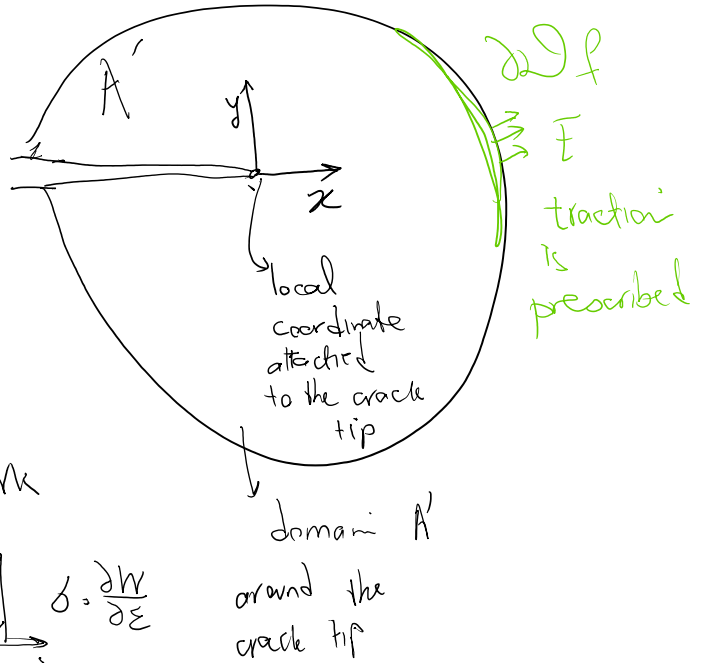


Why  $J = G$ ?

<https://rezaabedi.com/wp-content/uploads/Courses/FractureMechanics/J=G.pdf>

$$G = - \frac{d\Pi}{dA} = - \frac{d\Pi}{B da}$$

crack surface
crack length



$$\Pi = \underbrace{U_e}_{\text{internal energy}} - \underbrace{W_{ext}}_{\text{external work}}$$

$$U_e = \int_{A'} W(\epsilon) dV$$

strain energy density

1D  $\delta = \frac{\delta W}{\delta \epsilon}$

$$W_{ext} = \int_{A'} u \cdot p / b dV + \int_{\partial \Omega_f} \bar{u} \bar{t} ds$$

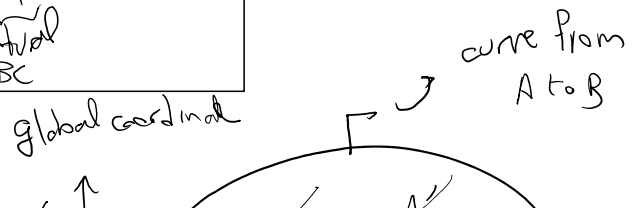
$$\Rightarrow G = - \frac{1}{B} \frac{d}{da} (\underbrace{\Pi}_{U_e - W_{ext}}) = - \frac{1}{B} \left( \frac{d}{da} \left( \int_{A'} W(\epsilon) dV - \int_{\partial \Omega_f} \bar{u} \bar{t} ds \right) \right)$$

let  $B=1$  for simplicity

$$G = - \int_{A'} \frac{d}{da} W(\epsilon) dV + \int_{\partial \Omega_f} \frac{d}{da} (\bar{u} \bar{t}) ds \quad (1)$$

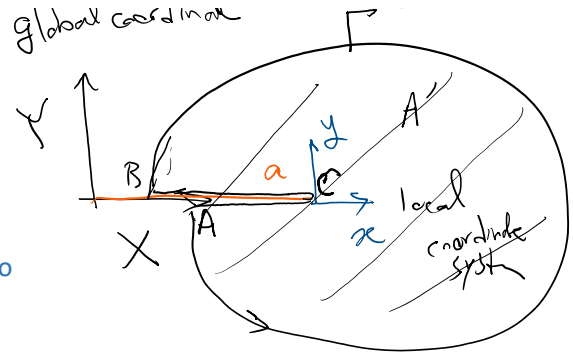
natural BC
global coordinate

$\frac{d}{da}$



$$\frac{df}{da} \Big|_{X \text{ is fixed}}$$

Since the J-integral is computed in the coordinate attached to the crack tip, we need to change the coordinate system used to express terms above and evaluate derivatives in the local coordinate system.



