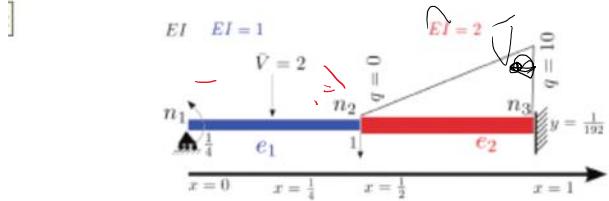


## Beam example: Assembly of global system



dof  
 $\bar{a}_1 = 0$   
 $\bar{a}_2 = \frac{1}{192}$   
 $\bar{a}_3 = 0$

forces  
 $F_{n1} = \frac{1}{4}$   
 $F_{n3} = 0$   
 $F_{n2} = -1$   
 $\bar{F}_{n3}$

1D elements

$$\mathbf{f}_e^e = \mathbf{f}_f + \mathbf{f}_N - \mathbf{f}_D$$

$$\mathbf{f}_D = \mathbf{k}_e^e \mathbf{a}$$

e	$e_1$	$e_2$
$\mathbf{k}_e^e$	$\mathbf{k}^{e_1} = \frac{1}{(\frac{1}{2})^3} \begin{bmatrix} 12 & 3 & -12 & 3 \\ 3 & 1 & -3 & 0.5 \\ -12 & -3 & 12 & -3 \\ 3 & 0.5 & -3 & 1 \end{bmatrix}$ $= \begin{bmatrix} 96 & 24 & -96 & 24 \\ 24 & 8 & -24 & 8 \\ -96 & -24 & 96 & -24 \\ 24 & 4 & -24 & 8 \end{bmatrix}$	$\mathbf{k}^{e_2} = \frac{2}{(\frac{1}{2})^3} \begin{bmatrix} 12 & 3 & -12 & 3 \\ 3 & 1 & -3 & 0.5 \\ -12 & -3 & 12 & -3 \\ 3 & 0.5 & -3 & 1 \end{bmatrix}$ $= \begin{bmatrix} 192 & 48 & -192 & 48 \\ 48 & 16 & -48 & 8 \\ -192 & -48 & 192 & -48 \\ 48 & 8 & -48 & 16 \end{bmatrix}$
$\mathbf{f}_r^e$	$(440) \text{ (1st eqn)} (\dot{\xi}_0 = 0)$ $\tilde{V} \begin{bmatrix} N_1^{e_1}(\xi_0) \\ N_2^{e_1}(\xi_0) \\ N_3^{e_1}(\xi_0) \\ N_4^{e_1}(\xi_0) \end{bmatrix} = -2 \begin{bmatrix} \frac{1}{2} \\ \frac{2}{8} \\ \frac{1}{8} \\ -\frac{2}{8} \end{bmatrix} = \begin{bmatrix} \frac{1}{2} \\ \frac{1}{4} \\ \frac{1}{8} \\ -\frac{1}{4} \end{bmatrix}$	$\text{equation (433), } \mathbf{r}^e \begin{bmatrix} q_1 \\ q_2 \end{bmatrix}^T$ $\begin{bmatrix} \frac{7}{20} \\ \frac{3}{20} \\ \frac{7}{20} \\ \frac{1}{20} \end{bmatrix} \begin{bmatrix} 0 \\ 10 \end{bmatrix} = \begin{bmatrix} \frac{3}{2} \\ \frac{4}{2} \\ \frac{1}{2} \\ -\frac{1}{2} \end{bmatrix}$
$\mathbf{f}_D^e$	$\mathbf{k}^{e_1} \mathbf{a}_1^e = \begin{bmatrix} 96 & 24 & -96 & 24 \\ 24 & 8 & -24 & 8 \\ -96 & -24 & 96 & -24 \\ 24 & 4 & -24 & 8 \end{bmatrix} \begin{bmatrix} 0 \\ 0 \\ 0 \\ 0 \end{bmatrix} = \begin{bmatrix} 1 & 0 \\ 2 & 0 \\ 3 & 0 \end{bmatrix}$	$\mathbf{k}^{e_2} \mathbf{a}_2^e = \begin{bmatrix} 192 & 48 & -192 & 48 \\ 48 & 16 & -48 & 8 \\ -192 & -48 & 192 & -48 \\ 48 & 8 & -48 & 16 \end{bmatrix} \begin{bmatrix} 0 \\ 0 \\ 0 \\ 0 \end{bmatrix} = \begin{bmatrix} 2 & \frac{7}{4} \\ 3 & \frac{1}{4} \\ 2 & \frac{1}{4} \\ 3 & -\frac{1}{4} \end{bmatrix}$
$\mathbf{f}_e^e$	$\mathbf{f}_e^e = \mathbf{f}_r^e + \mathbf{f}_N^e - \mathbf{f}_D^e = \begin{bmatrix} 1 & -\frac{1}{8} \\ 2 & -1 \\ 3 & \frac{1}{8} \end{bmatrix}$	$\mathbf{f}_e^e = \mathbf{f}_r^e + \mathbf{f}_N^e - \mathbf{f}_D^e = \begin{bmatrix} 2 & \frac{7}{4} \\ 3 & \frac{1}{4} \\ 2 & \frac{1}{4} \\ 3 & -\frac{1}{4} \end{bmatrix}$

$$\mathbf{k}^e = \frac{EI}{L^3} \begin{bmatrix} 12 & 6L^e & -12 & 6L^e \\ 4L^e & 12 & -6L^e & 4L^e \\ \text{sym.} & & 12 & -6L^e \\ & & & 4L^e \end{bmatrix} \quad \text{for constant } E \text{ and } I \quad (427)$$

$$\begin{aligned} \mathbf{k} &= \begin{bmatrix} 8 & -24 & 4 \\ 96+192 & 288 & 24 \\ -24+48 & 24 & 8+16 \end{bmatrix} \\ &= \begin{bmatrix} 8 & -24 & 4 \\ 288 & 24 & 24 \\ \text{sym.} & & \end{bmatrix} \\ \mathbf{F} &= \mathbf{F}_n + \mathbf{F}_e \\ &= \begin{bmatrix} \frac{1}{4} \\ 0 \\ 0 \end{bmatrix} + \begin{bmatrix} -\frac{1}{8} \\ \frac{7}{4} \\ -\frac{1}{4} \end{bmatrix} = \begin{bmatrix} -\frac{1}{8} \\ \frac{11}{4} \\ -\frac{1}{4} \end{bmatrix} \Rightarrow \end{aligned}$$

$$\boxed{\mathbf{f}_r^e \approx \mathbf{r}^e \begin{bmatrix} q_1 \\ q_2 \end{bmatrix} \quad \text{where } \mathbf{r}^e = L^e \begin{bmatrix} \frac{7}{20} \\ \frac{1}{20} \\ \frac{3}{20} \\ -\frac{1}{20} \end{bmatrix} \quad \text{exact for linear } q} \quad (433)$$

$$\mathbf{K} = \begin{bmatrix} 8 & -24 & 4 \\ -24 & 96+192 & -24+48 \\ 4 & -24+48 & 8+16 \end{bmatrix}$$

