int_{e_4} \begin{bmatrix} 0 \\ 0 \\ -2 \\ 2 \end{bmatrix} [0 \ 0 \ 2 \ 2] dx = \begin{bmatrix} 0 & 0 \\ 0 & 2 & -2 \\ -2 & 2 & 0 \end{bmatrix}$$

All four elements have local stiffnesses

$$k^e = \frac{AE^e}{L^e} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$$

for all elements $AE^e = 1, L^e = \frac{2}{4} = 0.5$

Local approach (element-centered)



$$k^{e1}, k^{e2}, k^{e3}, k^{e4} = \begin{bmatrix} 2 & -2 \\ -2 & 2 \end{bmatrix}$$

$$k^e = \begin{bmatrix} 1 & 1 \\ 1 & 1 \end{bmatrix} \begin{bmatrix} 2 & -2 \\ -2 & 2 \end{bmatrix}$$

$$k^{e2} = \begin{bmatrix} 1 & 2 \\ 2 & 3 \end{bmatrix} \begin{bmatrix} 2 & -2 \\ -2 & 2 \end{bmatrix}$$

$$k^{e3} = \begin{bmatrix} 2 & 3 \\ 3 & 4 \end{bmatrix} \begin{bmatrix} 2 & -2 \\ -2 & 2 \end{bmatrix}$$

$$k^{e4} = \begin{bmatrix} 3 & 4 \\ 4 & 5 \end{bmatrix} \begin{bmatrix} 2 & -2 \\ -2 & 2 \end{bmatrix}$$

Assembly

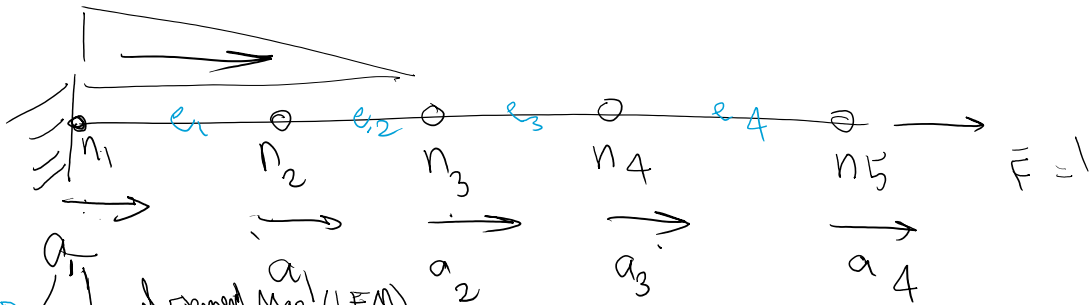
$$K = \begin{bmatrix} 2 & -2 & 0 & 0 \\ -2 & 2+2 & -2 & 0 \\ 0 & -2 & 2+2 & -2 \\ 0 & 0 & -2 & 2 \end{bmatrix}$$

$$K = \begin{bmatrix} 4 & -2 & 0 & 0 \\ -2 & 4 & -2 & 0 \\ 0 & -2 & 4 & -2 \\ 0 & 0 & -2 & 4 \end{bmatrix}$$

Before we solve the same bar example with this local (element-centered) approach, we discuss the concepts of node and dof.

$$n_f = 4$$

$$n_p = 1$$



Nodal Element Map (NEM)

$$NEM_{e1} = [1, 2]$$

$$NEM_{e2} = [2, 3]$$

$$NEM_{e4} = [4, 5]$$

the nodes the element has

Element

$$M_{e1} = [1, 1]$$

$$M_{e2} = [2, 2]$$

$$M_{e3} = [3, 3]$$

$$M_{e4} = [4, 4]$$

Element
dof Map (edof)

$$M_{e_1} = [1, 1]$$

$$M_{e_2} = [1, 2]$$

$$M_{e_3} = [3, 4]$$

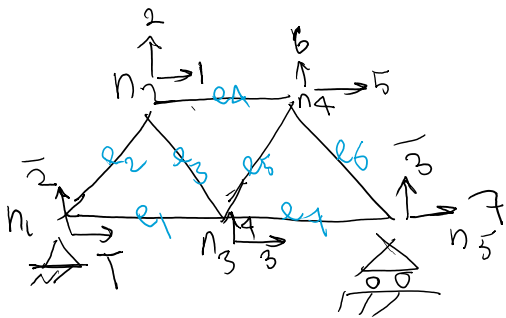
$$M_{e_4} = [4, 5]$$

nodes
element has

Node and dof numbers don't match even when there is one dof / node

Examples from problems that have more than 1 dof per node:

