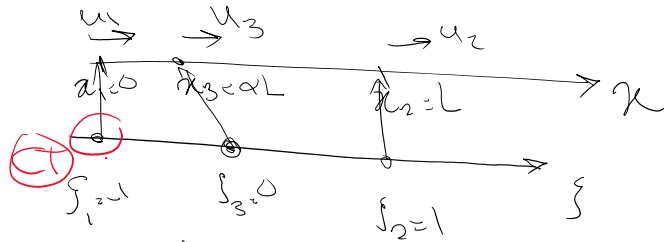


From last time:

$\epsilon \rightarrow 0$ at ξ
 $\xi \rightarrow -1$



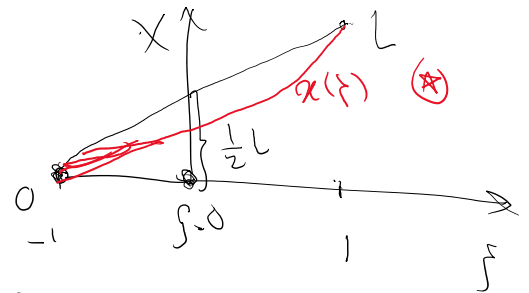
$$u(\xi) = u_1 N_1(\xi) + u_2 N_2(\xi) + u_3 N_3(\xi)$$

$$x(\xi) = x_1 N_1(\xi) + x_2 N_2(\xi) + x_3 N_3(\xi) \Rightarrow$$

$$x(\xi) = L \left(\alpha + \frac{\xi}{2} + \xi^2 \left(\frac{1}{2} - \alpha \right) \right)$$

$$E(x) = \frac{du}{dx} = \frac{\frac{du}{d\xi}}{\frac{dx}{d\xi}}$$

computed from 1
computed from (*)



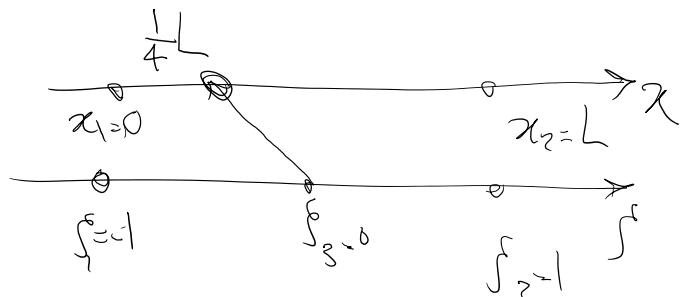
to create $\epsilon \rightarrow \infty$ @ $\xi = -1$
 denominator must be zero @ $\xi = -1$

$$\frac{dx}{d\xi} = L \left(\frac{1}{2} + 2\xi \left(\frac{1}{2} - \alpha \right) \right)$$

$$\text{for } \xi = -1 \Rightarrow \frac{dx}{d\xi}(\xi = -1) = L \left(\frac{1}{2} + 2(-1) \left(\frac{1}{2} - \alpha \right) \right) = 0$$

$$\alpha = \frac{1}{4}$$

for $\alpha = \frac{1}{4}$ from (*)



$$x = \frac{L}{4} (\xi + 1)^2 \Rightarrow \xi = 2\sqrt{\frac{x}{L}} - 1 \quad (2)$$

$$\text{eqn 1 } u(\xi) = \sum_{i=1}^3 u_i N_i(\xi) = u_1 \frac{\xi(\xi-1)}{2} + u_2 \frac{\xi(\xi+1)}{2} + u_3 (1-\xi^2)$$

$$u = u_1 + \sqrt{\frac{x}{L}} (-3u_1 - u_2 + 4u_3) + \frac{2x}{L} (u_1 + u_2 - 2u_3)$$

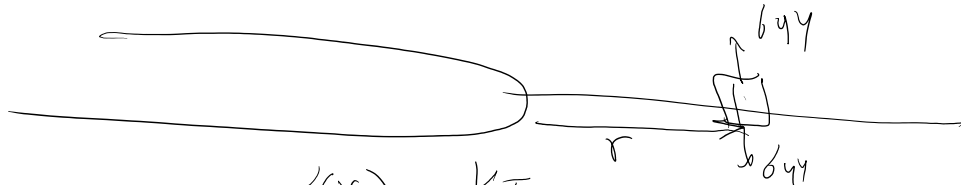
comparison

$v_1 = v_1 + \sqrt{\frac{L}{r}} (-u_1 - u_2 + 4u_3) + \frac{2u}{L} (u_1 + u_2 - 2u_3)$
 comparison
 to CT displacement

