

K-G relationship (cont.)

Mode I

$$G_I = \begin{cases} \frac{K_I^2}{E} & \text{plane stress} \\ (1 - \nu^2) \frac{K_I^2}{E} & \text{plane strain} \end{cases}$$

Mixed mode

$$G = \frac{K_I^2}{E'} + \frac{K_{II}^2}{E'} + \frac{K_{III}^2}{2\mu} \quad E' = \begin{cases} \frac{E}{1 - \nu^2} & \text{for plane strain} \\ E & \text{for plane stress} \end{cases}$$

- Equivalence of the strain energy release rate and SIF approach
- Mixed mode: G is scalar => mode contributions are additive
- Assumption: self-similar crack growth!!!

Self-similar crack growth: planar crack remains planar (*da* same direction as *a*)

Key use: by having K's we can determine if the crack grows or not.

Imagine we just have mode I fracture:

$G = \frac{K_I^2}{E'}$ does the crack grow?

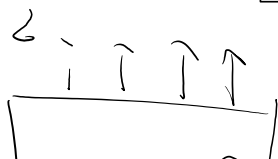
$G \geq R$ crack grows

$\frac{K_I^2}{E'} = R$

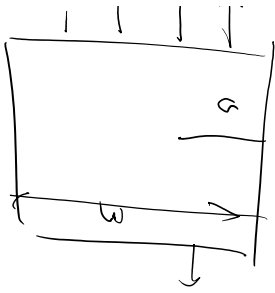
$$K_I = \sqrt{R E'} = K_{Ic}$$

K_{Ic} : critical stress intensity factor (toughness) Pa√m

crack growth potential



$K_{Ic}(\sigma) \leq \sqrt{\sigma^2 a} = K_{Ic}$ crack



$$K_I = f(a/w) \sigma \sqrt{\pi a} = K_{Ic} \quad \text{crack can grow}$$

Similarly for mode II

$$\frac{K_{II}^2}{E'} = R$$

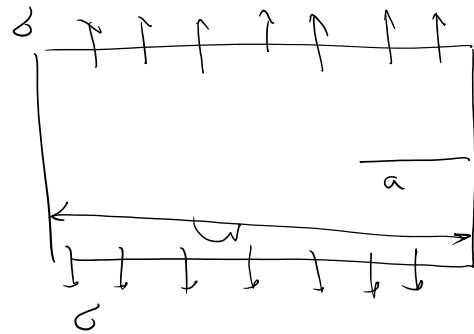
$$K_{II} = \sqrt{RE'} = K_{IIc} = K_{Ic}$$

from this perspective

Sample problems

$$K_I = f\left(\frac{a}{w}\right) \sigma \sqrt{\pi a} = K_{Ic}$$

$$\begin{bmatrix} -a \\ -\sigma \\ -K_{Ic} \end{bmatrix}$$



3 kinds of problem (2 given, find the 3rd):