$$\frac{df(x,a)}{da} \Big|_{X \text{ fixed}} =$$

$$X = a + x + \text{const} \Rightarrow$$

$$x = X - a - \text{const} \Rightarrow$$

$$\frac{\partial x}{\partial a} = -1$$

$$\frac{\partial f(x,a)}{\partial x} \left( \frac{\partial x}{\partial a} \right) + \frac{\partial f(x,a)}{\partial a} \left( \frac{\partial a}{\partial a} \right)$$

$\Rightarrow$

$$\textcircled{2} \quad \frac{df(x,a)}{da} \Big|_{X \text{ fixed}} = \frac{\partial f(x,a)}{\partial a} - \frac{\partial f(x,a)}{\partial x}$$

plug (2) in (1)

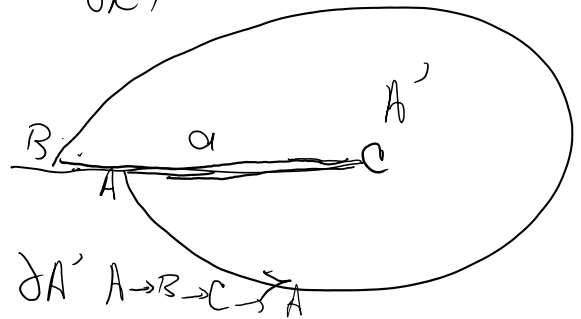
expanded derivative

$$G = \int_{A'} \frac{d}{da} W \epsilon dx - \int_{\partial A'} \frac{d}{da} (u \bar{t}) ds \quad \textcircled{1}$$

$$\frac{d}{da} (u \bar{t}) ds$$

$$\int_{\partial A'} \left( \frac{du}{da} \right) \bar{t} ds$$

$$G = \int_{A'} \left( \frac{\partial W}{\partial a} - \frac{\partial W}{\partial x} \right) + \int_{\partial A'} \left( \frac{\partial u}{\partial a} - \frac{\partial u}{\partial x} \right) \bar{t} ds$$



$$G = I_1 + I_2$$

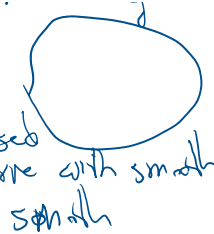
$$0 = I_1 = \int_{A'} \frac{\partial W}{\partial a} dx + \int_{\partial A'} \frac{\partial u}{\partial a} \bar{t} ds = 0 \quad (\text{see notes})$$

proof of this is similar to

$$J=0$$

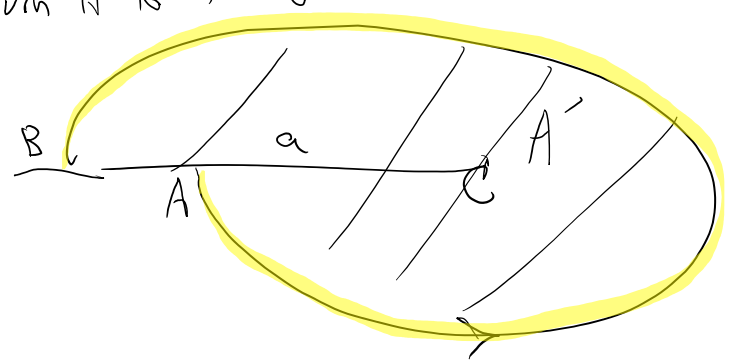
$$I_2 = \int_{A'} \frac{\partial W}{\partial x} dV + \int_{\partial A'} t ds$$

proof of this is ...  
 $\int_{\partial A'} t ds$  on a closed curve with smooth boundary



Want to integrate (divergence theorem)

turn it to boundary

$$G = \int_{A'} \frac{\partial W}{\partial x} dV + \int_{\partial A'} t ds$$


(3)

$$\int_{A'} \frac{\partial W}{\partial x_i} dV = \int_{\partial A'} W n_i ds$$

(4)