Ex 1, Truss

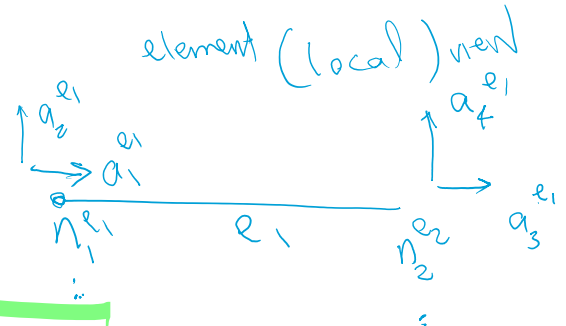


$$n_f = 7$$

$$n_p = 3$$

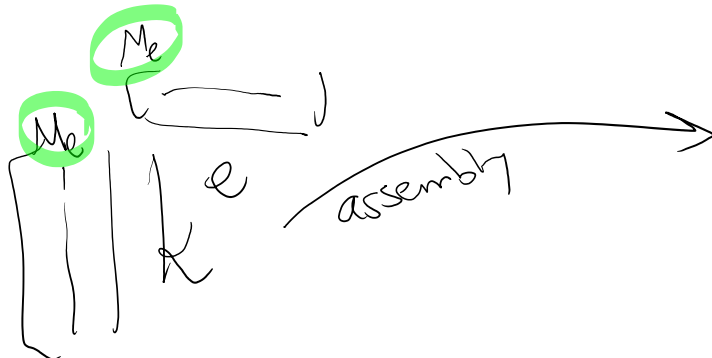
$$\frac{n_{dof}}{n_{node}} = 2$$

$$LEM_{e_1} = [1, 3] \text{ (nodes)}$$



Needed
for assembly

$$M_{e_1} = [1, 2, 3, 4] \text{ dof}$$



LEM is used in computer implementation to form M (edof)

plate
problem

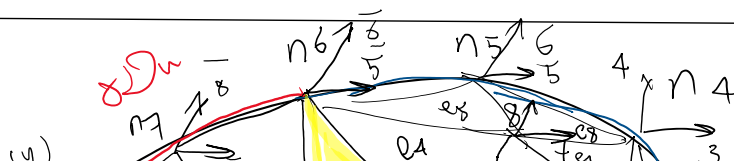
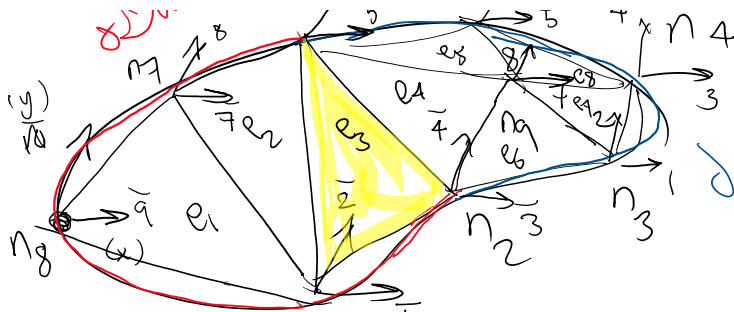


plate problem



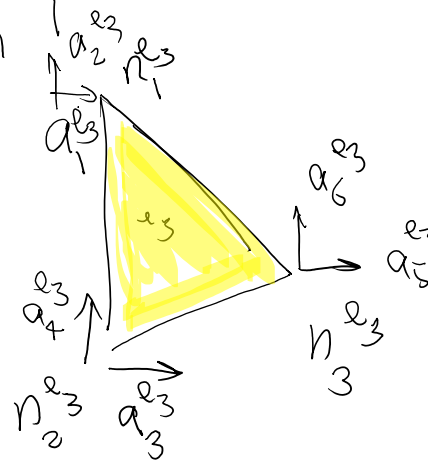
$n_f = 8$
 $n_p = 10$



$LEM_{e_3} = [6, 1, 2]$
(nodes)