(i) \underbrace{a} given, material (K_{Ic}) given \Rightarrow Find σ_{max}

eg measurement resolution in the absence of observable crack

$$\sigma_{max} = \frac{K_{Ic}}{\sqrt{\pi a} f(a/w)}$$

(ii) σ given, material (K_{Ic}) given \Rightarrow Find a_{max}

$$f\left(\frac{a_{max}}{w}\right) \sqrt{\pi a_{max}} = \frac{K_{Ic}}{\sigma}$$

in general this is a non-linear equation

(iii) σ given, a given \Rightarrow Find the appropriate material

$$K_I = \left(f\left(\frac{a}{w}\right) \sqrt{\pi a} \right) \sigma$$

Not very common

$$K_{Ic} = \left(f\left(\frac{a}{W}\right) \sqrt{\pi a} \right) \sigma$$

madrely
Not very common

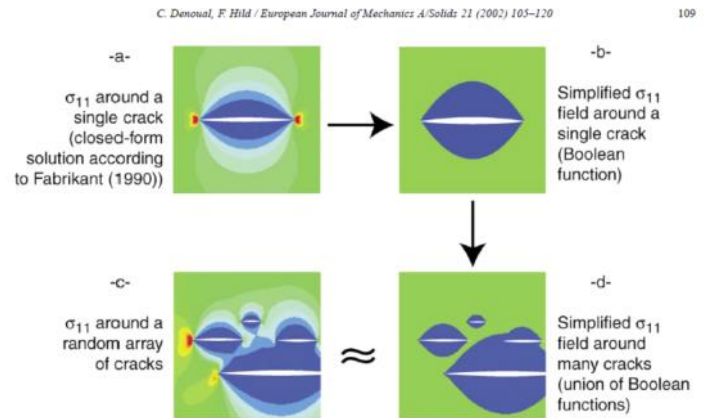
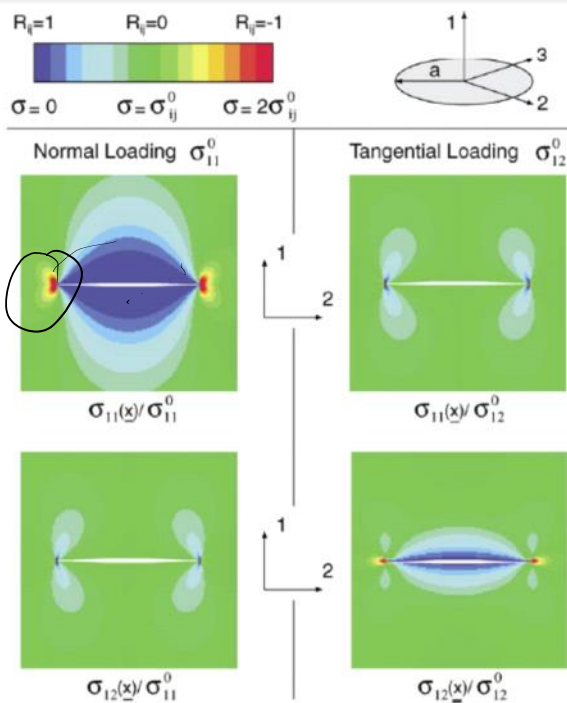
Dynamic fragmentation of brittle solids: a multi-scale model

Christophe Denoual^a, François Hild^{b,*}

^a DGA/CTA-Département Matériaux, Surfaces, Protection, 16 bis avenue Prieur de la Côte d'Or, F-94114 Arcueil Cedex, France, now at Département de Physique Théorique et Appliquée CEA/DAM, BP 12, F-91680 Bruyères le Chatel Cedex, France

^b LMT-Cachan, ENS Cachan/CNRS/University Paris 6, 61 avenue du Président Wilson, F-94235 Cachan Cedex, France

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Plastic Fracture Mechanics (PFM)

- When can we use LEFM

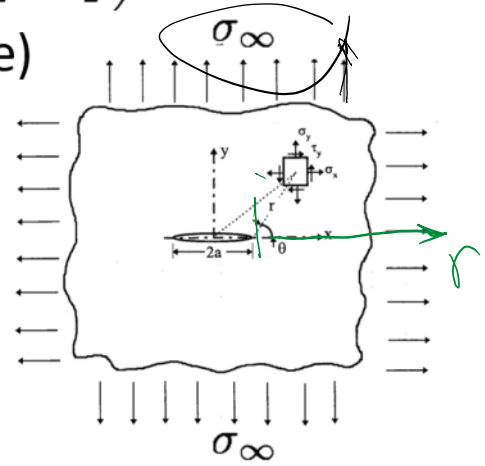
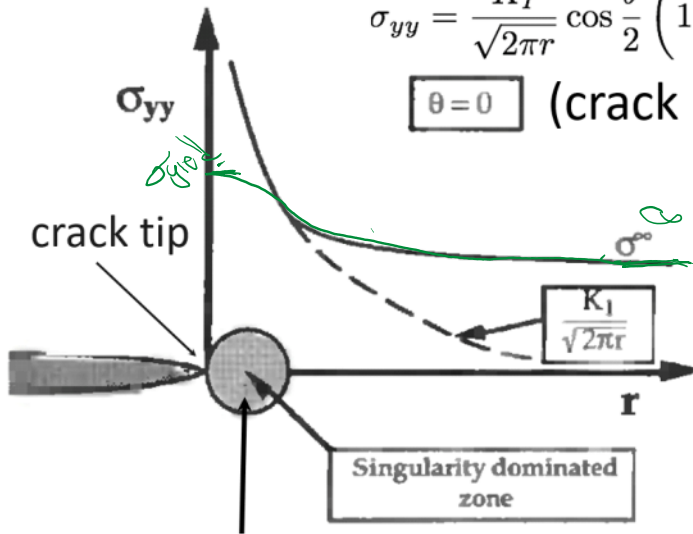
5.2. Plastic zone models