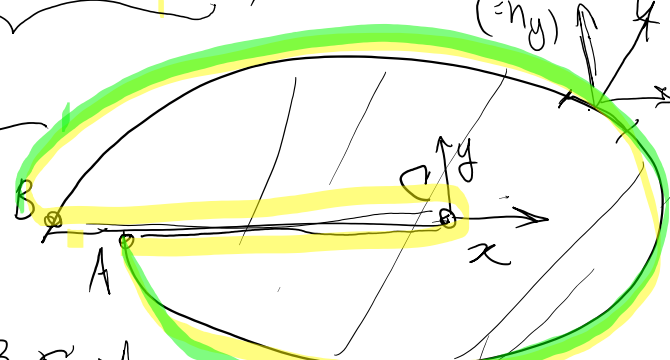
$$\int_{A'} \frac{\partial f}{\partial x_i} dV = \int_{\partial A'} f n_i ds \Rightarrow$$


$\int_{A'} \frac{\partial f}{\partial x_i} dV = \int_{\partial A'} f n_i ds$

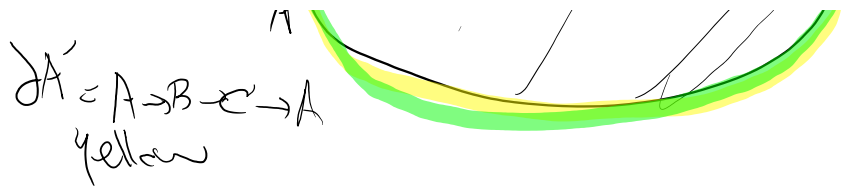
(4)

$$G = \int_{\partial A'} \left( W n_x + t \frac{du}{dx} \right) ds$$

green  $\int$  integral



$\partial A' = A \rightarrow B \rightarrow A_2$



$$G = \int \tau ds = \underbrace{\int_{A \rightarrow B} \tau ds}_J + \int_{B \rightarrow C} \tau ds + \int_{C \rightarrow A} \tau ds$$

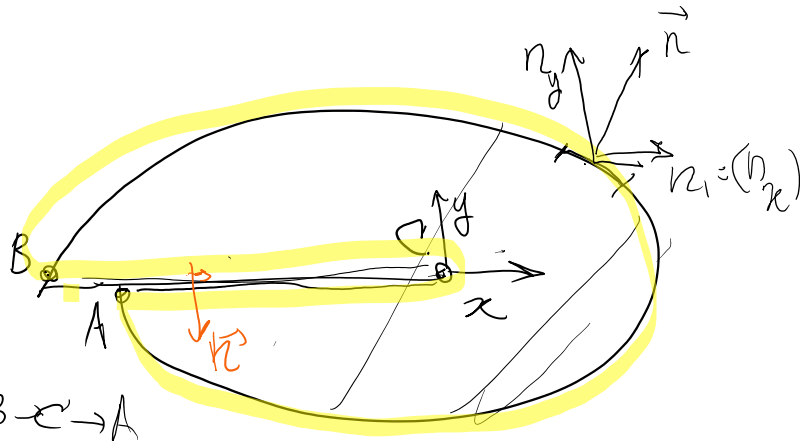
$$G = J + \int_{B \rightarrow C} \tau ds + \int_{C \rightarrow A} \tau ds \quad (5)$$