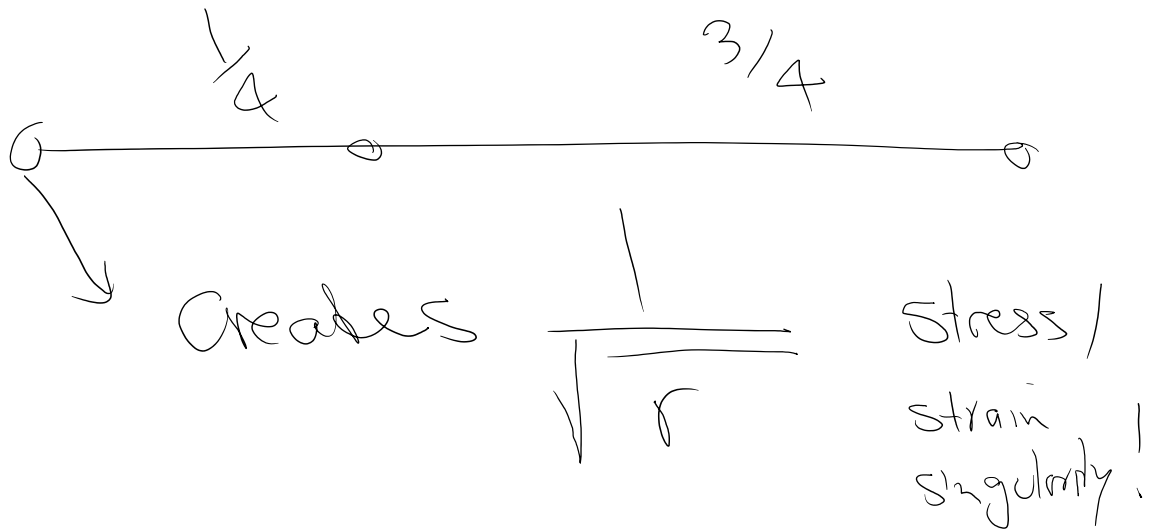


$$v(r) = \frac{K(1-\nu)}{\sqrt{2\pi r}} \sqrt{r}$$

$$E(u) = \frac{du}{dx} = \frac{1}{\sqrt{x}} \left(-\frac{3}{2} u_1 - \frac{u_2}{2} + 2u_3 \right) + \frac{2}{L} (u_1 + u_2 - 2u_3)$$



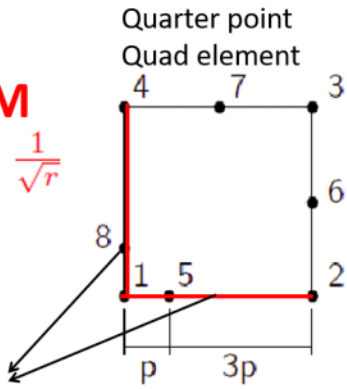
$$\sigma_{yy}(r) = \frac{KI}{\sqrt{2\pi r}}$$



Isoparametric singular elements

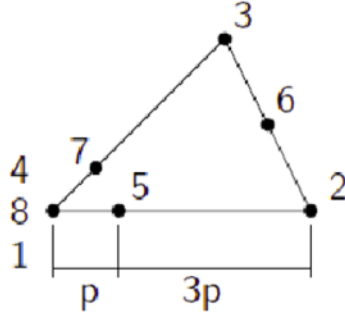
• **LEFM**

$$\epsilon, \sigma : \frac{1}{\sqrt{r}}$$



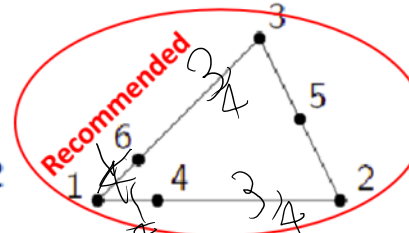
singular form $\frac{1}{\sqrt{r}}$ only along these lines
NOT recommended

Quarter point **collapsed** Quad element



Improvement:
- $\frac{1}{\sqrt{r}}$ from inside all element
Problem
- Solution inaccuracy and sensitivity when opposite edge 3-6-7 is curved

Quarter point Tri element



Improvement:
- Better accuracy and less mesh sensitivity

How should the finite element mesh look like around a CT?