- 1D Models: Irwin, Dugdale, and Barenbolt models

Singular dominated zone

$$\sigma_{yy} = \frac{K_I}{\sqrt{2\pi r}} \cos \frac{\theta}{2} \left(1 + \sin \frac{\theta}{2} \cos \frac{3\theta}{2} \right)$$

$\theta = 0$ (crack plane)

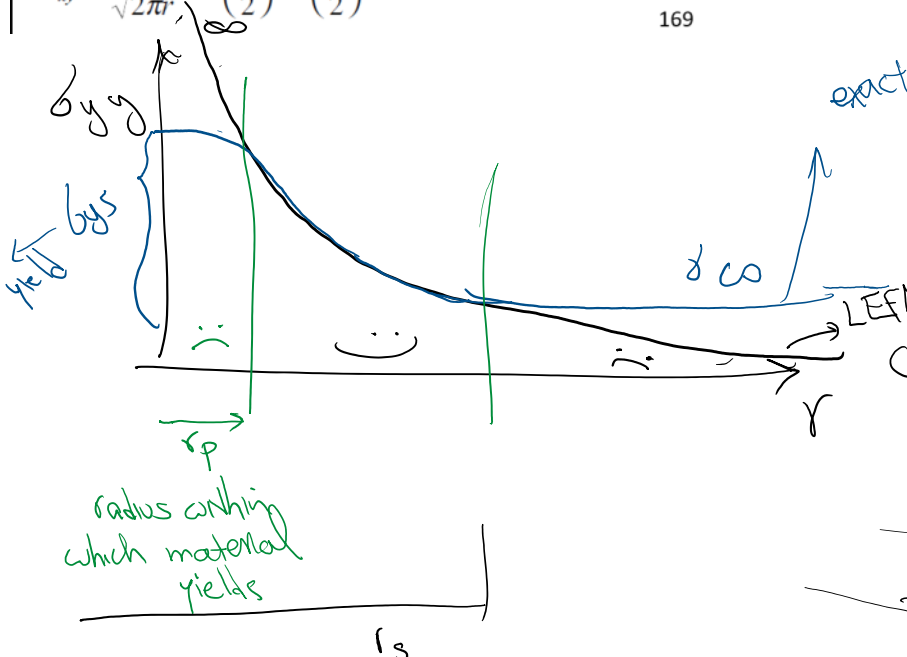
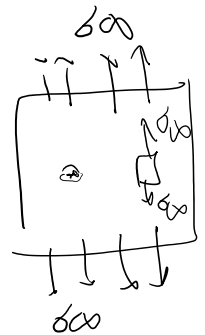


$$\sigma_{xx} = \frac{K_I}{\sqrt{2\pi r}} \cos \left(\frac{\theta}{2} \right) \left[1 - \sin \left(\frac{\theta}{2} \right) \sin \left(\frac{3\theta}{2} \right) \right]$$

$$\sigma_{yy} = \frac{K_I}{\sqrt{2\pi r}} \cos \left(\frac{\theta}{2} \right) \left[1 + \sin \left(\frac{\theta}{2} \right) \sin \left(\frac{3\theta}{2} \right) \right]$$

$$\tau_{xy} = \frac{K_I}{\sqrt{2\pi r}} \cos \left(\frac{\theta}{2} \right) \sin \left(\frac{\theta}{2} \right)$$