$$= \begin{bmatrix} 8 & -24 & 4 \\ -24 & 288 & 24 \\ 4 & 24 & 24 \end{bmatrix}$$

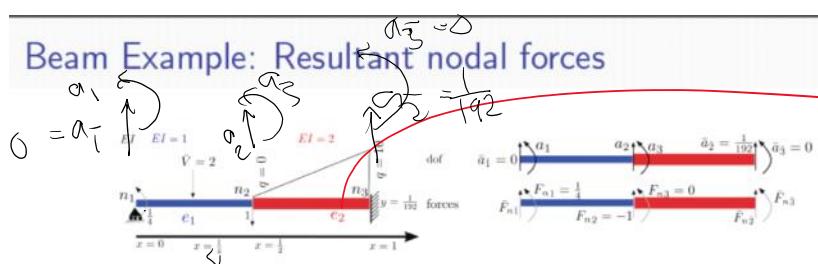
$$\mathbf{F} = \mathbf{F}_n + \mathbf{F}_e$$

$$= \begin{bmatrix} \frac{1}{4} \\ -\frac{1}{1} \\ 0 \end{bmatrix} + \begin{bmatrix} -\frac{1}{8} \\ -1+\frac{7}{3} \\ \frac{1}{9}+\frac{1}{3} \end{bmatrix} = \begin{bmatrix} -\frac{1}{8} \\ -\frac{10}{3} \\ \frac{11}{9} \end{bmatrix} \Rightarrow$$

$$\boxed{\mathbf{U} = \mathbf{K}^{-1} \mathbf{F}}$$

$$= \begin{bmatrix} a_1 \\ a_2 \\ a_3 \end{bmatrix} = \begin{bmatrix} -\frac{7}{144} \\ -\frac{144}{6912} \\ \frac{11}{6912} \end{bmatrix} = \begin{bmatrix} -.00607 \\ -.00332 \\ .029434 \end{bmatrix}$$

### Beam Example: Resultant nodal forces

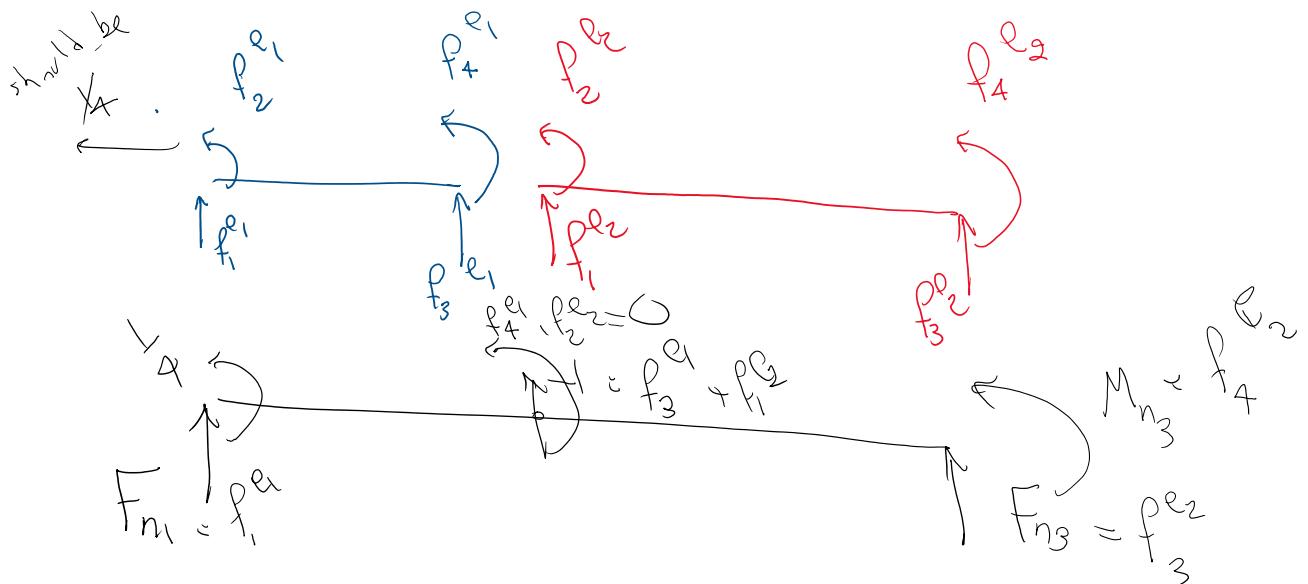


$$\mathbf{M}_{n_2} = [2/3, 1/2, 1/3]$$

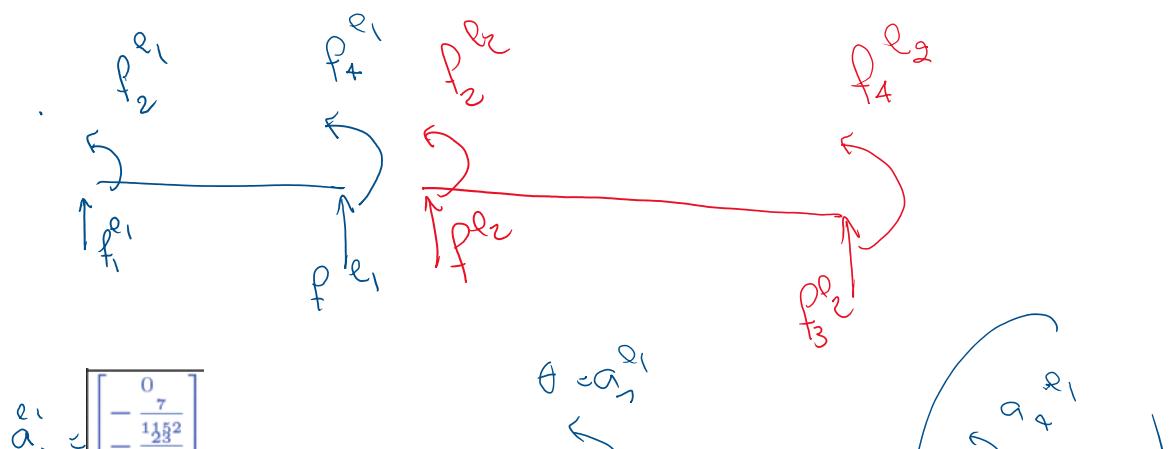
$$\mathbf{a}_2 = \left[ \begin{array}{c} 0 \\ 0 \\ \frac{1}{192} \end{array} \right]$$