Strain singularity at ξ means

$$\frac{dx}{d\xi} = L \left\{ 2\xi \left(\frac{1}{2} - \alpha \right) + \frac{1}{2} \right\} < 0$$

must be zero. Accordingly,

Singularity at infinity ($\xi \rightarrow -\infty$)

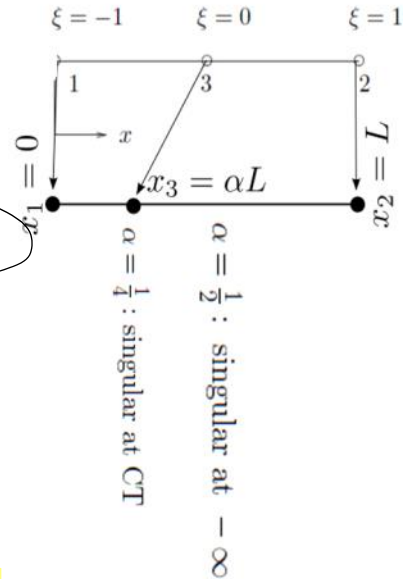
$$\alpha > \left(\frac{1}{2} \right)^-$$

Singularity at crack tip ($\xi = -1$)

$$\alpha = \frac{1}{4}$$

Singularity inside element (not of interest) ($-1 < \xi < 0$)

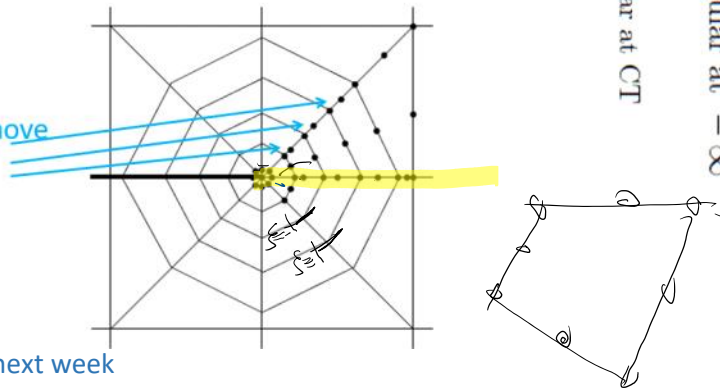
$$0 < \alpha < \frac{1}{4}$$



• **Transition elements:**

According to this analysis mid nodes of next layers move to $\frac{1}{2}$ point from $\frac{1}{4}$ point