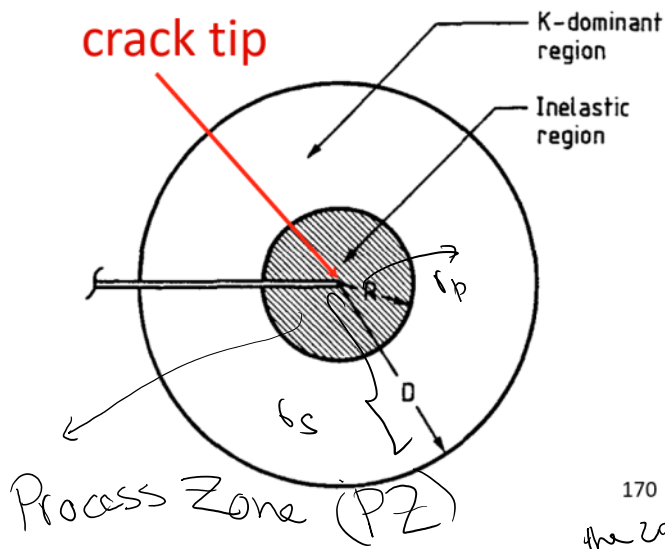
K-dominated zone



Singular dominant zone

LEFM is a good model when $r_s \gg r_p$

LEFM is a good model when $\sigma \gg \sigma_p$



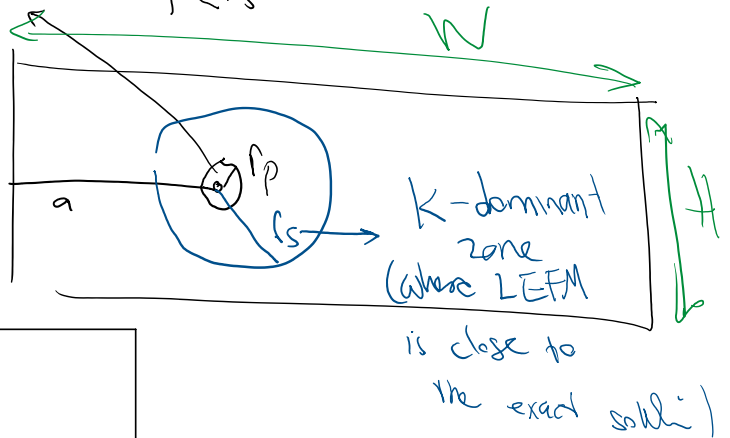
(SSY)
Small-scale yielding: LEFM still applies with minor modifications done by G. R. Irwin

$R \ll D$

process zone size (P-ZS)

SSY

170
 the zone that we have significant non-linear response (e.g. plasticity)



SSY: $r_p \ll r_s, a, W, H, \dots$
 SSY \rightarrow we can use LEFM

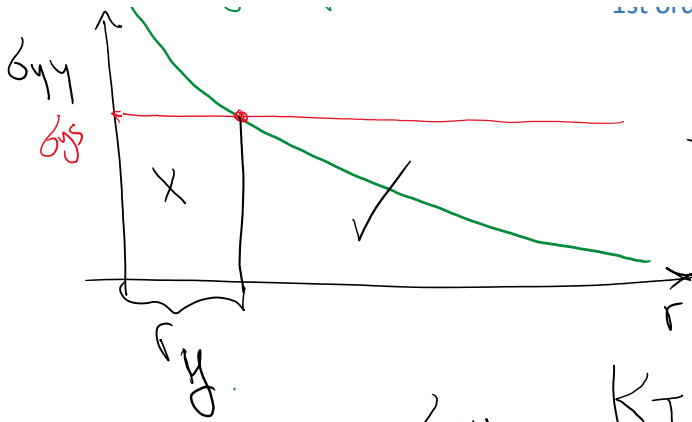
$r_p = ?$ today
 $r_s = ?$

Find r_p :