$M_{e_3} = [5, 6, 1, 2, 3, 4]$

$\frac{n_{dof}}{n_{node}} \Rightarrow$

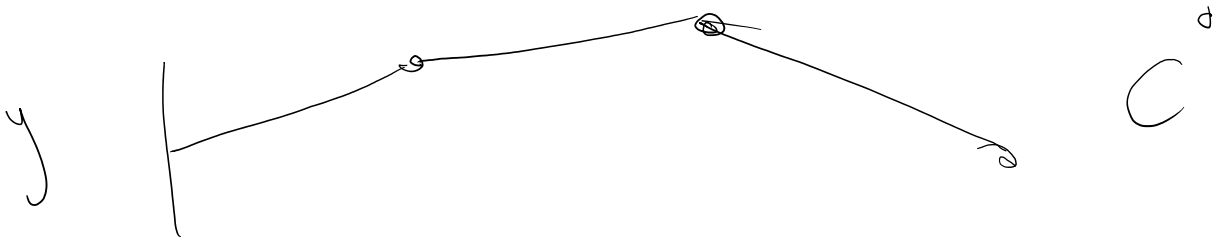


Ex3 Bending problem

$(EI y'')'' + q = 0$

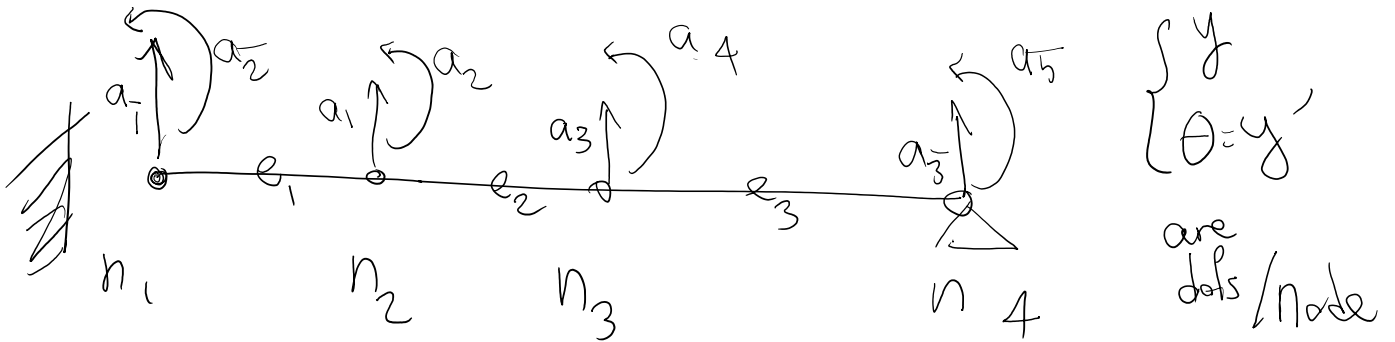
$M = 4 \quad m = \frac{M}{2} = 2$

$C^{m-1} = C^1$
global continuity needed

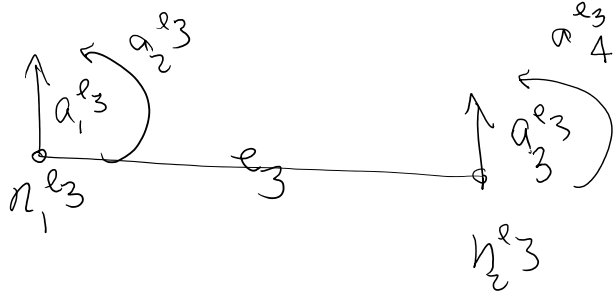




hat functions
don't work for a beam problem



$\frac{2 \text{ d.o.f}}{\text{node}} = 2$



$LEM_{e3} [3, 4]$ (nodes)