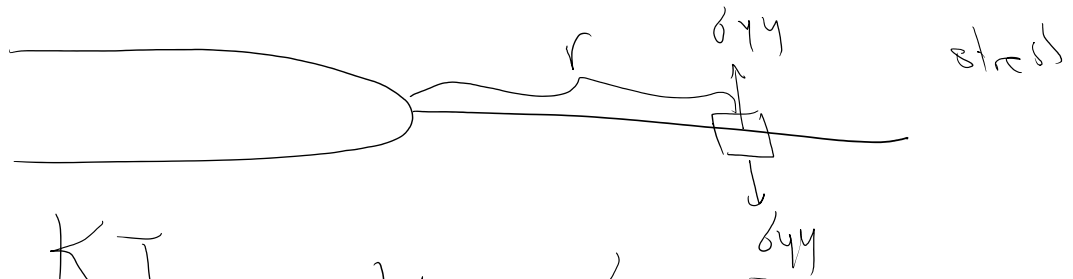
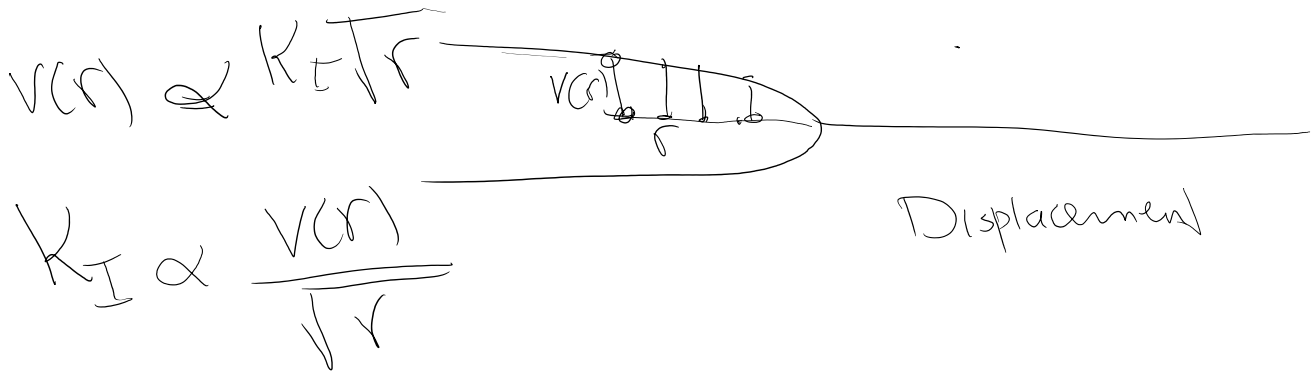
Lynn and Ingraffea 1977)



We'll create meshes like this next week

Now that we have the FEM mesh around the crack tip, how do we calculate K and G?