$$\mathbf{a}_2 = \left[ -.00332, .029434, \frac{1}{192} \right]$$

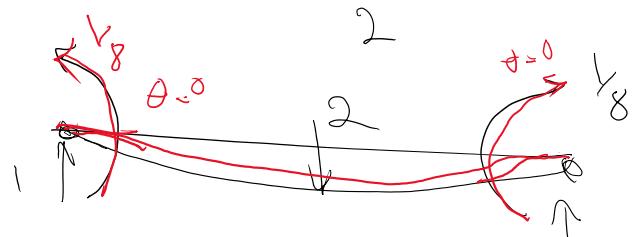
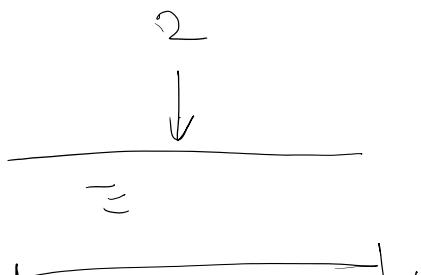
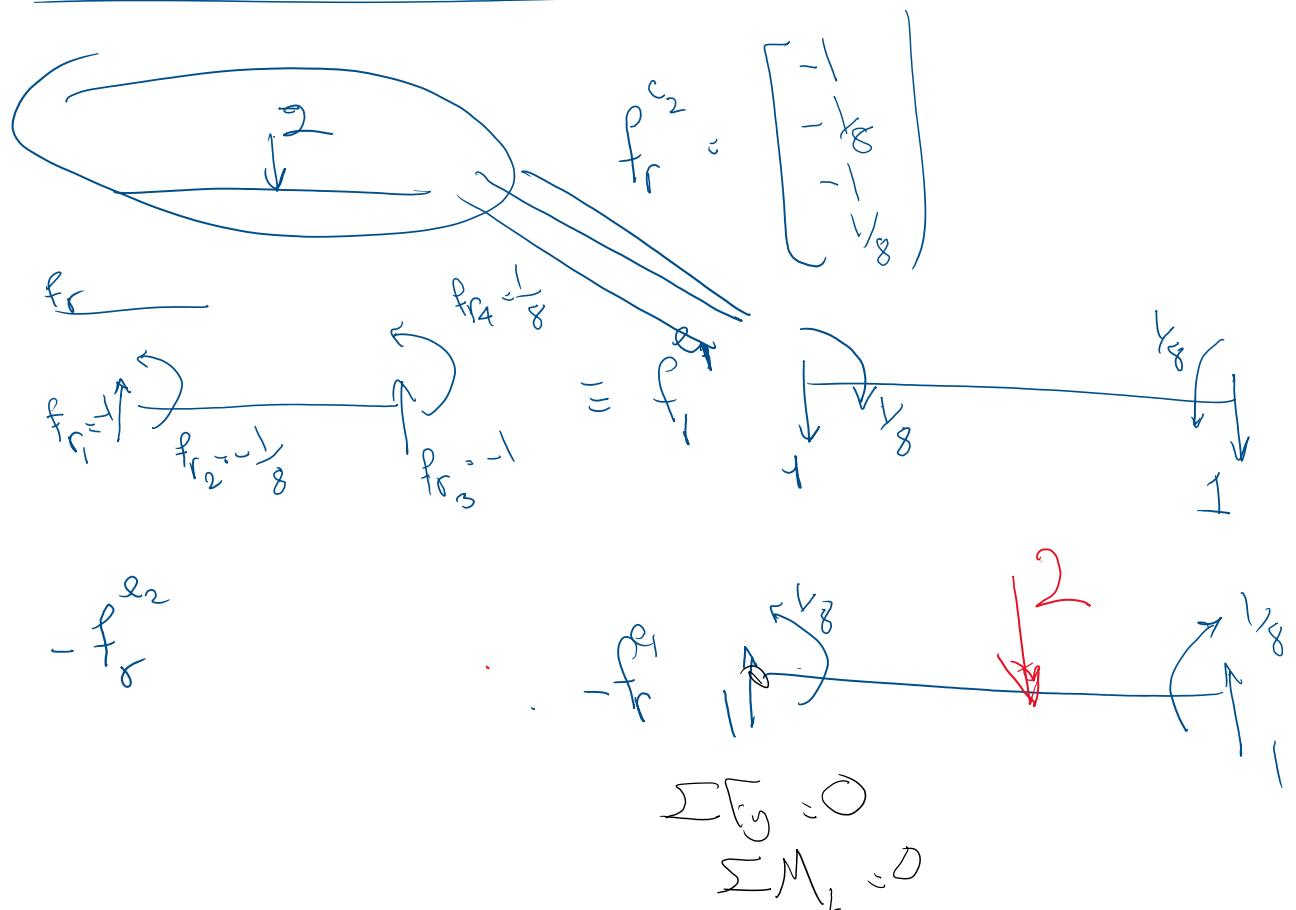
e	$\mathbf{e}_1$	$\mathbf{e}_2$
$\mathbf{u}^e$	$\begin{bmatrix} 0 \\ 0 \\ \frac{7}{144} \\ -\frac{144}{6912} \\ \frac{11}{6912} \\ \frac{1}{192} \end{bmatrix}$	$\begin{bmatrix} 0 \\ 0 \\ \frac{7}{144} \\ -\frac{144}{6912} \\ \frac{11}{6912} \\ 0 \end{bmatrix}$