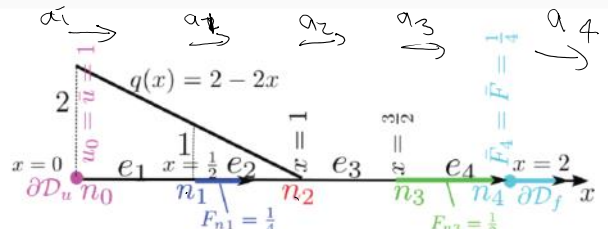
we need this for assembly

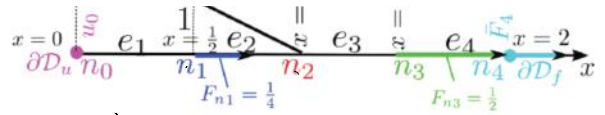
$M_{e3} = [3, 4, \bar{3}, 5]$

We solve this with a local approach and cover a few new things along the way:

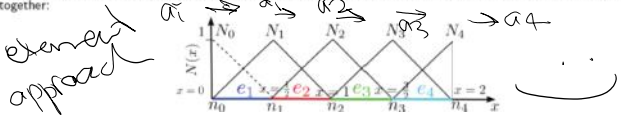
$LEM_{e1} [0, 1]$ $LEM_{e2} [1, 2]$ - -

Bar Example: Overview



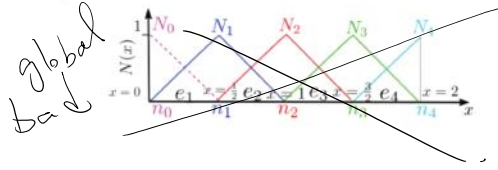


Local approach is element centered: As we eventually the form of the shape functions change element to element, it is more convenient to first divide the integration domain, calculate element level matrices and vectors, and add them together:



- We discretize the domain shown $D = [0, 2]$ to **four** elements e_1, e_2, e_3, e_4 .
- The problem has **five** nodes n_0, n_1, n_2, n_3, n_4 at $x = 0, \frac{1}{2}, 1, \frac{3}{2}$ and 2 respectively.
- Nodes $\{n_1, n_2, n_3, n_4\}$ are **free** $\Rightarrow n_f = 4$.
- Node n_0 is **prescribed** (on ∂D_u) with the value $\bar{u}_1 = \bar{u} = 1 \Rightarrow n_p = 1$.
- The material and section properties are chosen: $E = 1, A = 1$.

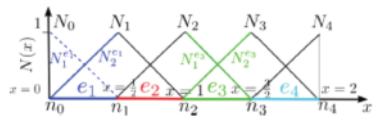
Finite element shape functions:



$$M_{e_1} = \begin{bmatrix} 1 & 1 \end{bmatrix} \quad M_{e_2} = \begin{bmatrix} 1 & 2 \end{bmatrix} \quad M_{e_3} = \begin{bmatrix} 2 & 3 \end{bmatrix} \quad M_{e_4} = \begin{bmatrix} 3 & 4 \end{bmatrix}$$

We already did the stiffness assembly:

Example: Formation of K and F



- Local stiffness matrix: Since $A = 1, E = 1, L = \frac{1}{2}$ for all elements in the example shown, local stiffness matrix is:

$$k^{e_i} = \begin{bmatrix} 2 & -2 \\ -2 & 2 \end{bmatrix}, i = 1, 2, 3, 4.$$

- Assembly to global system: Around the local stiffness for e_3 we have the corresponding dof of the local dof in the global system:

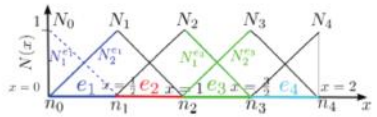
$$\begin{matrix} 2 & 3 \\ 2 & -2 \\ 3 & 2 \end{matrix}$$

This means that for example $k_{11}^{e_3}$ will be added to $K_{22}, k_{12}^{e_3}$ to K_{23} and so forth:

$$K^{e_3} = \begin{bmatrix} 0 & 0 & 0 & 0 \\ 0 & 2 & -2 & 0 \\ \text{sym.} & & 2 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix}$$

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Example: Formation of K and F



Similarly for element e_1 the local stiffness matrix k^{e_1} and corresponding **free** global dof are:

$$\begin{matrix} 0 & 1 \\ 0 & 2 & -2 \\ 1 & -2 & 2 \end{matrix}$$

Note that 0 in the first row and column means that dof 1 in the element does not correspond to any free global dofs (n_0 is a prescribed node on ∂D_u). In This case, only $k_{22}^{e_1} = 2$ is added to K_{11} from e_1 . The corresponding global level matrix from element e_1 is:

$$K^{e_1} = \begin{bmatrix} 2 & 0 & 0 & 0 \\ & 0 & 0 & 0 \\ \text{sym.} & & 0 & 0 \\ & & & 0 \end{bmatrix}$$