$\sigma_{yy} \uparrow$
 $\sigma_{yy} = \frac{Kt}{\sqrt{2\pi r}}$

Plastic correction:
 1st order approximation

$\sigma_{yy} \uparrow$



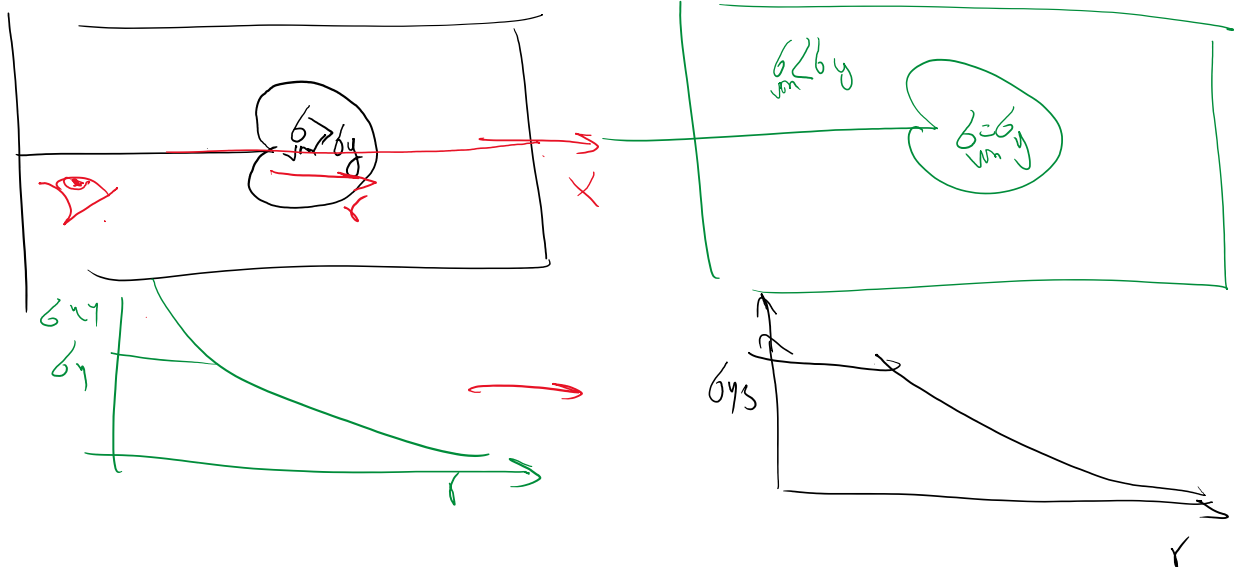
$$\sigma_{yy} = \frac{K_I}{\sqrt{2\pi r}} \quad \text{and @ } r_y : \sigma_{yy} = \sigma_{ys}$$

$$\sigma_{ys} = \frac{K_I}{\sqrt{2\pi r_y}} \Rightarrow$$

①

$$r_y = \frac{1}{2\pi} \left(\frac{K_I}{\sigma_{ys}} \right)^2$$

$$K_I: [E][W]^{1/2} \\ \sigma_{ys}: [\sigma]$$



Constitutive equation

~~X~~ breaks down

Hook's law
Isotropic

$$\sigma_{ij} = D_{ijkl} \epsilon_{kl} \quad i, j, k, l = 1, 2, 3$$

3D

$$T_{ij} = \lambda \delta_{ij} E_{kk} + 2\mu E_{ij} \quad \text{or} \quad \mathbf{T} = \lambda \mathbf{I}_E + 2\mu \mathbf{E}$$

2D (plane strain)

$$\begin{Bmatrix} \sigma_{xx} \\ \sigma_{yy} \\ \sigma_{zz} \\ \tau_{xy} \end{Bmatrix} = \frac{E}{(1+\nu)(1-2\nu)} \begin{bmatrix} (1-\nu) & \nu & 0 \\ \nu & (1-\nu) & 0 \\ \nu & \nu & 0 \\ 0 & 0 & \frac{1-2\nu}{2} \end{bmatrix} \begin{Bmatrix} \epsilon_{xx} \\ \epsilon_{yy} \\ \gamma_{xy} \end{Bmatrix}$$

2D (plane stress)

$$\begin{Bmatrix} \sigma_{xx} \\ \sigma_{yy} \\ \tau_{xy} \end{Bmatrix} = \frac{1}{1-\nu^2} \begin{bmatrix} 1 & \nu & 0 \\ \nu & 1 & 0 \\ 0 & 0 & \frac{1-\nu}{2} \end{bmatrix} \begin{Bmatrix} \epsilon_{xx} \\ \epsilon_{yy} \\ \gamma_{xy} \end{Bmatrix}$$