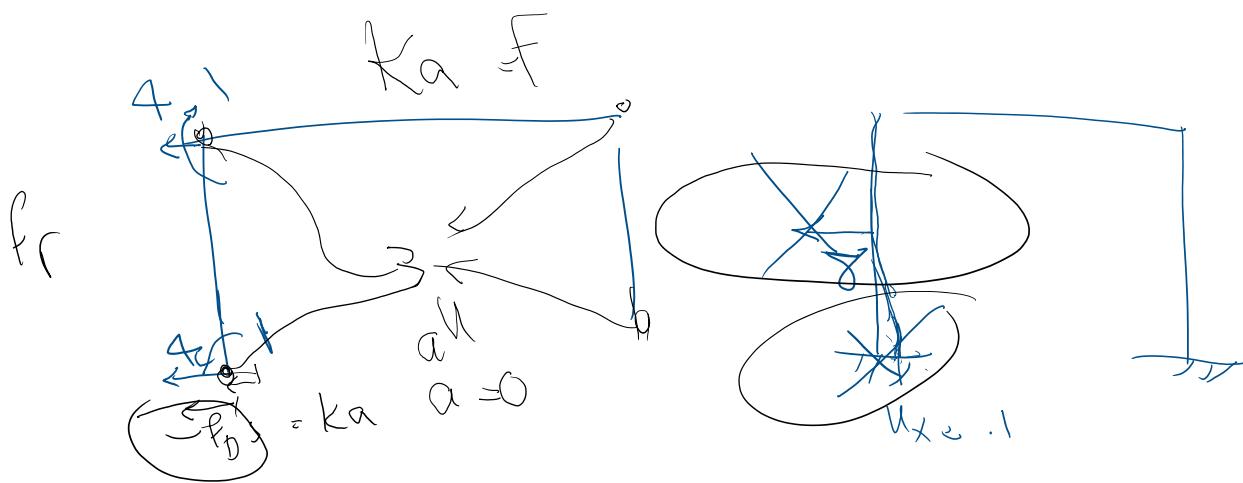
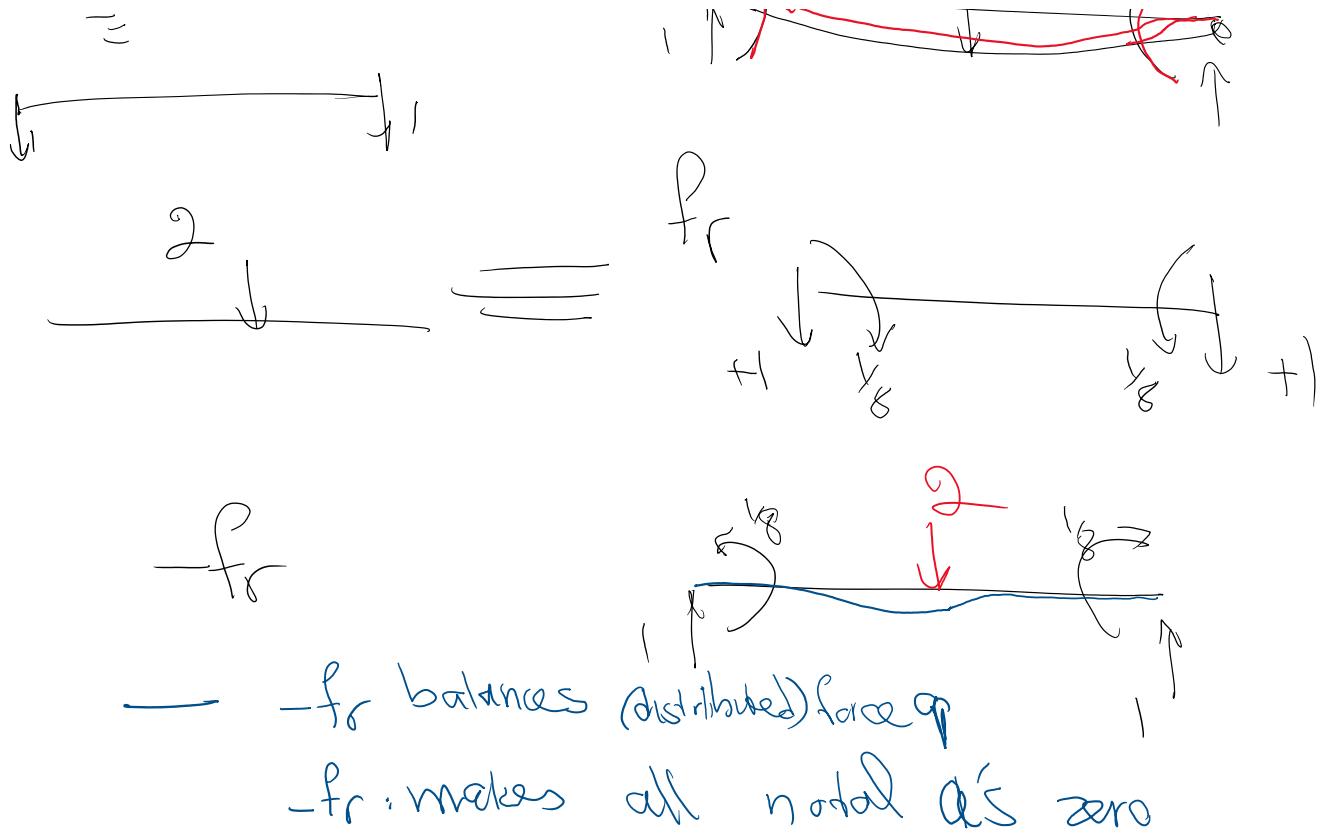


Basically, once we get the nodal forces, we can calculate the support forces:

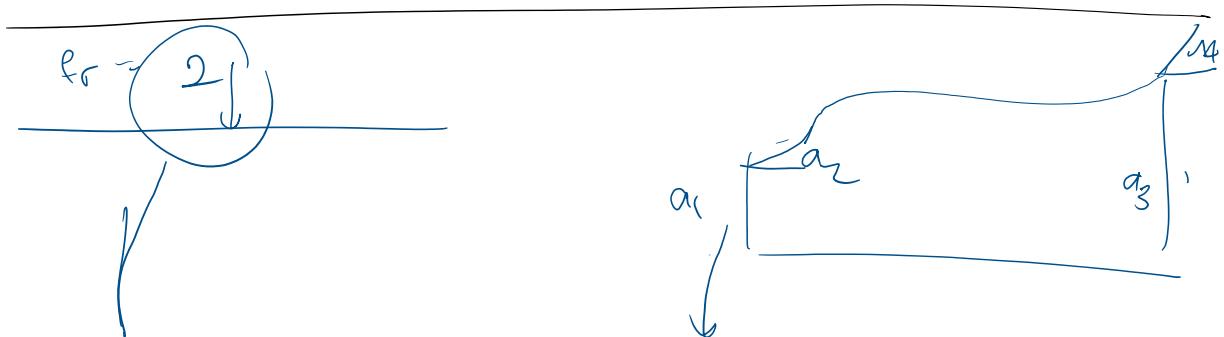


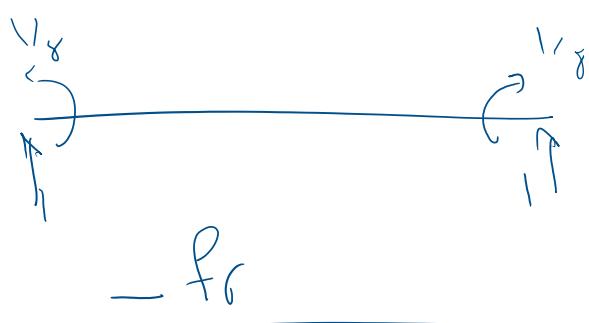
The diagram illustrates the assembly of a finite element stiffness matrix. It shows a global coordinate system with nodes labeled  $e_1$ ,  $e_2$ ,  $e_3$ , and  $e_4$ . A local coordinate system is defined by vectors  $a_1^{e_1}$ ,  $a_2^{e_1}$ , and  $a_3^{e_1}$  originating from node  $e_1$ . The angle between  $a_1^{e_1}$  and the horizontal axis is  $\theta = \alpha_1^{e_1}$ . A transformation matrix  $T = [a_1^{e_1} \ a_2^{e_1}]$  is shown, relating local coordinates to global coordinates. A local stiffness matrix  $K^{e_1}$  is associated with node  $e_1$ . The global stiffness matrix  $K$  is assembled by summing the contributions from all nodes, as indicated by the summation symbol  $\Sigma$ .