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- For example for our sample problem, these maps are,

$$\begin{aligned}
 M^{e1} &= [0 \ 1]^T & \bar{M}^{e1} &= [1 \ 0]^T \\
 M^{e2} &= [1 \ 2]^T & \bar{M}^{e2} &= [0 \ 0]^T \\
 M^{e3} &= [2 \ 3]^T & \bar{M}^{e3} &= [0 \ 0]^T \\
 M^{e4} &= [3 \ 4]^T & \bar{M}^{e4} &= [0 \ 0]^T
 \end{aligned}$$

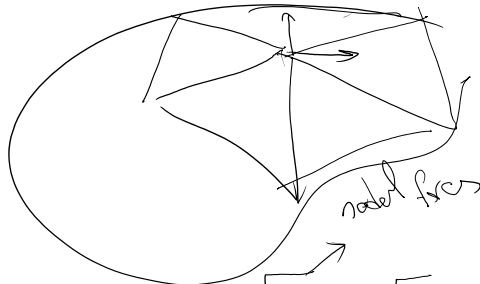
- We also combine both maps into M_i^e that includes both free and prescribed dofs:

$$M^{e1}_i = [\bar{1} \ 1]^T \quad M^{e2}_i = [1 \ 2]^T \quad M^{e3}_i = [2 \ 3]^T \quad M^{e4}_i = [3 \ 4]^T$$

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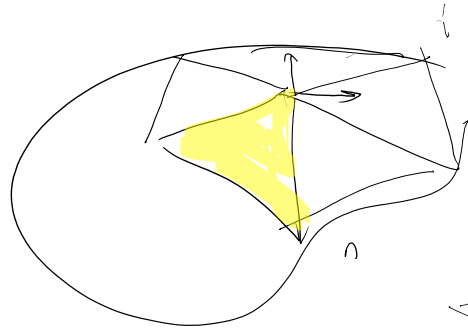
$$\mathbf{K} = \begin{bmatrix} 2+2 & -2 & 0 & 0 \\ -2 & 2+2 & -2 & 0 \\ 0 & -2 & 2+2 & -2 \\ 0 & 0 & -2 & 2 \end{bmatrix} = \begin{bmatrix} 4 & -2 & 0 & 0 \\ -2 & 4 & -2 & 0 \\ 0 & -2 & 4 & -2 \\ 0 & 0 & -2 & 2 \end{bmatrix}$$

We form the Element level force vector and assemble it to the global force vector



$$K_a \cdot f = F_n + f_e$$

from elements



k^e assembled to the global structure

$$f_{ee} = f_r + f_N - f_D$$

Also assembled to the global force F_e

$$F_e = F_r + F_N - F_D$$

↓
↓
↓

Source term
Neumann BC (2D/3D elements)
Dirichlet BC

$$F_D = K_{fp} a_p$$

$$f_D = k_a^e \quad \text{for any element}$$

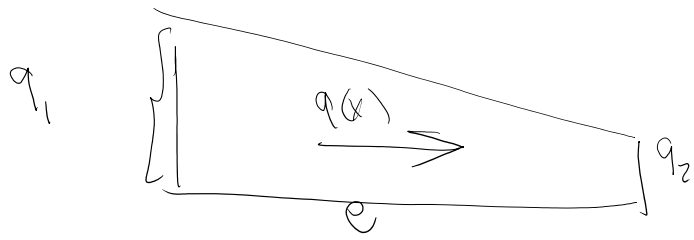
I'll show you

$$f_r^e = \frac{1}{6} \begin{bmatrix} 2 & 1 \\ 1 & 2 \end{bmatrix} \begin{bmatrix} q_1 \\ q_2 \end{bmatrix}$$