$$\int (W n_x + \tau \frac{d\bar{u}}{dx}) ds$$

$B \rightarrow C$   
 $n_x = 0$   
on crack surface  $\tau = 0$

crack surface is traction-free  
 $\tau = 0$

$\delta A' = A \rightarrow B \rightarrow C \rightarrow A$   
yellow

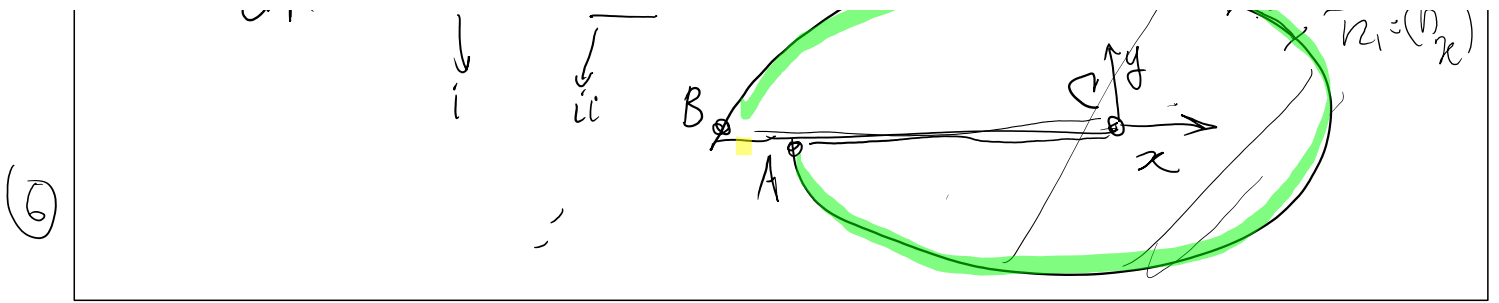


similarly  $\int_{C \rightarrow A} \tau ds = 0$

(5)  $\Rightarrow$

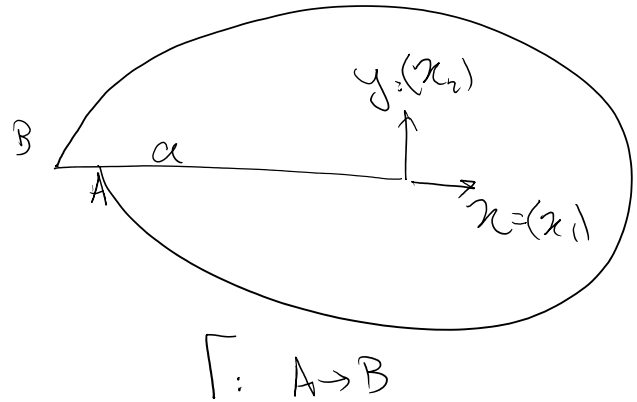
$$G = J = \int_A^B (W n_x + \tau \frac{d\bar{u}}{dx}) ds$$

$$G = -\frac{d\Pi}{dA} = \underbrace{-\frac{dU_e}{dA}}_i + \underbrace{\frac{dW_{ext}}{dA}}_{ii}$$



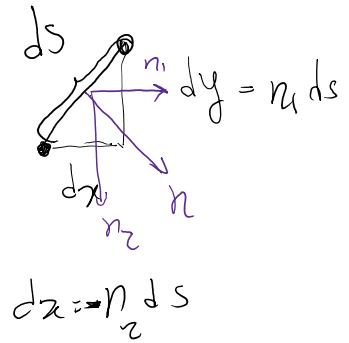
Relation of KI, KII to J1 and J2 (in-plane fracture)

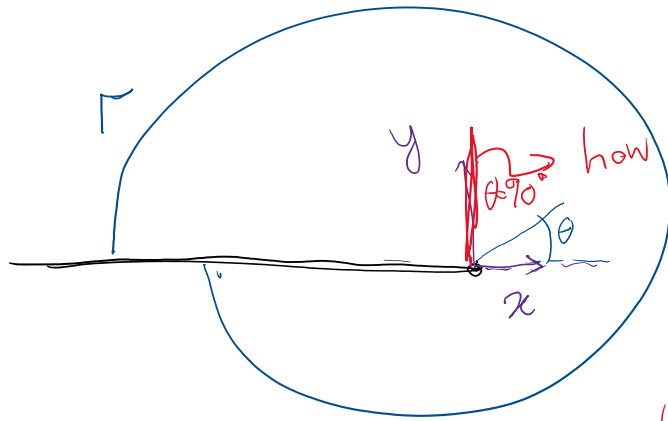
$$J_{K_i} = \int_{\Gamma} \left( W(\epsilon) n_k + \vec{t} \frac{d\vec{u}}{dx_k} \right) ds$$