6.1.3. Extraction of K (SIF), G

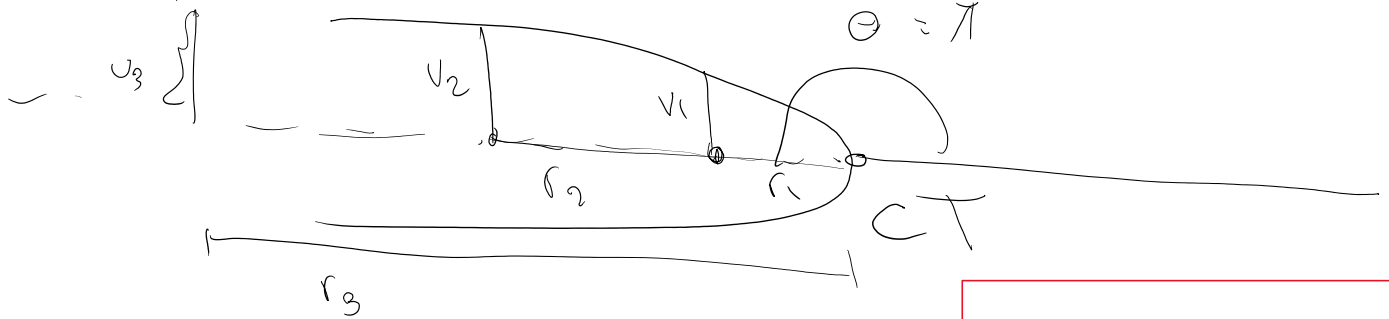


$$\sigma_{yy}(r) = \frac{K_I}{\sqrt{2\pi r}}$$

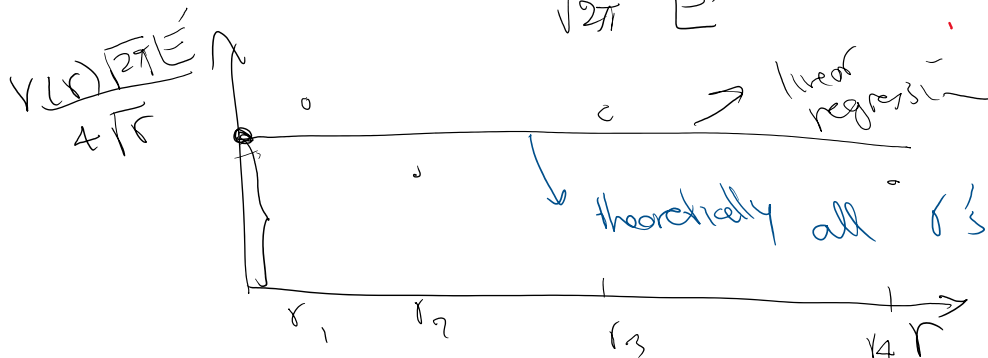
$$K_I = \sigma_{yy} \sqrt{2\pi r}$$

very poor computationally

Displacement approach

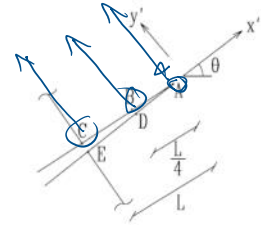


$$V(r, \theta = \pi) = \frac{4 K_I \sqrt{r}}{\sqrt{2\pi} E'} \Rightarrow \textcircled{3} K_I = \frac{V(r) \sqrt{2\pi} E'}{4 \sqrt{r}}$$



- With special CT elements (1/4, 3/4 mid-node location), we can also calculate K from nodal displacements

or alternatively from the first quarter point element:



$$v = K_I \frac{\kappa + 1}{2G} \sqrt{\frac{r}{2\pi}}$$

$$\left. \begin{aligned} u' &= \bar{u}'_A + (-3\bar{u}'_A + 4\bar{u}'_B - \bar{u}'_C) \sqrt{\frac{r}{L}} + (2\bar{u}'_A + 2\bar{u}'_C - 4\bar{u}'_B) \frac{r}{L} \\ v' &= \bar{v}'_A + (-3\bar{v}'_A + 4\bar{v}'_B - \bar{v}'_C) \sqrt{\frac{r}{L}} + (2\bar{v}'_A + 2\bar{v}'_C - 4\bar{v}'_B) \frac{r}{L} \end{aligned} \right\} \rightarrow$$

$$K_I = \frac{2G}{\kappa + 1} \sqrt{\frac{2\pi}{L}} (-3\bar{v}'_A + 4\bar{v}'_B - \bar{v}'_C)$$

Recall for 1D

$$u = u_1 + \frac{\sqrt{x}}{\sqrt{L}} (-3u_1 - u_2 + 4u_3) + \frac{2x}{L} (u_1 + u_2 - 2u_3)$$

Handwritten derivation:

$$K_I \frac{\kappa + 1}{2G} \sqrt{\frac{r}{2\pi}} = (-3\bar{v}'_A + 4\bar{v}'_B - \bar{v}'_C) \frac{r}{L}$$

$$\Rightarrow K_I = \frac{2G}{\kappa + 1} \sqrt{\frac{2\pi}{L}} (-3\bar{v}'_A + 4\bar{v}'_B - \bar{v}'_C)$$

pure mode I fracture

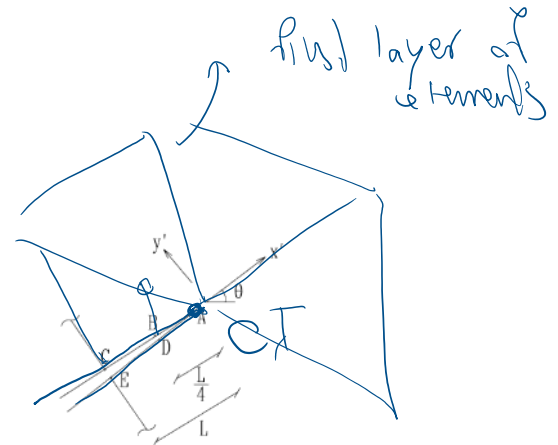
What about mixed mode fracture? Can we use behind the crack displacements to extract KI and KII?

$$\begin{Bmatrix} K_I \\ K_{II} \end{Bmatrix} = \frac{1}{2} \frac{2G}{\kappa + 1} \sqrt{\frac{2\pi}{L}} \begin{bmatrix} 0 & 1 \\ 1 & 0 \end{bmatrix} \begin{bmatrix} -3\bar{u}'_A + 4(\bar{u}'_B - \bar{u}'_D) - (\bar{u}'_C - \bar{u}'_E) \\ -3\bar{v}'_A + 4(\bar{v}'_B - \bar{v}'_D) - (\bar{v}'_C - \bar{v}'_E) \end{bmatrix}$$