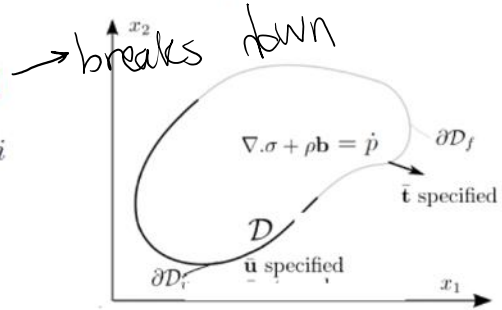
$$\text{2D (plane stress)} \begin{Bmatrix} \sigma_{xx} \\ \sigma_{yy} \\ \tau_{xy} \end{Bmatrix} = \frac{1}{1-\nu^2} \begin{bmatrix} 1 & \nu & 0 \\ \nu & 1 & 0 \\ 0 & 0 & \frac{1-\nu}{2} \end{bmatrix} \begin{Bmatrix} \varepsilon_{xx} \\ \varepsilon_{yy} \\ \gamma_{xy} \end{Bmatrix}$$

$$p = \rho v$$

Balance of linear momentum

$$\nabla \cdot \sigma + \rho \mathbf{b} = \dot{\mathbf{p}} \quad \text{OR} \quad \sigma_{ij,j} + \rho b_j = \rho \ddot{u}_i$$

- Static: $\dot{\mathbf{p}} = 0$
- No body force $\mathbf{b} = 0$

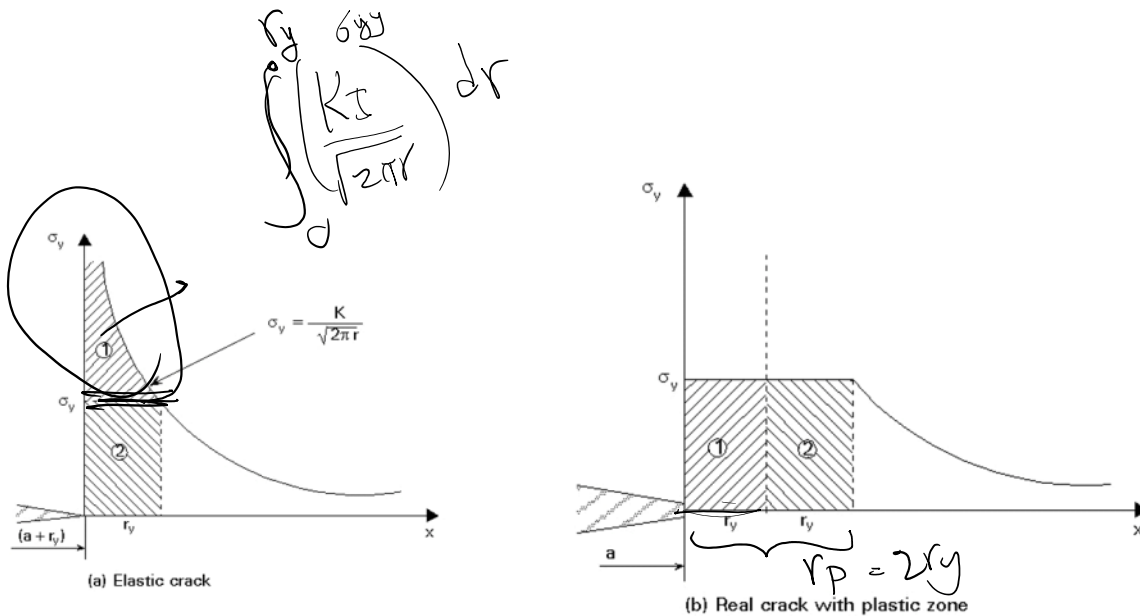


107

To get the actual Process zone size (PZS) = r_y or r_p , we need to solve the problem (BC + PDE + constitutive equations) from the beginning to find the correct solution. Where around the crack the material yields or undergoes large inelastic deformation is called the PZS.

Rather than doing this difficult problem, we can do a bit better than the 1st order approximation as shown below:

2. Irwin's plastic correction

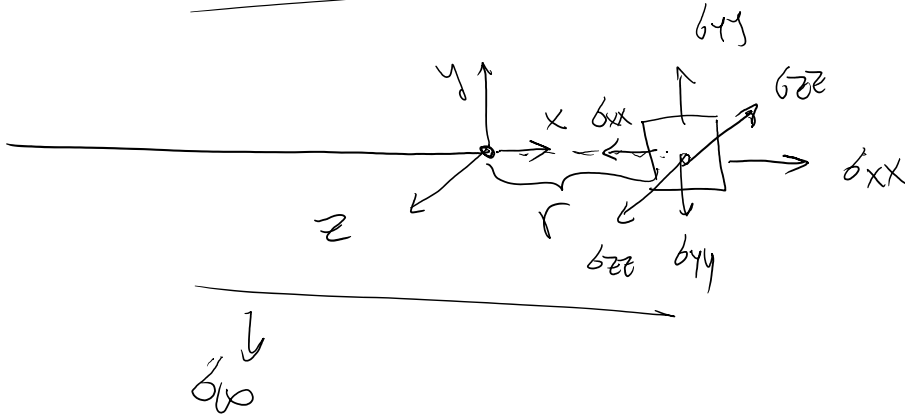


$$r_p = 2 r_y = \frac{1}{\pi} \left(\frac{KI}{\sigma_{ys}} \right)^2$$

This is better than r_y , but still is not doing any stress redistribution.

$$\sigma_{ys} = ?$$

how is it related to σ_Y yield strength



$\sigma_{xy} = 0$ (model)
 $\left\{ \begin{array}{l} \sigma_{xz} = 0 \\ \sigma_{yz} = 0 \end{array} \right.$ we're solving an in-plane problem

$\sigma_{xx}, \sigma_{yy}, \sigma_{zz}$ are principal stresses

In 2D & 3D a common stress measure to evaluate plasticity is the von Mises stress

$$\sigma_{vm} = \sqrt{\frac{(\sigma_1 - \sigma_2)^2 + (\sigma_2 - \sigma_3)^2 + (\sigma_3 - \sigma_1)^2}{2}}$$

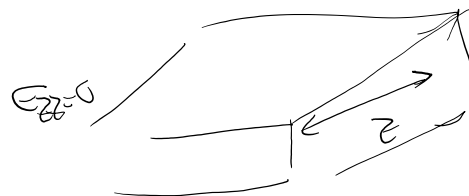
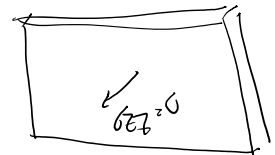
yield stress σ_Y
 $\sigma_{vm} \leq \sigma_Y$

$$\sigma_{yy} = \frac{K_I}{\sqrt{2\pi r}}$$

$$\sigma_{xx} = \frac{K_I}{\sqrt{2\pi r}}$$

$\sigma_{vm} = \sigma_Y$ material is on the yield surface

$\left. \begin{array}{l} = 0 \\ \neq 0 \end{array} \right\} \begin{array}{l} \text{p. stress} \\ \text{p. strain} \end{array}$



P. stress

$$\sigma_1 = \sigma_2 = \frac{K_I}{\sqrt{2\pi r}}$$

$\sigma_3 = 0$

Mohr circle

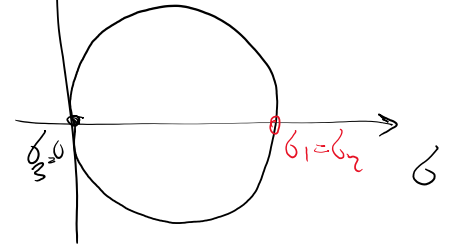
1. 0.0.0.0

$$\sigma_1 = \sigma_2 = \frac{\sqrt{2}K_f}{\sqrt{2\pi r}}$$

$$\sigma_{vm} = \sqrt{\frac{(\sigma_1 - \sigma_2)^2 + (\sigma_2 - \sigma_3)^2 + (\sigma_3 - \sigma_1)^2}{2}} = \sigma_1$$

$$= \sigma_{yy} = \sigma_y$$

0.3.0.0



3-a

p. stress

yielding condition is $\sigma_{yy} = \sigma_y$

p. strain $\sigma_1 = \sigma_2 = \sigma_{xx} = \sigma_{yy} = \frac{K_f}{\sqrt{2\pi r}}$

$\epsilon_{zz} \neq 0$ but we know $\epsilon_{zz} = 0$ but

$$\epsilon_{zz} = \frac{1}{E} (\sigma_{zz} - \nu\sigma_{xx} - \nu\sigma_{yy}) = 0 \rightarrow \sigma_{zz} - 2\nu\sigma_{yy} = 0$$