$$f_e = f_r - f_D$$





$$f_e = k_a a_e$$

actual nodal force is

$$= - (f_r - f_b)$$

$$f_r + f_b$$

$$f_e = f_r + f_b - f_D - k_a a_e$$

that we assembled to  
global force

$$f_e$$

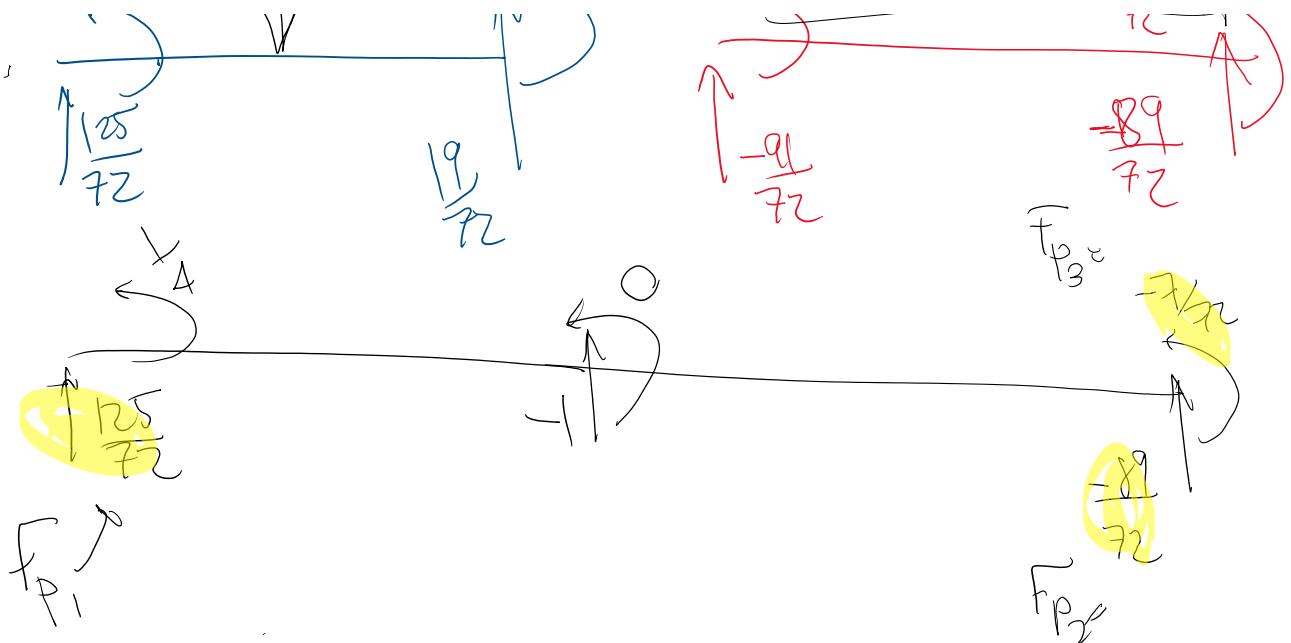
is acting on nodes of element

$$= -f_r - f_N + f_D$$

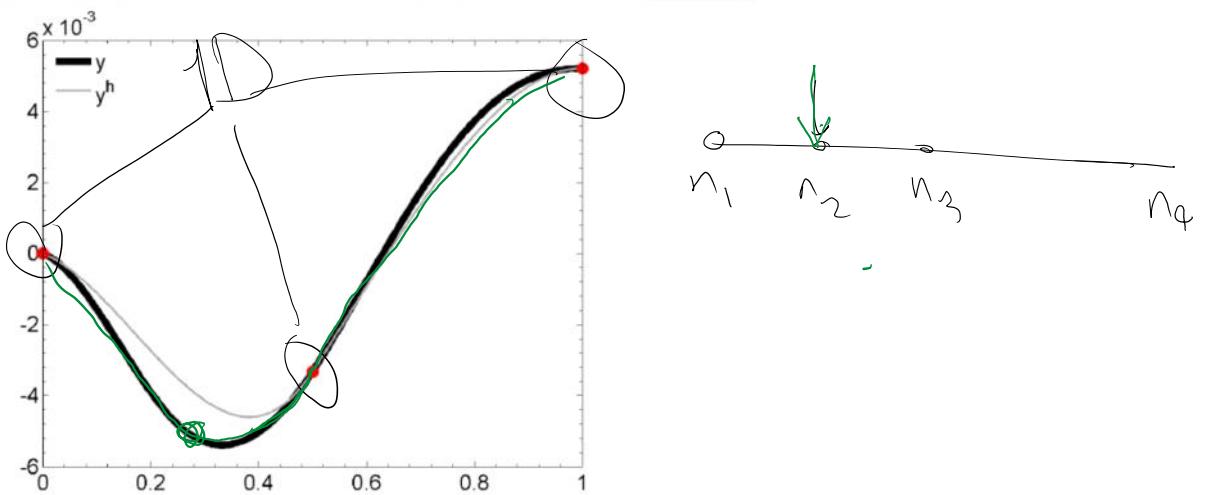
$$k_a$$

$$\begin{aligned} -f_e &= f_r - f_N - f_D \\ \begin{bmatrix} 96 & 24 & -96 & 24 \\ 24 & 8 & -24 & 4 \\ -96 & -24 & 96 & -24 \\ 24 & 4 & -24 & 8 \end{bmatrix} & \begin{bmatrix} 0 \\ -\frac{7}{128} \\ -\frac{1152}{6912} \\ \frac{3}{128} \end{bmatrix} - \begin{bmatrix} -1 \\ -\frac{1}{8} \\ -1 \\ \frac{1}{8} \end{bmatrix} = \begin{bmatrix} \frac{125}{72} \\ \frac{1}{72} \\ \frac{1}{144} \\ \frac{17}{144} \end{bmatrix} \\ \begin{bmatrix} 192 & 48 & -192 & 48 \\ 48 & 16 & -48 & 8 \\ -192 & -48 & 192 & -48 \\ 48 & 8 & -48 & 16 \end{bmatrix} & \begin{bmatrix} -\frac{23}{192} \\ \frac{6912}{192} \\ \frac{128}{192} \\ 0 \end{bmatrix} - \begin{bmatrix} \frac{3}{4} \\ \frac{4}{7} \\ \frac{12}{7} \\ -\frac{1}{8} \end{bmatrix} = \begin{bmatrix} -\frac{91}{72} \\ \frac{144}{89} \\ -\frac{72}{72} \\ -\frac{7}{72} \end{bmatrix} \end{aligned}$$



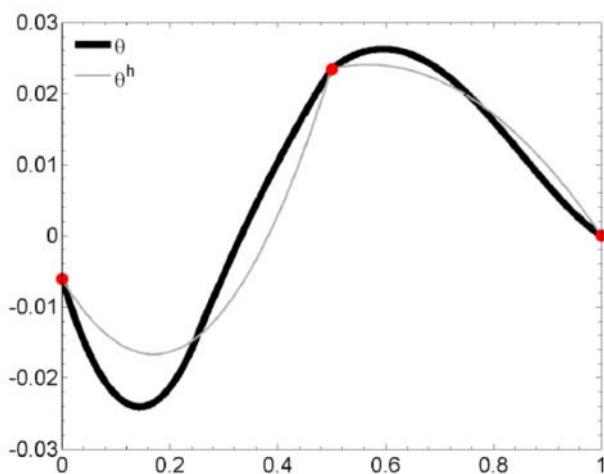


### Beam Example: FEM vs. Exact solution: $y$



- Similar to bar element **FEM** and exact solutions match at nodes.
- This behavior is restricted to certain problems in 1D with constant material properties along the element and does not extend to more general cases.

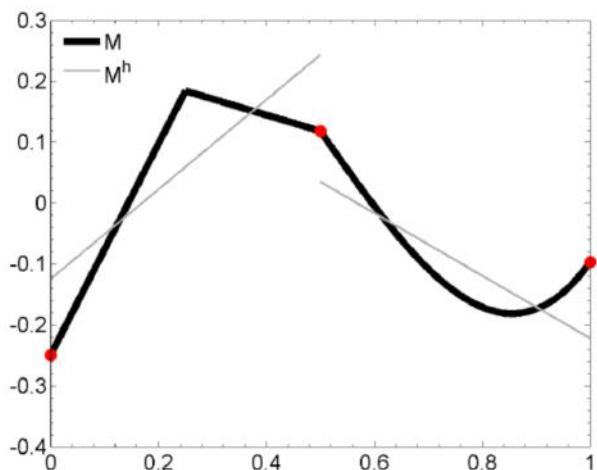
## Beam Example: FEM vs. Exact solution: $\theta$



- Rotations at nodes (rotational dofs) match those from exact solution.
- Again we emphasize that while this behavior is shared for certain types of 1D problems, it does not extend to more general cases.