170

Mixed mode generalization:

we'll do this next week too

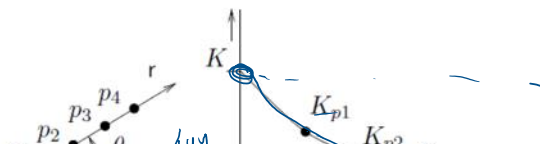


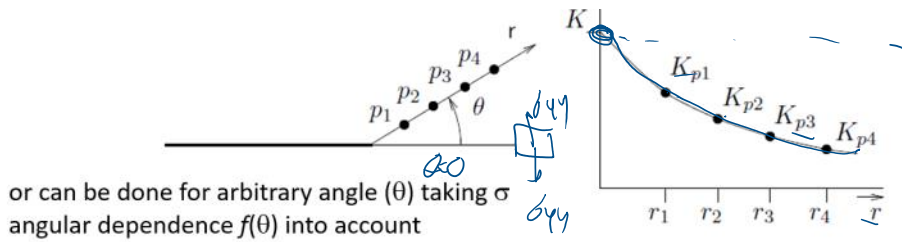
K from the stress field ahead of the crack

2. Stress

$$\sigma_{yy}(r, \theta=0) = \frac{K_I}{\sqrt{2\pi r}} \Rightarrow K_I = \sqrt{2\pi r} \sigma_{yy}(r)$$

$$K_I = \lim_{r \rightarrow 0} \left(\sqrt{2\pi r} \sigma_{22} |_{\theta=0} \right) ; K_{II} = \lim_{r \rightarrow 0} \left(\sqrt{2\pi r} \sigma_{12} |_{\theta=0} \right)$$





Computationally, it's much better to use the displacement approach for the following reasons:

$$1. \delta \propto \frac{1}{\sqrt{r}} \quad \text{and} \quad v \propto \sqrt{r}$$

$\text{as } r \rightarrow 0$

much more well-behaved and easier to capture numerically

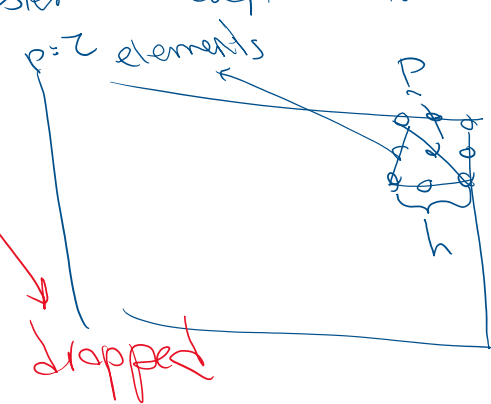
$$2. \|u - u_{\text{exact}}\|_{\text{error}} \propto Ch^{P+1}$$

in displacement

$$\| \epsilon - \epsilon_{\text{exact}} \| \propto Ch^P$$

$$\epsilon = \frac{\nabla u \cdot \nabla u}{2}$$

1 der



Strain & stress are less accurate in FEM in general

3. stress approach is sensitive to traction

→ stress approach is sensitive to mesh on the crack surfaces
 stress approach is even worse

Other approaches indirectly calculate K from G (or J)

2. K from energy approaches

1. Elementary crack advance (two FEM solutions for a and $a + \Delta a$)
2. Virtual Crack Extension: Stiffness derivative approach
3. J-integral based approaches (next section)

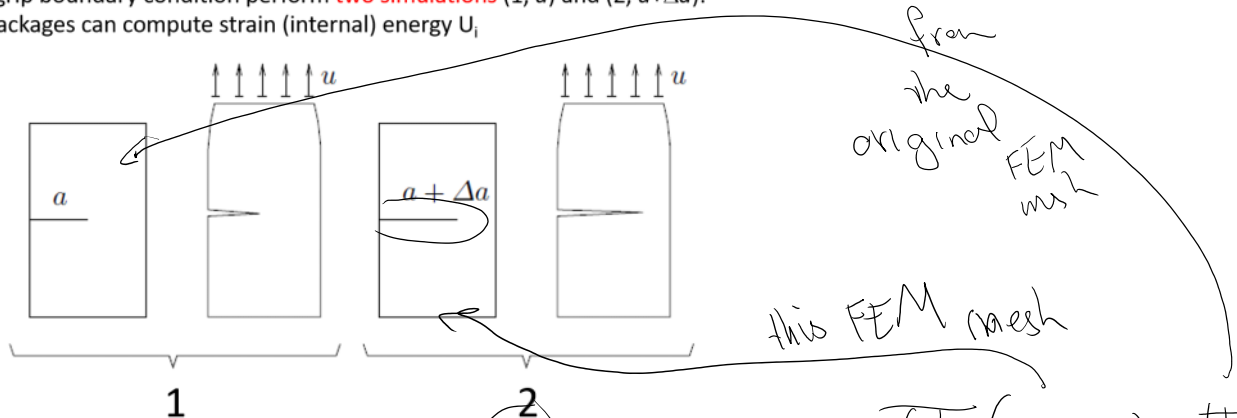
After obtaining G (or $J=G$ for LEFM) K can be obtained from