$$J_1 = \int_{\Gamma} \left( -W n_1 + \vec{t} \frac{du}{dx_1} \right) ds = \int_{\Gamma} -W dy + \vec{t} \frac{du}{dx} ds$$

$$J_2 = \int_{\Gamma} \left( -W n_2 + \vec{t} \frac{du}{dx_2} \right) ds = \int_{\Gamma} W dx + \vec{t} \frac{du}{dy} ds$$





how much energy is released when the crack kinks 90°

$$J_1 = G(\theta = 0)$$

$$J_2 = G(\theta = 90^\circ)$$

$$J_1 = \int (-W dy + \vec{T} \frac{du}{dx} ds) = G(\theta = 0)$$

$$J_2 = \int (W dx + \vec{T} \frac{du}{dx} ds) = G(\theta = \frac{\pi}{2})$$

(7)

$K_I, K_{II} \iff J_1 \& J_2$

Recall

$$J_1 = G(\theta = 0) = \frac{K_I^2 + K_{II}^2}{E'} \quad (8i)$$

$$J_2 = G(\theta = \frac{\pi}{2}) = -\frac{2K_I K_{II}}{E'} \quad (8ii)$$

in-plane fracture

$$E' = \begin{cases} E & \text{p. stress} \\ \frac{E}{1-\nu^2} & \text{p. strain} \end{cases}$$

8ii  $K_{II} = \frac{J_2 E'}{2K_I}$  plug in (8i)

$$K_I^2 + \underbrace{\left( \frac{J_2 E'}{2K_I} \right)^2}_{1.7} = J_1 E'$$

multiply by 1.7

$$(K_{\pm}^2)^2 - \underbrace{\gamma_1 E'}_{k^2} (K_{\pm}^2) + \frac{(\gamma_2 E')^2}{4} = 0$$

$$z = K_{\pm}^2$$

$$z^2 - \gamma_1 E' z + \frac{(\gamma_2 E')^2}{4} = 0$$

$$K_{\pm}^2 = \pm \sqrt{\frac{E'}{2} \left( \gamma_1 \pm \sqrt{\gamma_1^2 - \gamma_2^2} \right)}$$