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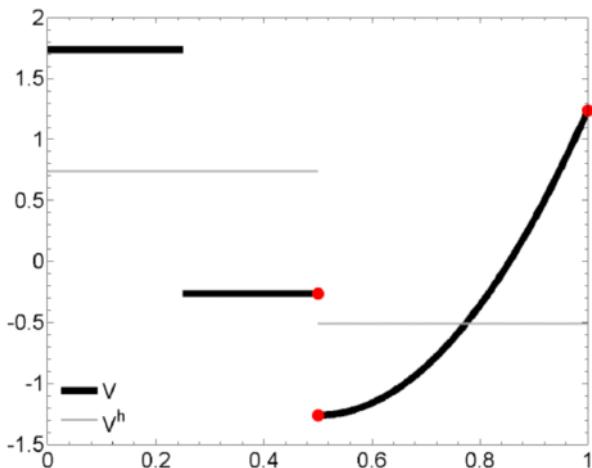
## Beam Example: FEM vs. Exact solution: $M$



- It is clear that FEM solutions for  $M$  are much less accurate than those for  $y$  and  $\theta$  when compared to exact solution.
- This is a **general behavior where FEM accuracy decreases for solution derivatives**.

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## Beam Example: FEM vs. Exact solution: $V$



order 3 element

$$\| u - u^{\text{exact}} \|_2 = Ch^{P+1}$$

$$\| u' - u^{\text{exact}} \| = Ch^{P+1-1}$$

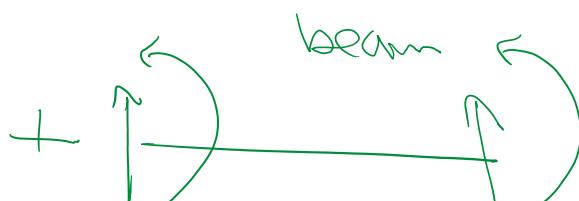
$$\| \bar{u}^{(s)} - u^{\text{exact}(s)} \| = Ch^{P-s}$$

Frame



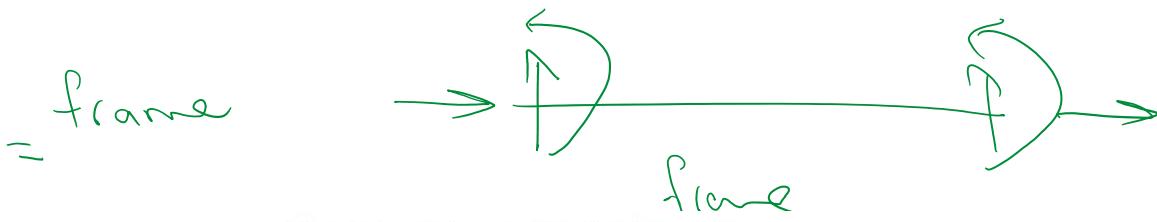
- Axial deformation (in local coord.)

$$\frac{EA}{L} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix} \begin{bmatrix} \bar{u}_1 \\ \bar{u}_2 \end{bmatrix} = \begin{bmatrix} f_{\bar{x}1} \\ f_{\bar{x}2} \end{bmatrix}$$



- Beam bending

$$\frac{EI}{L^3} \begin{bmatrix} 12 & 6L & -12 & 6L \\ 6L & 4L^2 & -6L & 2L^2 \\ -12 & -6L & 12 & -6L \\ 6L & 2L^2 & -6L & 4L^2 \end{bmatrix} \begin{bmatrix} \bar{v}_1 \\ \bar{\theta}_1 \\ \bar{v}_2 \\ \bar{\theta}_2 \end{bmatrix} = \begin{bmatrix} f_{\bar{y}1} \\ \bar{c}_1 \\ f_{\bar{y}2} \\ \bar{c}_2 \end{bmatrix}$$

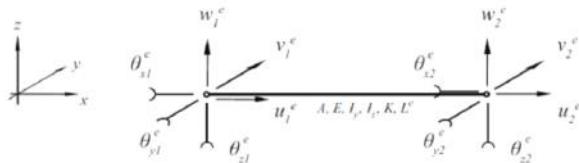


- Element matrix equation (local coord.)

$$\begin{bmatrix} a_1 & 0 & 0 & -a_1 & 0 & 0 \\ 0 & 12a_2 & 6La_2 & 0 & -12a_2 & 6La_2 \\ 0 & 6La_2 & 4L^2a_2 & 0 & -6La_2 & 2L^2a_2 \\ -a_1 & 0 & 0 & a_1 & 0 & 0 \\ 0 & -12a_2 & -6La_2 & 0 & 12a_2 & -6La_2 \\ 0 & 6La_2 & 2L^2a_2 & 0 & -6La_2 & 4L^2a_2 \end{bmatrix} \begin{bmatrix} \bar{u}_1 \\ \bar{v}_1 \\ \bar{\theta}_1 \\ \bar{u}_2 \\ \bar{v}_2 \\ \bar{\theta}_2 \end{bmatrix} = \begin{bmatrix} \bar{f}_{x1} \\ \bar{f}_{y1} \\ \bar{c}_1 \\ \bar{f}_{x2} \\ \bar{f}_{y2} \\ \bar{c}_2 \end{bmatrix}$$

$a_1 = \frac{EA}{L}$   
 $a_2 = \frac{EI}{L^3}$

### 3D frame element: local system



A 3D 12-freedom beam element defined in a local system

source: B. Torstenfelt; <http://www.solid.iei.liu.se/Education/TMHL02/Book-Bars%20and%20Beams.pdf>

- 12 local dofs are:

$$\mathbf{a}^{eT} = \{u_1^e \ v_1^e \ w_1^e \ \theta_{x1}^e \ \theta_{y1}^e \ \theta_{z1}^e \ u_2^e \ v_2^e \ w_2^e \ \theta_{x2}^e \ \theta_{y2}^e \ \theta_{z2}^e\} \quad (445)$$

- They correspond to axial and bending displacements, and torsional and bending rotations:

$$u_1^e \ v_1^e \ w_1^e \ \theta_{x1}^e \ \theta_{y1}^e \ \theta_{z1}^e \quad \text{dofs of node } n_1 \quad (446a)$$

$$u_2^e \ v_2^e \ w_2^e \ \theta_{x2}^e \ \theta_{y2}^e \ \theta_{z2}^e \quad \text{dofs of node } n_2 \quad (446b)$$

$$\{u_1^e \ u_2^e\} \quad \text{axial displacements} \quad (446c)$$

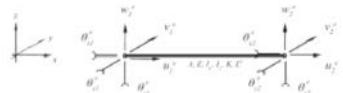
$$\{v_1^e \ v_2^e \ w_1^e \ w_2^e\} \quad \text{vertical displacements} \quad (446d)$$

$$\{\theta_{x1}^e \ \theta_{x2}^e\} \quad \text{axial (torsional) rotation} \quad (446e)$$

$$\{\theta_{y1}^e \ \theta_{z1}^e \ \theta_{y2}^e \ \theta_{z2}^e\} \quad \text{bending rotations} \quad (446f)$$