$$K_I^2 = E'G \quad \checkmark$$

$$E' = \begin{cases} E & \text{plane stress} \\ \frac{E}{1-\nu^2} & \text{plane strain} \end{cases}$$

2.1 Elementary crack advance

For fixed grip boundary condition perform **two simulations** (1, a) and (2, $a+\Delta a$):
 All FEM packages can compute strain (internal) energy U_i



$$G = - \frac{dU_e}{da} = - \frac{d}{da} (U_e) \approx - \frac{U_e(a+\Delta a) - U_e(a)}{\Delta a}$$

$$G = - \frac{dT_e}{dA} = - \frac{1}{B} \left(\frac{dT_e}{da} \right) \approx \frac{1}{B} \left(\frac{|T_e|^{(a+\Delta a)} - |T_e|^{(a)}}{\Delta a} \right)$$

1. Delta a small -> bad run into finite precision errors
2. Delta a large -> derivative approximation is getting worse

Δa → 0

Other pitfall is that we need two FEM solutions

2.2 Virtual crack extension

$$G = - \frac{dT_e}{dA} = - \frac{1}{B} \frac{dT_e}{da}$$

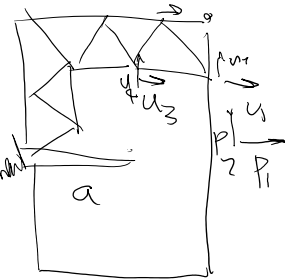
$$T_e = U_e - W \rightarrow \text{external work}$$

in FEM

$$U_e = \frac{1}{2} u^t K u$$

↑
node displacement vector

↓
stiffness matrix



$$W = P u$$

$$G = - \frac{1}{B} \frac{d(U_e - W)}{da} = - \frac{1}{B} \frac{d}{da} \left(\frac{1}{2} u^t K u - P u \right)$$

↑
equal

$$G = - \frac{1}{B} \left(\frac{1}{2} \left(\frac{du}{da} \right)^t K u + \frac{1}{2} u^t \frac{dK}{da} u + \frac{1}{2} u^t K \frac{du}{da} - \frac{dP}{da} u - P \frac{du}{da} \right)$$

$$\left(\frac{du}{da} \right)^t K u = \left(\left(\frac{du}{da} \right)^t K u \right)^t = u^t K^t \frac{du}{da}$$

in FEM $K^t = K$

$$= u^t K \frac{du}{da}$$

in $U^{-1} K^{-1}$

$$= U^T K \frac{dU}{d\epsilon}$$

$$B G = \left(-\frac{dU}{da} K U + P \frac{dU}{da} \right) = \frac{1}{2} U^T \frac{dK}{da} U + \frac{dP}{da} U$$

FEM

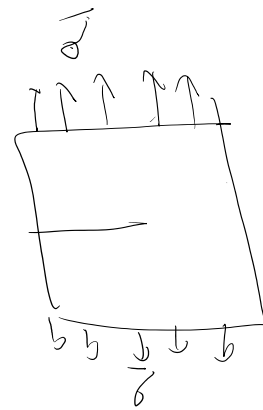
$$K U = P$$

$$B G = -\frac{1}{2} U^T \frac{dK}{da} U + \frac{dP}{da} U$$

in many problems

loads do not depend on ϵ

$$\frac{dP}{da} = 0$$



$$\Rightarrow \text{most cases } B G = -\frac{1}{2} U^T \frac{dK}{da} U$$