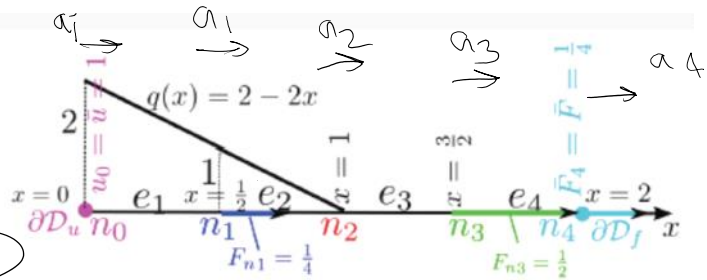
I'll show you

1D bar

$$F_f = \frac{1}{6} [1 \quad 2] [q_1 \quad q_2]$$



We already have the global stiffness matrix.
Let's calculate the global force vector:



$$K a = F_n + F_e$$

node
next
node
element

$$F_n = \begin{bmatrix} 1/4 \\ 0 \\ 1/2 \\ 1/4 \end{bmatrix}$$

$$F_e = F_f + \cancel{F_n} - F_D$$

0 for 1D elements

F_e : calculate f_e^e for all elements

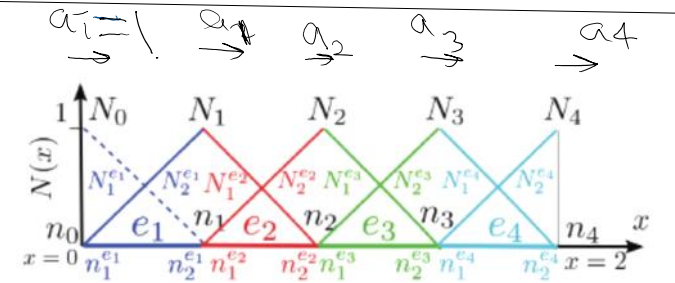
$$f_e^e = f_n^e + \cancel{f_n^e} - f_D^e$$

(1D)

$$f_e^e = k^{e_1, e_1} a = \begin{bmatrix} 2 & -2 \\ -2 & 2 \end{bmatrix} \begin{bmatrix} a_1 \\ a_2 \end{bmatrix}$$

$$= \begin{bmatrix} 2 & -2 \\ -2 & 2 \end{bmatrix} \begin{bmatrix} a_1 \\ a_1 \end{bmatrix} = \begin{bmatrix} 2 & -2 \\ -2 & 2 \end{bmatrix} \begin{bmatrix} 1 \\ 1 \end{bmatrix}$$

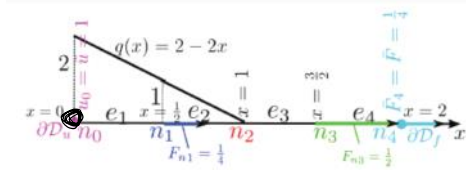
use use



$$M_{e_1} = \begin{bmatrix} 1 & 1 \end{bmatrix}$$



use use zero instead



$$f_D^{e_1} = \begin{bmatrix} 2 & -2 \\ -2 & 2 \end{bmatrix} \begin{bmatrix} 1 \\ 0 \end{bmatrix} = \begin{bmatrix} 2 \\ -2 \end{bmatrix}$$

free dof are put zero

distributed forces at the nodes of element

$$f_r^{e_1} = \frac{L_{e_1}}{6} \begin{bmatrix} 2 & 1 \\ 1 & 2 \end{bmatrix} \begin{bmatrix} q_1 \\ q_2 \end{bmatrix} = \frac{1}{6} \begin{bmatrix} 2 & 1 \\ 1 & 2 \end{bmatrix} \begin{bmatrix} 2 \\ 1 \end{bmatrix} = \frac{1}{12} \begin{bmatrix} 5 \\ 4 \end{bmatrix}$$

$$f^{e_1} = f_r^{e_1} + \cancel{f_D^{e_1}} - f_D = \frac{1}{12} \begin{bmatrix} 5 \\ 4 \end{bmatrix} - \begin{bmatrix} 2 \\ -2 \end{bmatrix} = \begin{bmatrix} -19/12 \\ 28/12 \end{bmatrix}$$



assembly

$$F_e = \begin{bmatrix} 1 \\ 2 \\ 3 \\ 4 \end{bmatrix} \begin{bmatrix} 7/3 \end{bmatrix}$$

$$M_{e_1} = \begin{bmatrix} 1 & 1 \end{bmatrix}$$