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### 3D frame element: local system stiffness matrix



A 3D 12-freedom beam element defined in a local system

$$K^e = \begin{bmatrix} a_1 & 0 & 0 & 0 & 0 & 0 & -a_1 & 0 & 0 & 0 & 0 & 0 \\ b_1 & 0 & 0 & 0 & b_2 & 0 & -b_1 & 0 & 0 & 0 & 0 & b_2 \\ c_1 & 0 & -c_2 & 0 & 0 & 0 & -c_1 & 0 & -c_2 & 0 & 0 & 0 \\ d_1 & 0 & 0 & 0 & 0 & 0 & 0 & -d_1 & 0 & 0 & 0 & 0 \\ e_1 & 0 & 0 & 0 & c_2 & 0 & 0 & c_4 & 0 & 0 & 0 & 0 \\ f_1 & 0 & -b_2 & 0 & 0 & 0 & 0 & 0 & 0 & b_4 & 0 & 0 \\ g_1 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ h_1 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ i_1 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ j_1 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ k_1 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ l_1 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \end{bmatrix}$$

where

$$a_1 = EA/L^e \quad d_1 = GK/L^e$$

$$b_1 = 12EI_z/L^{e^3} \quad b_2 = 6EI_z/L^{e^2} \quad b_3 = 4EI_z/L^e \quad b_4 = 2EI_z/L^e$$

$$c_1 = 12EI_y/L^{e^3} \quad c_2 = 6EI_y/L^{e^2} \quad c_3 = 4EI_y/L^e \quad c_4 = 2EI_y/L^e$$

source: B. Torstenfelt; <http://www.solid.iei.liu.se/Education/TMHL02/Book-Bars%20and%20Beams.pdf>

① **EA** contributes to **axial displacement**.

② **EI<sub>y</sub>** and **EI<sub>z</sub>** contributes to **bending vertical displacement and rotations**.

③ **GK** contributes to **axial (torsional) rotation**. **G** is the torsional modulus and **K** is torsional proportional constant for the cross section.

$GK$  contributes to axial (torsional) rotation.  $G$  is the torsional modulus and  $K$  is torsional proportional constant for the cross section.

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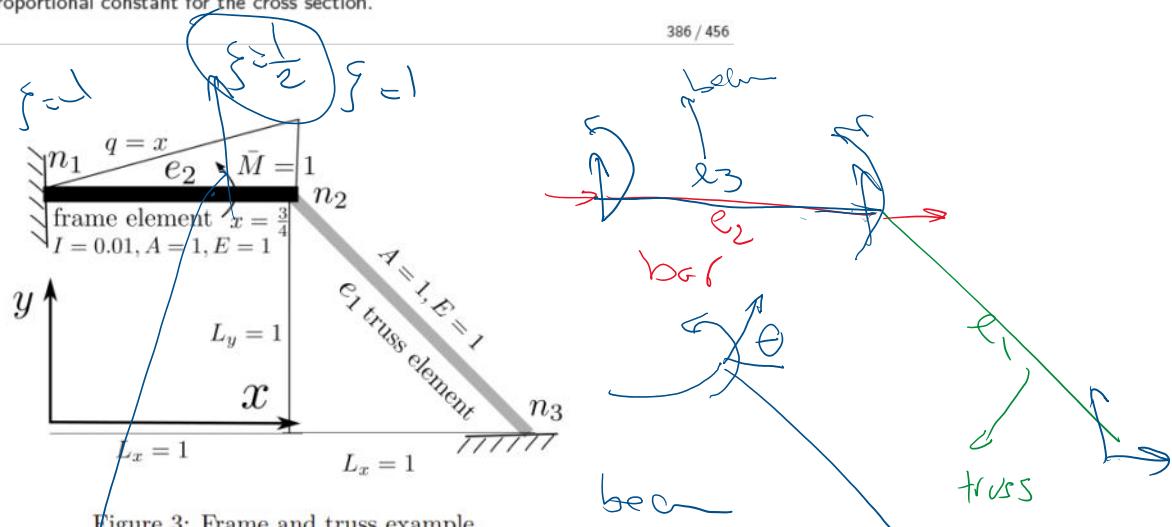
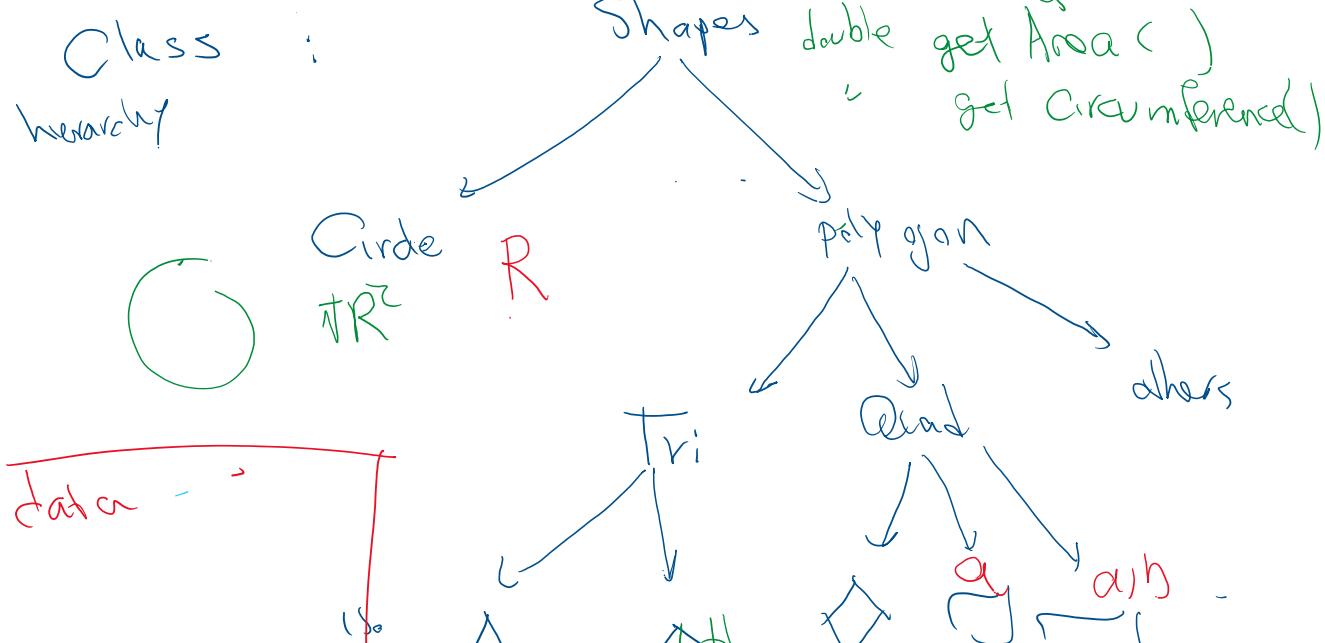
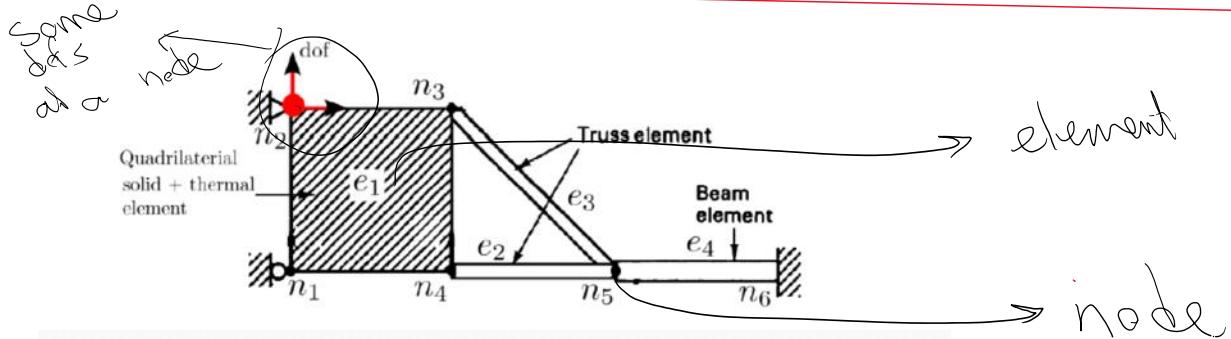
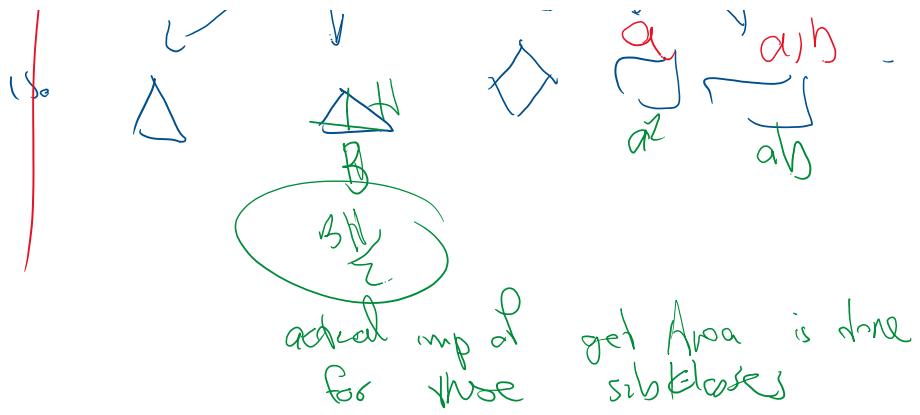