9

$$K_{\pm} = a$$

$$K_{\mp} = b$$

$$a = \sqrt{\frac{E'}{2} (\gamma_1 + \sqrt{\gamma_1^2 - \gamma_2^2})}$$

$$K_{\pm} = -a$$

$$K_{\mp} = -b$$

$$b = \sqrt{\frac{E'}{2} (\gamma_1 - \sqrt{\gamma_1^2 - \gamma_2^2})}$$

$$K_{\pm} = b$$

$$K_{\mp} = a$$

$$K_{\pm} = -b$$

$$K_{\mp} = -a$$

$\sigma$  is higher than  $\tau$

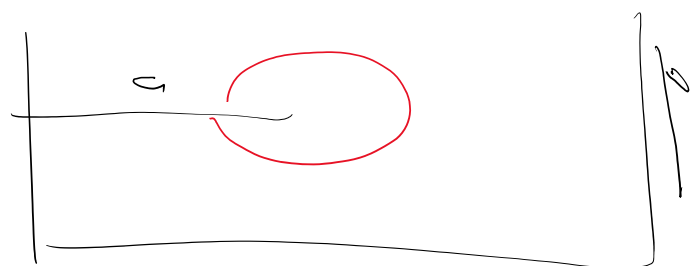
Choose the correction solution at hand

$$\gamma_1 = 109 \quad E' = 1$$

$$\gamma_2 = -60$$

$$a, b = \pm \sqrt{\frac{\gamma_1 \pm \sqrt{\gamma_1^2 - \gamma_2^2}}{2 E'}}$$

$$= \pm \sqrt{\frac{109 \pm \sqrt{109^2 - (-60)^2}}{2}}$$



$$z = \frac{10}{3}$$

$$\begin{cases} K_{\pm} = 10 \\ K_{\mp} = 3 \end{cases}$$

$$= \pm \frac{\sqrt{109 \pm \sqrt{109^2 - (60)^2}}}{2}$$

$$= \frac{10}{3}$$

$$\left| \begin{array}{l} K_I = 10 \\ K_{II} = 3 \end{array} \right|$$

## Energy release rate of J integral: Assumptions

1. Homogeneous body

2. Linear or non-linear elastic solid

no plasticity

3. No inertia, or body forces; no initial stresses

4. No thermal loading

5. 2-D stress and deformation field

6. Plane stress or plane strain

7. Mode I loading

8. Stress free crack

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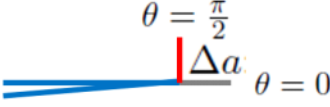
We'll see a more general form of J integral removes ALL these restrictions!



# J-K relationship

In fact both  $J_1$  (J) and  $J_2$  are related to SIFs:

$$J_1 = \int_{\Gamma} \left( w dy - t \frac{\partial u}{\partial x} d\Gamma \right) \quad \text{J1 \& J2: crack advance for } (\theta = 0, 90) \text{ degrees}$$

$$J_2 = \int_{\Gamma} \left( w dx - t \frac{\partial u}{\partial y} d\Gamma \right)$$


$$J = J_1 - iJ_2 \quad \text{Hellen and Blackburn (1975)}$$

$$= \frac{(1+\nu)(1+\kappa)}{4E} (K_I^2 + K_{II}^2 + 2iK_I K_{II})$$

$$J_1 = \frac{K_I^2 + K_{II}^2}{E'}$$

$$J_2 = \frac{-2K_I K_{II}}{E'}$$

$$E' = \begin{cases} E & \text{plane strain} \\ \frac{E}{1-\nu^2} & \text{plane stress} \end{cases}$$

Note that if  $K_I = a, K_{II} = b$  is a solution the general solution is:

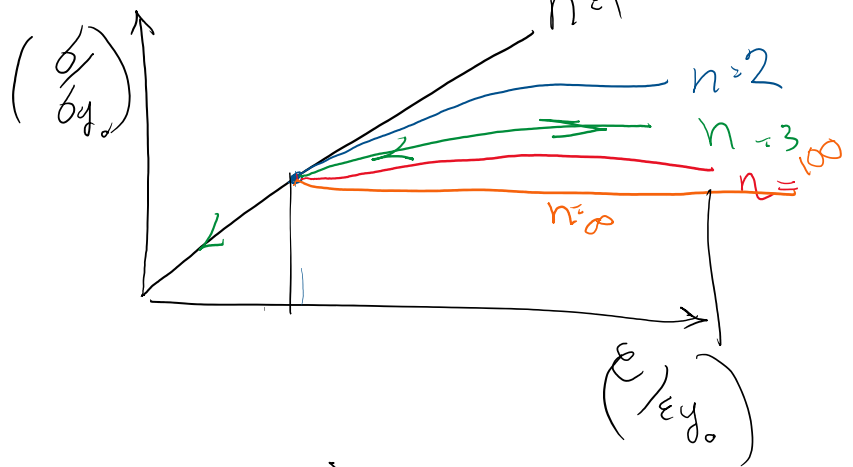
$$K_I = \pm a, K_{II} = \pm b \text{ and } K_I = \pm b, K_{II} = \pm a$$

5.3.5 plastic crack tip fields:  
Hutchinson, Rice, Rosengren (HRR) solution

Ramberg-Osgood model (Nonlinear elasticity)

$$\frac{\epsilon}{\epsilon_{y_0}} = \underbrace{\frac{\sigma}{\sigma_{y_0}}}_{\text{like } \epsilon_e} + \alpha \underbrace{\left( \frac{\sigma}{\sigma_{y_0}} \right)^n}_{\text{like } \epsilon_p}$$

$\epsilon = \epsilon_e + \epsilon_p$   
in plasticity





$$\frac{\sigma}{E y_0} = \left( \frac{\sigma}{\sigma_0} \right) + \alpha \left( \frac{\sigma}{\sigma_0} \right)^n$$

this term that is similar to  $\epsilon_p$  dominates

we ignore this

$$E \propto \sigma^n$$

close to the crack tip in RC model