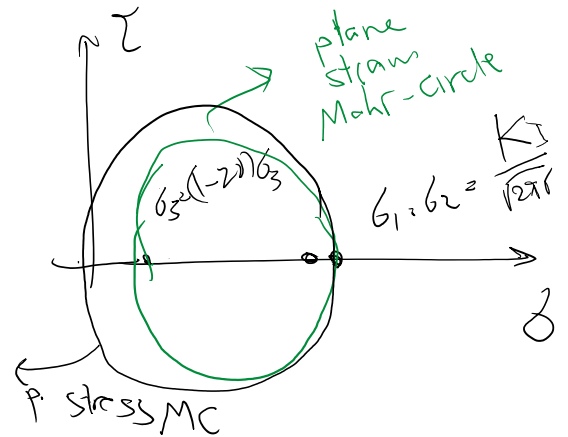
because

$$\sigma_{xx} = \sigma_{yy}$$

$$\sigma_{zz} = 2\nu\sigma_{xx}$$

$$\sigma_{vm} = \sqrt{\frac{(\sigma_1 - \sigma_2)^2 + (\sigma_1 - \sigma_3)^2 + (\sigma_2 - \sigma_3)^2}{2}}$$

$$= \sqrt{\frac{(\sigma_{yy} - \sigma_{zz})^2 + (\sigma_{yy} - \sigma_{zz})^2}{2}} = \sigma_{yy} - \sigma_{zz}$$



$$\sigma_{vm} = \sigma_{yy} - \sigma_{zz} = (1 - 2\nu)\sigma_{yy} = \sigma_y$$

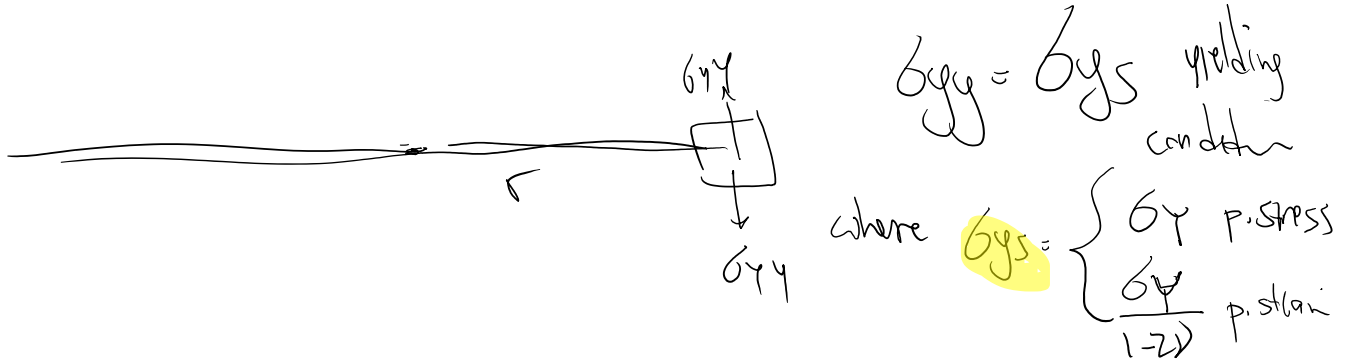
3b

p. strain

$$\sigma_{yy} = \frac{\sigma_y}{1 - 2\nu} \text{ corresponds to yielding}$$

1 - 2) to yielding

3-a
 p. stress
 yielding condition is $\sigma_{yy} = \sigma_Y$



$$r_p = \frac{1}{\pi} \left(\frac{KI}{\sigma_{Ys}} \right)^2 = \begin{cases} \frac{1}{\pi} \left(\frac{KI}{\sigma_Y} \right)^2 & \text{p. strain} \\ \frac{(1-2\nu)^2}{\pi} \left(\frac{KI}{\sigma_Y} \right)^2 & \text{p. stress} \end{cases}$$

p. stress

e.g. $\nu = .2$

$$r_p \approx \begin{cases} \frac{1}{\pi} \left(\frac{KI}{\sigma_Y} \right)^2 & \text{p. stress} \\ \frac{1}{3\pi} \left(\frac{KI}{\sigma_Y} \right)^2 & \text{p. strain} \end{cases}$$

r_p plane strain