Figure 3: Frame and truss example.

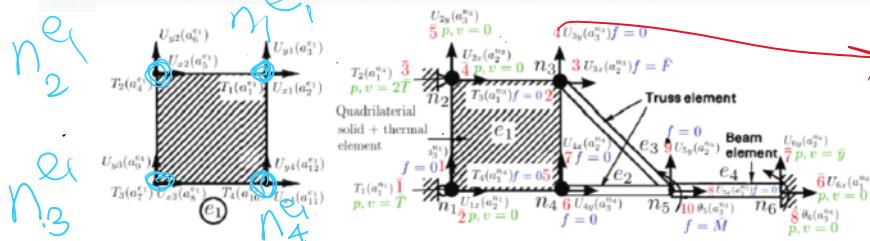
$$f_f^M = M \frac{d}{dx} \begin{bmatrix} N_1 \\ N_2 \\ N_3 \\ N_4 \end{bmatrix} = -M \frac{d}{dx} \begin{bmatrix} M_1 \\ M_2 \\ M_3 \\ M_4 \end{bmatrix} \times \begin{bmatrix} 1 \\ h/L \end{bmatrix}$$

Coding the FEM method





### FEM Solver Objects: 1. Element: Data



- id: Clearly, id of  $e_1$  is 1.
- neNodes ( $n_n^e$ ): Number of element nodes (e.g., for element 1  $n_n^e = 4$ ).
- eNodes (LEM): Indices of element nodes in global system; e.g., for  $e_1$ :

$$\text{LEM}^e_1 = [3 \ 2 \ 1 \ 4]$$

- number of element dof (nedof:  $n_f^e$ ): can be different than  $n_n^e$ , for  $e_1 : n_f^e = 12$ .
- edofs ( $a_f^e$ ):  $n_f^e$  vector of dofs; e.g., for  $e_1$ :

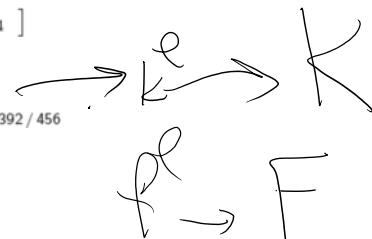
$$a_f^e = \begin{bmatrix} a_1^{e1} & a_2^{e1} & \dots & a_{n_f^e}^{e1} \end{bmatrix}$$

$$= [T_1 \quad U_{x1} \quad U_{y1} \quad T_2 \quad U_{x2} \quad U_{y2} \quad T_3 \quad U_{x3} \quad U_{y3} \quad T_4 \quad U_{x4} \quad U_{y4}]$$

- dofMap ( $M_t^e$ ): map from element to global dofs; e.g., for  $e_1$  we will observe:

$$M^{e1}_t = [2 \ 3 \ 4 \ \bar{3} \ \bar{4} \ \bar{5} \ \bar{1} \ \bar{2} \ 1 \ 5 \ 6 \ 7]$$

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- stiffness matrix ( $\mathbf{k}^e$ ):  $n_f^e \times n_f^e$  local stiffness (conductivity, etc.) matrix.
  - force vectors ( $\mathbf{f}_{de}$ ,  $\mathbf{f}_{oe}$ ,  $\mathbf{f}_{ee}$ ): A variety of element  $n_f^e$  load vectors such as essential BC ( $\mathbf{f}_D^e$ ), sum of other forces ( $\mathbf{f}_o^e = \mathbf{f}_r^e + \mathbf{f}_N^e + \dots$ , etc.), and element total  $\mathbf{f}_e^e = \mathbf{f}_o^e - \mathbf{f}_D^e$ .
  - {physics}: Physics represented by the element; e.g., for  $e_1$  physics are solid and thermal.
  - $eType$ : one or more than one object that identify the element type. Examples are shape and order of element (linear triangle, etc.).
- use  
don't  
need  
these*

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```
class PhyElement
{
    int id;
    int nNodes; // # element nodes
    vector<int> eNodes; // element node vector
    vector<PhyNode*> eNodePtrs;
    int nedof; // # element dof
    VECTOR edofs; // element dofs
    vector<int> dofMap;
    ElementType eType;
    int matID;
    MATRIX ke; // element stiffness matrix
    VECTOR foe; // element force vector from all sources other than essential BC
    VECTOR fde; // element essential BC force
    VECTOR fee; // all element forces
```

Functions:

- Calculate forces
- Calculate stiffness
- Assemble

$$\frac{\Delta E}{2} [1 \quad -1 \quad -1] \text{ stress } \frac{\Delta E}{L} [k_b - k_f]$$

